

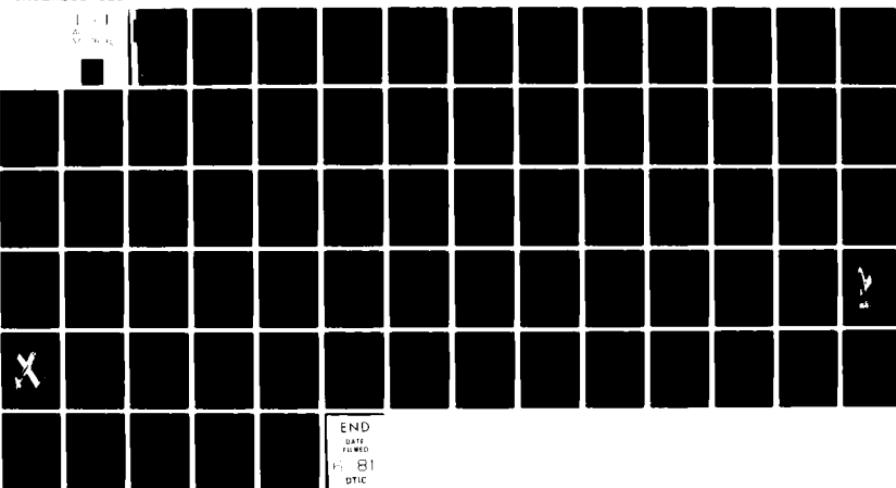
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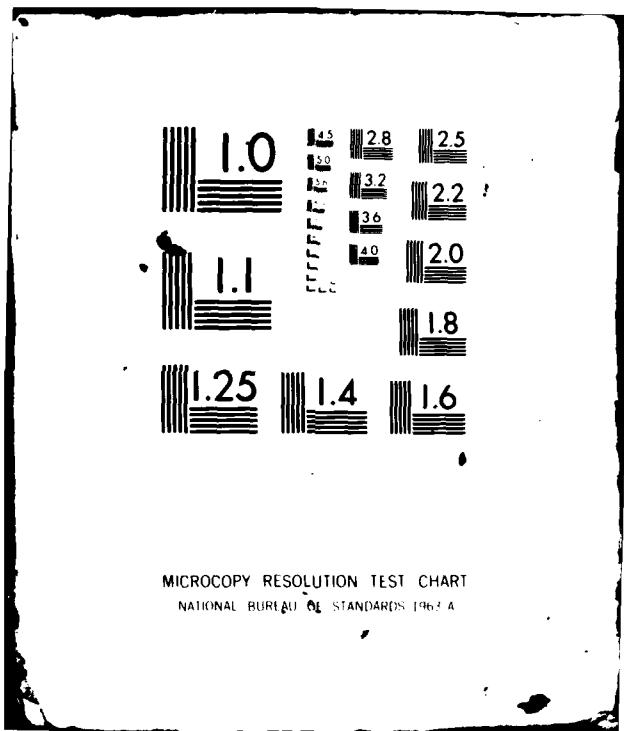
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AERODYNAMICS NOTE 401

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THE EFFECT OF A FLARE PACK ON THE  
HIGH SPEED PERFORMANCE OF THE  
JINDIVIK MK203B TARGET AIRCRAFT

by

B. D. FAIRLIE

Approved for Public Release.



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AERODYNAMICS NOTE 401

6 THE EFFECT OF A FLARE PACK ON THE  
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JINDIVIK MK203B TARGET AIRCRAFT

by

B. D. FAIRLIE

(12) 72

SUMMARY

Transonic wind tunnel tests are reported on a 1/20th scale model of the Jindivik target aircraft for Mach numbers in the range 0.5 to 0.9. The purpose of these tests was to determine the effect of the addition of a flare pack to the lower rear fuselage on the lateral and longitudinal stability of the aircraft and on the tailplane and elevator effectiveness. The results indicate that the effect of the flare pack is very small and should not significantly degrade the stability or performance of the aircraft.

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**ABSTRACT**

Transonic wind tunnel tests are reported on a 1/20th scale model of the Jindivik target aircraft for Mach numbers in the range 0.5 to 0.9. The purpose of these tests was to determine the effect of the addition of a flare pack to the lower rear fuselage on the lateral and longitudinal stability of the aircraft and on the tailplane and elevator effectiveness. The results indicate that the effect of the flare pack is very small and should not significantly degrade the stability or performance of the aircraft.

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## NOTATION

$a_1$	Tailplane effectiveness = $\partial C_{LT}/\partial \alpha_T = \partial C_{LT}/\partial \eta_T$
$a_2$	Elevator effectiveness = $\partial C_{LT}/\partial \eta$
$b$	Model nominal wingspan = 11·40 in (289·56 mm)
$C_C$	Cross wind force coefficient = Cross wind force/ $\frac{1}{2}\rho v^2 S$
$C_D$	Drag force coefficient = Drag force/ $\frac{1}{2}\rho v^2 S$
$C_L$	Lift force coefficient = Lift force/ $\frac{1}{2}\rho v^2 S$
$C_{LT}$	Tailplane lift force coefficient = Tailplane lift force/ $\frac{1}{2}\rho v^2 S_T$
$C_X$	Axial force coefficient = Axial force/ $\frac{1}{2}\rho v^2 S$
$C_{XB}$	Base force coefficient = $(p_B - p)S_B/\frac{1}{2}\rho v^2 S$
$C_Y$	Side force coefficient = Side force/ $\frac{1}{2}\rho v^2 S$
$C_Z$	Normal force coefficient = Normal force/ $\frac{1}{2}\rho v^2 S$
$C_I$	Rolling moment coefficient = Rolling moment about centre of gravity/ $\frac{1}{2}\rho v^2 S b$
$C_m$	Pitching moment coefficient = Pitching moment about centre of gravity/ $\frac{1}{2}\rho v^2 S c$
$C_n$	Yawing moment coefficient = Yawing moment about centre of gravity/ $\frac{1}{2}\rho v^2 S b$
$c$	Model wing chord = 2·40 in. (60·96 mm)
$l_v$	Effective dihedral = $(\partial C_I/\partial \beta) \alpha$ , $M = \text{const.}$
$M$	Free stream Mach number
$n_v$	Weathercock stability derivative = $(\partial C_n/\partial \beta) \alpha$ , $M = \text{const.}$
$p$	Free stream static pressure
$p_B$	Model base pressure
$R$	Reynolds number based on model wing chord.
$S$	Model nominal wing area = 27·36 in <sup>2</sup> (17651 mm <sup>2</sup> )
$S_B$	Model base area = 0·83 in <sup>2</sup> (535·5 mm <sup>2</sup> )
$S_T$	Model nominal tailplane area = 5·265 in <sup>2</sup> (3396 mm <sup>2</sup> )
$v$	Free stream velocity
$\alpha$	Angle of incidence: the angle between the wind vector and its projection in the chordal plane.
$\alpha_1$	Alternative angle of incidence: the angle between the model axis and the projection of the wind vector on the model plane of symmetry.
$\alpha_T$	Tailplane angle of incidence.
$\beta$	Angle of sideslip: the angle between the wind vector and its projection on the model plane of symmetry.
$\eta$	Elevator angle with respect to tail plane chord.
$\eta$	Elevator angle to trim ( $C_m = 0$ ).

$\eta_T$	Tailplane angle with respect to fuselage reference line.
$\bar{\eta}_T$	Tailplane angle to trim ( $C_m = 0$ ).
$\rho$	Free stream density.
$\phi$	Model roll angle.

Notes:

- (i) See Figure 1 for sign conventions for forces and moments and attitude angles.
- (ii) The nominal aircraft centre of gravity was taken to be located at  $0.2 c$  and  $0.1125$  in.  
( $2.8575$  mm) (model scale) below the fuselage reference line.
- (iii) Since the full scale aircraft and the model were both manufactured before the introduction of SI units, all dimensions have been expressed in feet and inches, with the equivalent SI unit following in brackets where appropriate.

## 1. INTRODUCTION

For a particular application of the Jindivik target aircraft it was proposed that a flare pack be mounted on the lower rear fuselage of some aircraft. A comparison of the lateral and longitudinal stability and tailplane and elevator effectiveness with and without the flare pack fitted was requested by the Government Aircraft Factory and a wind tunnel test programme proposed. A 1/20th scale complete model of the Jindivik 203B was available and a scale model of the flare pack was manufactured and mounted on this model. The present note reports six-component force and moment measurements conducted in the transonic wind tunnel of the Aeronautical Research Laboratories during August and September 1979.

## 2. TEST DETAILS

### 2.1 Model

Throughout the test programme a 1/20th scale complete metal model of the Jindivik 203B aircraft was used. The major features of this configuration were as follows: short wing with Mk 8 fuel pods (with fins), nominal  $+1^\circ$  twisted flap, fixed aileron drooped at  $+1\frac{1}{2}^\circ$ . No intake ducting was represented. An unsurfaced landing skid was represented in the stowed position on the underside of the fuselage. For all tests except those for determining tailplane and elevator effectiveness, the tailplane angle was set to the current standard value ( $\eta_T = -\frac{1}{2}^\circ$ ) with the elevator undeflected. The main dimensions of the model and full-scale aircraft are given in Table 1 and a sketch of the aircraft is presented in Figure 2.

The flare pack was mounted as shown in Figure 3 with its centreline parallel to the fuselage reference line. This figure also gives the dimensions of the flare pack which were as specified by the Government Aircraft Factories. Photographs of the model with the flare pack in place are presented in Figures 4 and 5.

Owing to the low Reynolds number of the tests (approximately  $0.45 \times 10^6$ ) boundary layer transition was fixed on all windswept surfaces by bands distributed of roughness. These roughness bands were placed on the upper and lower surfaces of the wing, tailplane, fin, fuselage, pods and pod fins and consisted of approximately 3 mm wide bands made up of particles 0.15 mm diameter with a coverage of 10–20%.

### 2.2 Wind Tunnel

The tests were conducted in the transonic wind tunnel of these laboratories. The nominal dimensions of the tunnel test section are 0.81 m by 0.53 m. For these tests, all test section walls were longitudinally slotted (Fig. 6) with an open area ratio at the model location of 10.5%.

The maximum frontal cross-sectional area of the model at zero incidence was 5.36 in<sup>2</sup> (3458 mm<sup>2</sup>) giving a blockage ratio of 0.81%. Since the range of incidence of the tests was limited, no tunnel interference corrections were applied to the results.

Mach number and dynamic pressure were derived from measurements of the static pressure in the plenum chamber surrounding the test section and of the static pressure in the contraction entry, assuming these to be the static and total pressures respectively of the test section flow.

### 2.3 Test Programme

Six component force and moment coefficient were measured for both the clean aircraft and with the flare pack in place. The tests covered a range of incidence of  $-3^\circ \leq \alpha \leq 7^\circ$  (at zero sideslip) and a sideslip range of  $0^\circ \leq \beta \leq 4^\circ$  (at zero incidence). Model attitude was corrected

for sting and balance deflections under load. The range of Mach number tested was  $0.5 \leq m \leq 0.90$  in eight steps.

In addition, tests were conducted to determine the effect of the flare pack on tailplane and elevator effectiveness. These tests covered seven tailplane angles in the range  $-2\frac{1}{2}^\circ \leq \eta_T \leq +3\frac{1}{2}^\circ$  (for  $-3^\circ \leq \alpha \leq 7^\circ$ ,  $\beta = 0^\circ$ ,  $\eta = 0^\circ$ ) and six elevator angles in the range  $-15^\circ \leq \eta \leq +10^\circ$  (for  $0^\circ \leq \alpha \leq 6^\circ$ ,  $\beta = 0^\circ$ ,  $\eta_T = +\frac{1}{2}^\circ$ ).

Tailplane and elevator angles referred to in the tabulated results in Tables 2 and 3 are nominal angles: the corresponding measured angles are tabulated below.

Nominal		Measured		Nominal		Measured	
$\eta_T$	$\eta$	$\eta_T$	$\eta$	$\eta$	$\eta_T$	$\eta$	$\eta_T$
$-2\frac{1}{2}^\circ$	$0^\circ$	$-2.52^\circ$	$0.00^\circ$	$-15^\circ$	$\frac{1}{2}^\circ$	$-17.45^\circ$	$0.50^\circ$
$-1\frac{1}{2}^\circ$	$0^\circ$	$-1.42^\circ$	$0.00^\circ$	$-10^\circ$	$\frac{1}{2}^\circ$	$-10.25^\circ$	$0.50^\circ$
$-\frac{1}{2}^\circ$	$0^\circ$	$-0.50^\circ$	$0.00^\circ$	$-5^\circ$	$\frac{1}{2}^\circ$	$-4.50$	$0.50^\circ$
$\frac{1}{2}^\circ$	$0^\circ$	$0.50^\circ$	$0.00^\circ$	$0^\circ$	$\frac{1}{2}^\circ$	$0.00^\circ$	$0.51^\circ$
$1\frac{1}{2}^\circ$	$0^\circ$	$1.78^\circ$	$0.00^\circ$	$5^\circ$	$\frac{1}{2}^\circ$	$5.50^\circ$	$0.52^\circ$
$2\frac{1}{2}^\circ$	$0^\circ$	$2.68^\circ$	$0.00^\circ$	$10^\circ$	$\frac{1}{2}^\circ$	$10.25^\circ$	$0.44^\circ$
$3\frac{1}{2}^\circ$	$0^\circ$	$3.90^\circ$	$0.00^\circ$				

Measurement accuracy for the above angles was  $\pm 0.02^\circ$  for  $\eta_T$  and  $\pm 0.05^\circ$  for  $\eta$ .

For all tests, the Reynolds number (based on wing chord) was kept approximately constant at  $0.45 \pm 0.03 \times 10^6$  by varying tunnel pressure.

### 3. RESULTS AND DISCUSSION

#### 3.1 Longitudinal Stability

Figures 7 and 8 show the effect of the flare pack on the variation of lift coefficient with incidence, and on the variation of lift curve slope with Mach number. In both cases, the effect of the flare pack is extremely small. The effect on pitching moment is shown in Figure 9 where pitching moment coefficient is plotted against lift coefficient and in Figure 10 where pitching moment coefficient is plotted against Mach number. A slight increase in nose up pitching moment with the flare pack in place is evident in both these figures. This increase is probably associated with a decrease in tail lift due to interference from the flare pack. Drag coefficient is plotted against lift coefficient in Figure 11 and against Mach number in Figure 12. As would be expected the flare pack gives rise to a small increment in drag coefficient. This increment is always less than 0.005 and the major effect of the flare pack is to produce a small "drag creep" just before the drag rise Mach number.

#### 3.2 Lateral Stability

The effect of the flare pack on yawing moment coefficient and on the weathercock stability derivative is shown in Figures 13 and 14. The increase in lateral area contributed by the flare pack leads to a small increase in weathercock stability derivative ( $\partial C_w / \partial \beta$ ) giving a slightly greater stability margin throughout the tested range of Mach number. The increase in lateral area also accounts for the increased slopes of the sideforce coefficient versus sideslip curves presented in Figure 15, but once again the effect is quite small. As would be expected, the effect of the flare pack on the variation of rolling moment coefficient with sideslip and of effective dihedral ( $\partial C_l / \partial \beta$ ) with Mach number plotted in Figures 16 and 17 is virtually nonexistent.

### 3.3 Tailplane and Elevator Effectiveness

The effect of the flare pack on tailplane and elevator effectiveness is obtained from comparisons with curves derived for the clean aircraft in Reference 1. The slight increase in nose up pitching moment caused by the flare pack is reflected in the curves of tailplane angle to trim ( $C_m = 0$ ) and tailplane effectiveness plotted against Mach number in Figures 18 and 19. Although there is significant scatter in the results, the general trend is for an increase in tailplane angle to trim and a decrease in tailplane effectiveness due to the addition of the flare pack, although once again the effects are very small. No such trends are apparent in Figures 20 and 21 in which elevator angle to trim and elevator effectiveness are plotted against Mach number. In this case the curves with the flare pack fitted are very close to those for the clean aircraft.

## 4. CONCLUSIONS

Transonic wind tunnel tests have been carried out to determine the effect of the addition of a flare pack to the Jindivik 203B aircraft on its lateral and longitudinal stability and tailplane and elevator effectiveness. The tests were conducted on a 1/20th scale model equipped with a six-component strain gauge balance. The tests covered angles of incidence from  $-3^\circ$  to  $7^\circ$  and angles of sideslip from  $0^\circ$  to  $4^\circ$  for Mach numbers between 0.5 and 0.9. In general, the effect of the flare pack was found to be very small, the only significant changes being to pitching moment, drag and yawing moment. The effect on pitching moment was to produce a slight increase in nose up pitch throughout the test range. This increase, which is evidently associated with a decrease in tailplane lift due to interference from the flare pack, was also reflected in a small increase in tailplane angle to trim and a corresponding decrease in tailplane effectiveness. The effect on drag was to produce a small drag increment (less than 0.005) with the major effect being a small "drag creep" before the compressibility drag rise. Due to the increase in lateral area, the effect of the flare pack on yawing moment was an increase in the weathercock stability derivative.

None of these effects produced a significant degradation in the overall stability of the aircraft and hence, on the basis of these tests, the addition of the flare pack should not significantly affect the performance of the aircraft.

#### **REFERENCES**

1. Fairlie, B. D. Transonic wind tunnel measurements of the tailplane and elevator effectiveness of the Jindivik target aircraft.  
Aeronautical Research Laboratories, Tech. Note, Aero. 400, 1980.

**TABLE 1**  
**Main Dimensions of Model and Full Scale Aircraft**

	Model Scale	Full Scale
<b>(1) Wing</b>		
Chord	2.40 in.	48.00 in.
Span (nominal)	11.40 in.	228.00 in.
To centreline of pods	11.82 in.	236.48 in.
Gross wing area	27.36 in. <sup>2</sup>	76.0 ft. <sup>2</sup>
Wing section		NACA 64-106 (modified)
Aspect ratio	4.75	
Taper ratio	1.0	
Leading edge sweep	0°	
Trailing edge sweep	0°	
Dihedral	2.5°	
Incidence relative to F.R.L.	1.0°	
<b>(2) Flaps</b>		
Chord	0.60 in.	12.00 in.
Span (per side)	3.45 in.	68.95 in.
Area (per side)	2.07 in. <sup>2</sup>	827.4 in. <sup>2</sup>
Distance from inboard end to aircraft datum	0.79 in.	15.75 in.
Neutral   Inboard	+2°	
Setting   Outboard	0°	
Nominal setting	+1°	
<b>(3) Ailerons</b>		
Chord	0.60 in.	12.00 in.
Span (per side)	1.35 in.	27.00 in.
Area (per side)	0.81 in. <sup>2</sup>	324.0 in. <sup>2</sup>
Distance from inboard end to aircraft datum	4.25 in.	84.96 in.
Neutral setting	0°	
<b>(4) MK.8 Pods</b>		
Overall length	7.22 in.	144.33 in.
Diameter	0.625 in.	12.50 in.
Distance from pod centreline to F.R.L.	5.91 in.	118.24 in.
Distance from pod nose to 25% wing chord	3.70 in.	74.09 in.
<b>(5) Pod Fins</b>		
Distance from pod nose to fin trailing edge	6.25 in.	125.00 in.
Root chord	1.20 in.	24.00 in.
Tip chord	0.60 in.	12.00 in.
Maximum height above pod centreline	0.91 in.	18.25 in.
Angle between fin and F.R.L.	25°	
Aerofoil section		NACA 64-008 (modified)
<b>(6) Tailplane</b>		
Chord	1.35 in.	27.00 in.
Span	3.90 in.	78.00 in.
Gross area	5.265 in. <sup>2</sup>	2106.0 in. <sup>2</sup>
Aspect ratio	2.89	
Taper ratio	1.0	
Sweep at quarter chord	0°	
Incidence relative to F.R.L.	-1°	-0°27'
Aerofoil section		NACA 64-006

**TABLE 1 (Continued)**  
**Main Dimensions of Model and Full Scale Aircraft**

	Model Scale	Full Scale
(7) Elevators		
Chord	0·45 in.	9·00 in.
Span (per side)	1·63 in.	32·64 in.
Area (per side)	0·73 in. <sup>2</sup>	293·4 in. <sup>2</sup>
Distance from inboard end to F.R.L.	0·28 in.	5·55 in.
(8) Fin		
Maximum height above F.R.L.	2·64 in.	52·80 in.
Gross area above tailplane chord	2·35 in. <sup>2</sup>	938·9 in. <sup>2</sup>
Sweep back of leading edge		$14\frac{1}{2}^{\circ}$
Sweep back of trailing edge		$0^{\circ}$
Tip chord	1·20 in.	24·00 in.
Root chord	1·60 in.	32·00 in.
Aerofoil section		(NACA 64-006 (modified))
(9) Fuselage		
Overall length (excluding pitot probe)	13·99 in.	279·75 in.
Maximum height excluding skid beam	1·90 in.	38·04 in.
Maximum width	1·65 in.	33·00 in.
Distance of 25% wing chord aft of STN 'O'	6·61 in.	132·25 in.
Distance of 25% wing chord below F.R.L.	0·41 in.	8·25 in.
Distance of 25% tailplane chord aft of STN 'O'	12·27 in.	245·45 in.
Distance of 25% tailplane chord above F.R.L.	0·86 in.	17·24 in.
Distance of fin trailing edge aft of STN 'O'	13·17 in.	263·37 in.

**INDEX TO TABLES 2 AND 3**

Table No.	Flare pack	$\eta$	$\eta_T$	$\alpha$	$\beta$
2	Off	0°	-1°	-3° to +7°	0°
	Off	0°	-1°	0°	0° to +4°
3A	On	0°	-1°	-3° to +7°	0°
	On	0°	-1°	0°	0° to +4°
3B	On	0°	-21°	-3° to +7°	0°
3C	On	0°	-11°	-3° to +7°	0°
3D	On	0°	1°	-3° to +7°	0°
3E	On	0°	11°	-3° to +7°	0°
3F	On	0°	21°	-3° to +7°	0°
3G	On	0°	31°	-3° to +7°	0°
3H	On	-15°	2°	0° to 6°	0°
3I	On	-10°	1°	0° to 6°	0°
SJ	On	-5°	1°	0° to 6°	0°
3K	On	0°	1°	0° to 6°	0°
3L	On	5°	1°	0° to 6°	0°
3M	On	10°	1°	0° to 6°	0°

### NOTATION FOR TABLES 2 AND 3

The following notation refers to the computer generated data listings in Tables 2 and 3. Where appropriate, the corresponding notation from the main body of the note is also included

Table 2 Notation	Main Body Notation	Explanation
SER	—	Serial number
REYN	$R$	Reynold's number
MACH	$M$	Free stream Mach number
INCID.	$\alpha$	Angle of incidence
LIFT.	$C_L$	Lift coefficient
PITCH	$C_T$	Pitching moment coefficient
DRAG	$C_D$	Drag force coefficient
NORMAL	$C_Z$	Normal force coefficient
AXIAL	$C_X$	Axial force coefficient
CLSQ.	$C_L^2$	Lift coefficient squared
BASE	$C_{XB}$	Base force coefficient
AINC	$\alpha_1$	Alternative angle of incidence
SLIP	$\beta$	Angle of sideslip
CROSS	$C_C$	Cross wind force coefficient
YAW M.	$C_n$	Yawing moment coefficient
ROLL M.	$C_l$	Rolling moment coefficient
RANG	$\phi$	Roll angle
SIDE F	$C_Y$	Side force coefficient

**TABLE 2**  
**Tabulated Results**  
**Clean Aircraft**

TABLE 2  
CLEAN AIRCRAFT  
ETA = 6° ETRACT = -1.72°

SER	REVN.	INCH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLEA.	BASE.	ATMC.	SLIP.	CROSS.	YAW H.	ROLL H.	RANG.	SIDE F
090	0.435	0.904	07.43	0.8750	-0.1207	0.1727	-0.8901	-0.0558	0.7684	0.0023	07.43	00.00	-0.0018	-0.0010	-0.0009	179.9	0.0017
091	0.435	0.904	06.91	0.8389	-0.1130	0.1571	-0.8518	-0.0529	0.7037	0.0022	06.91	00.00	-0.0016	-0.0013	-0.0005	179.9	0.0015
092	0.435	0.904	06.38	0.7976	-0.1126	0.1429	-0.8097	-0.0509	0.6361	0.0024	06.38	00.00	-0.0013	-0.0012	-0.0009	179.9	0.0012
094	0.435	0.904	06.87	0.7534	-0.1047	0.1453	-0.7655	-0.0518	0.5675	0.0024	06.87	00.00	-0.0014	-0.0011	-0.0006	179.9	0.0013
095	0.435	0.904	05.87	0.7566	-0.1061	0.1300	-0.7661	-0.0496	0.5724	0.0024	05.87	00.00	-0.0016	-0.0013	-0.0005	179.9	0.0013
096	0.439	0.902	05.39	0.7210	-0.1008	0.1207	-0.7292	-0.0506	0.5397	0.0024	05.39	00.00	-0.0012	-0.0014	-0.0008	179.9	-0.0003
097	0.435	0.900	04.31	0.6337	-0.0847	0.0939	-0.6415	-0.0497	0.4040	0.0022	04.31	00.00	-0.0001	-0.0000	0.0004	180.0	0.0000
098	0.435	0.900	03.26	0.5447	-0.0761	0.0849	-0.5687	-0.0614	0.2965	0.0024	03.26	00.00	0.0008	-0.0004	0.0010	180.0	-0.0009
099	0.439	0.904	01.14	0.3151	-0.0525	0.0598	-0.3163	-0.0517	0.0992	0.0019	01.14	00.00	0.0007	-0.0003	0.0014	180.0	-0.0008
100	0.476	0.900	-00.39	0.0434	-0.0127	0.0494	-0.0426	-0.0483	0.0018	0.0019	-00.39	00.00	0.0030	0.0003	0.0018	180.0	-0.0031
101	0.429	0.900	-03.14	-0.2628	0.0359	0.0574	-0.2654	-0.0415	0.0683	0.0018	-03.14	-00.01	0.0036	0.0004	0.0022	180.0	-0.0037
102	0.432	0.860	07.13	0.8369	-0.0913	0.1570	-0.8521	-0.0539	0.7036	0.0021	07.13	00.00	-0.0016	-0.0010	0.0010	180.0	0.0018
103	0.432	0.860	06.91	0.8241	-0.1026	0.1493	-0.8362	-0.0471	0.6791	0.0020	06.91	00.00	-0.0009	-0.0013	-0.0012	179.9	0.0008
104	0.429	0.878	06.39	0.7904	-0.0975	0.1365	-0.8008	-0.0456	0.6246	0.0024	06.39	00.00	-0.0019	-0.0015	-0.0016	179.9	0.0018
105	0.429	0.879	05.87	0.7535	-0.0993	0.1233	-0.7621	-0.0434	0.5677	0.0024	05.87	00.00	-0.0013	-0.0011	-0.0007	179.9	0.0012
106	0.432	0.861	05.35	0.7126	-0.0835	0.1132	-0.7202	-0.0441	0.5077	0.0026	05.35	00.00	-0.0013	-0.0012	-0.0005	179.9	0.0010
107	0.432	0.879	04.32	0.6342	-0.0649	0.0937	-0.6396	-0.0436	0.4021	0.0020	04.32	00.00	-0.0014	-0.0010	-0.0001	179.9	0.0012
108	0.432	0.864	03.20	0.5847	-0.0561	0.0721	-0.5918	-0.0437	0.2999	0.0020	03.20	00.00	0.0002	-0.0004	0.0010	180.0	-0.0003
109	0.432	0.879	01.47	0.3393	-0.0351	0.0550	-0.3404	-0.0462	0.1150	0.0048	01.47	00.00	0.0008	0.0000	0.0012	180.0	-0.0009
110	0.432	0.879	-00.97	0.0671	-0.0404	0.0436	-0.0684	-0.0434	0.0014	0.0017	-00.97	00.00	0.0022	0.0004	0.0016	180.0	-0.0023
111	0.432	0.860	-02.13	-0.2571	0.0340	0.0518	-0.2595	-0.0362	0.0689	0.0016	-02.13	-00.01	0.0042	0.0007	0.0018	180.0	-0.0043
114	0.442	0.859	07.40	0.8177	-0.1027	0.1519	-0.8306	-0.0432	0.6685	0.0020	07.40	00.00	-0.0013	-0.0013	-0.0035	180.0	0.0012
115	0.444	0.861	06.91	0.8118	-0.0957	0.1411	-0.8230	-0.0406	0.6588	0.0018	06.91	00.00	-0.0019	-0.0013	0.0004	180.0	0.0018
116	0.440	0.862	06.40	0.7922	-0.0875	0.1228	-0.8022	-0.0416	0.6291	0.0019	06.40	00.00	-0.0021	-0.0014	-0.0012	179.9	0.0020
117	0.442	0.859	05.89	0.7534	-0.0785	0.1193	-0.7678	-0.0389	0.5767	0.0019	05.89	00.00	-0.0014	-0.0014	-0.0006	179.9	0.0013
118	0.440	0.860	05.37	0.7136	-0.0704	0.1066	-0.7267	-0.0367	0.5181	0.0020	05.37	00.00	-0.0011	-0.0013	-0.0002	179.9	0.0010
119	0.440	0.859	04.34	0.6450	-0.0538	0.0866	-0.6500	-0.0374	0.4159	0.0024	04.34	00.00	0.0002	-0.0009	0.0003	180.0	-0.0003
120	0.442	0.859	03.20	0.5604	-0.0397	0.0737	-0.5628	-0.0394	0.3139	0.0019	03.20	00.00	0.0008	-0.0005	0.0013	180.0	-0.0009
121	0.440	0.861	01.19	0.3494	-0.0191	0.0510	-0.3505	-0.0421	0.1220	0.0017	01.19	00.00	0.0015	0.0002	0.0017	180.0	-0.0016
122	0.440	0.860	-00.96	0.0645	0.0076	0.0395	-0.0639	-0.0391	0.0040	0.0016	-00.96	00.00	0.0008	-0.0004	0.0018	180.0	-0.0009
123	0.442	0.859	-03.13	-0.2406	0.0283	0.0489	-0.2428	-0.0310	0.0578	0.0014	-03.13	-00.01	0.0028	0.0003	0.0021	180.0	-0.0029
124	0.437	0.840	07.39	0.8160	-0.1032	0.1468	-0.8282	-0.0387	0.6688	0.0019	07.39	00.00	-0.0014	-0.0011	0.0005	179.9	0.0012
125	0.437	0.840	06.50	0.7892	-0.0786	0.1383	-0.7399	-0.0376	0.6228	0.0018	06.50	00.00	-0.0008	-0.0005	0.0022	180.0	0.0010
126	0.437	0.841	06.40	0.7796	-0.0721	0.1239	-0.7887	-0.0344	0.6077	0.0016	06.40	00.00	-0.0004	-0.0012	0.0017	180.0	0.0007
127	0.437	0.840	05.89	0.7693	-0.0639	0.1179	-0.7774	-0.0362	0.5918	0.0019	05.89	00.00	-0.0022	-0.0014	-0.0010	179.9	0.0011
128	0.437	0.840	05.38	0.7303	-0.0391	0.1036	-0.7371	-0.0328	0.3336	0.0018	05.38	00.00	-0.0013	-0.0010	0.0000	179.9	0.0012
129	0.434	0.841	04.35	0.6567	-0.0424	0.0957	-0.6613	-0.0326	0.4312	0.0020	04.35	00.00	-0.0007	-0.0011	0.0005	180.0	0.0006
130	0.434	0.839	03.31	0.5736	-0.0270	0.0700	-0.5768	-0.0349	0.3288	0.0018	03.31	00.00	-0.0003	-0.0013	0.0005	180.0	0.0008
131	0.434	0.841	01.20	0.3456	-0.0047	0.0469	-0.3466	-0.0380	0.1193	0.0017	01.20	-00.01	0.0015	0.0003	0.0017	180.0	-0.0016

TABLE 2  
CLEAN AIRCRAFT  
 $\text{ETA} = 0^\circ \text{ ETRACT} = -1/2^\circ$

SER	REVN.	HINCH.	INCID.	LIFT.	PITCH.	DRAE	NORML.	AXIAL.	Clsa.	Base.	Ainc.	Slip.	Cross.	Van N.	Roll N.	Kang.	Side F.
132	0.437	0.840	-00. 96	0. 0503	0. 0146	0. 0378	-0. 0377	0. 0373	0. 0023	0. 0045	-00. 96	-00. 04	0. 0040	0. 0002	0. 0046	180. 0	-0. 0011
133	0.437	0.840	-03. 12	-0. 2222	0. 0264	0. 0129	0. 2141	-0. 0180	0. 0537	0. 0013	-03. 12	-00. 01	0. 0023	0. 0003	0. 0019	180. 0	-0. 0024
136	0.455	0.802	07. 42	0. 8106	-0. 0717	0. 1412	-0. 0220	0. 0123	0. 6568	0. 0020	07. 42	10. 00	0. 0000	-0. 0007	0. 0026	180. 0	-0. 0003
137	0.455	0.800	06. 91	0. 8049	-0. 0787	0. 1306	-0. 0249	0. 0210	0. 6477	0. 0018	06. 91	10. 00	0. 0000	-0. 0007	0. 0021	180. 0	-0. 0003
138	0.455	0.804	06. 41	0. 7867	-0. 0620	0. 1200	-0. 7953	-0. 0295	0. 6187	0. 0016	06. 41	10. 00	-0. 0003	-0. 0007	0. 0022	180. 0	0. 0003
139	0.455	0.800	05. 92	0. 7597	-0. 0243	0. 1064	-0. 7628	-0. 0258	0. 5711	0. 0019	05. 92	10. 00	0. 0002	-0. 0006	0. 0020	180. 0	-0. 0003
140	0.455	0.803	05. 42	0. 7536	-0. 0390	0. 1003	-0. 7600	-0. 0266	0. 5681	0. 0020	05. 42	10. 00	-0. 0019	-0. 0013	0. 0021	180. 0	0. 0018
141	0.455	0.801	04. 39	0. 6666	-0. 0171	0. 0801	-0. 6398	-0. 0253	0. 4713	0. 0020	04. 39	10. 00	0. 0002	-0. 0004	0. 0016	180. 0	-0. 0003
142	0.455	0.800	03. 25	0. 5819	0. 0008	0. 0613	-0. 5845	-0. 0255	0. 3385	0. 0017	03. 25	10. 00	0. 0004	-0. 0003	0. 0023	180. 0	-0. 0009
143	0.455	0.799	04. 20	0. 3163	0. 0109	0. 0424	-0. 3972	-0. 0344	0. 1000	0. 0014	04. 20	10. 00	0. 0016	0. 0000	0. 0018	180. 0	-0. 0017
144	0.455	0.799	-00. 96	0. 0521	0. 0197	0. 0181	-0. 0516	-0. 0177	0. 0027	0. 0013	-00. 96	-10. 04	0. 0003	0. 0001	0. 0018	180. 0	-0. 0010
145	0.455	0.799	-03. 11	-0. 2125	0. 0297	0. 0411	0. 2143	-0. 0283	0. 0450	0. 0013	-03. 11	-10. 04	0. 0021	0. 0003	0. 0022	180. 0	-0. 0023
146	0.434	0.750	07. 39	0. 8146	-0. 0529	0. 1350	-0. 8252	-0. 0272	0. 6335	0. 0016	07. 39	10. 00	0. 0016	-0. 0004	0. 0044	180. 0	-0. 0017
147	0.437	0.751	06. 89	0. 8026	-0. 0591	0. 1223	-0. 8117	-0. 0239	0. 6441	0. 0017	06. 89	10. 00	0. 0025	-0. 0006	0. 0044	180. 0	-0. 0026
148	0.434	0.750	06. 39	0. 7853	-0. 0436	0. 1120	-0. 7930	-0. 0222	0. 6166	0. 0017	06. 39	10. 00	0. 0029	-0. 0002	0. 0043	180. 0	-0. 0030
149	0.437	0.751	05. 89	0. 7652	-0. 0211	0. 1021	-0. 7718	-0. 0212	0. 5856	0. 0017	05. 89	10. 00	0. 0024	-0. 0002	0. 0041	180. 0	-0. 0023
150	0.437	0.751	05. 40	0. 7440	-0. 0121	0. 0864	-0. 7493	-0. 0163	0. 3523	0. 0017	05. 40	10. 00	-0. 0006	-0. 0007	0. 0025	180. 0	0. 0003
151	0.434	0.749	04. 37	0. 6647	0. 0095	0. 0686	-0. 6691	-0. 0162	0. 4416	0. 0017	04. 37	10. 00	0. 0001	-0. 0004	0. 0022	180. 0	-0. 0002
152	0.424	0.749	03. 30	0. 5400	0. 0117	0. 0514	-0. 5624	-0. 0216	0. 2915	0. 0016	03. 30	10. 00	0. 0006	-0. 0002	0. 0024	180. 0	-0. 0009
153	0.434	0.750	04. 17	0. 2917	0. 0130	0. 0411	-0. 2326	-0. 0130	0. 0850	0. 0014	01. 17	10. 00	0. 0006	-0. 0001	0. 0017	180. 0	-0. 0007
154	0.437	0.751	-00. 96	0. 0474	0. 0219	0. 0263	-0. 0469	-0. 0157	0. 0021	0. 0014	-00. 96	10. 00	0. 0043	0. 0001	0. 0048	180. 0	-0. 0016
155	0.437	0.751	-03. 09	-0. 1995	0. 0345	0. 0397	0. 2012	-0. 0277	0. 0297	0. 0012	-03. 09	10. 00	0. 0028	0. 0002	0. 0021	180. 0	-0. 0023
156	0.414	0.699	07. 36	0. 8158	-0. 0569	0. 1354	-0. 8057	-0. 0241	0. 6549	0. 0017	07. 36	10. 00	0. 0012	-0. 0002	0. 0038	180. 0	-0. 0013
157	0.414	0.701	06. 86	0. 7969	-0. 0422	0. 1179	-0. 8074	-0. 0198	0. 6382	0. 0018	06. 86	10. 00	0. 0046	0. 0000	0. 0044	180. 0	-0. 0018
158	0.414	0.701	06. 36	0. 7768	-0. 0273	0. 1062	-0. 7858	-0. 0175	0. 6064	0. 0017	06. 36	10. 00	0. 0022	-0. 0001	0. 0045	180. 0	-0. 0023
159	0.417	0.700	05. 86	0. 7531	-0. 0127	0. 0940	-0. 7586	-0. 0148	0. 5671	0. 0017	05. 86	10. 00	0. 0045	-0. 0001	0. 0039	180. 0	-0. 0016
160	0.414	0.700	05. 35	0. 7168	-0. 0014	0. 0826	-0. 7215	-0. 0137	0. 5137	0. 0016	05. 35	10. 00	0. 0012	-0. 0001	0. 0031	180. 0	-0. 0013
161	0.414	0.701	04. 31	0. 6186	0. 0095	0. 0645	-0. 6220	-0. 0161	0. 3629	0. 0017	04. 31	10. 00	0. 0002	0. 0000	0. 0029	180. 0	-0. 0003
162	0.416	0.703	03. 26	0. 5069	0. 0103	0. 0520	-0. 5094	-0. 0214	0. 2568	0. 0017	03. 26	10. 00	0. 0013	0. 0002	0. 0027	180. 0	-0. 0014
163	0.414	0.700	04. 14	0. 2753	0. 0127	0. 0403	-0. 2762	-0. 0134	0. 0757	0. 0014	01. 14	-10. 04	0. 0010	0. 0003	0. 0017	180. 0	-0. 0011
164	0.414	0.701	00. 97	0. 0434	0. 0233	0. 0283	-0. 0426	-0. 0147	0. 0018	0. 0013	-00. 97	10. 00	0. 0019	0. 0000	0. 0017	180. 0	-0. 0020
165	0.414	0.700	-03. 07	-0. 1891	0. 0369	0. 0404	0. 1892	-0. 0287	0. 0286	0. 0013	-03. 07	10. 00	0. 0021	0. 0001	0. 0020	180. 0	-0. 0022
166	0.431	0.199	07. 28	0. 7837	-0. 0317	0. 1159	-0. 7923	-0. 0138	0. 6140	0. 0019	07. 28	10. 00	0. 0000	0. 0001	0. 0026	180. 0	-0. 0004
167	0.431	0.501	06. 78	0. 7528	-0. 0181	0. 1021	-0. 7597	-0. 0107	0. 5666	0. 0016	06. 78	10. 00	-0. 0002	0. 0002	0. 0024	180. 0	0. 0001
170	0.434	0.502	06. 27	0. 7160	-0. 0088	0. 0893	-0. 7215	-0. 0086	0. 5125	0. 0017	06. 27	10. 00	-0. 0001	0. 0002	0. 0024	180. 0	0. 0000
171	0.434	0.500	05. 75	0. 6724	-0. 0032	0. 0782	-0. 6770	-0. 0085	0. 4520	0. 0018	05. 75	-10. 04	-0. 0001	0. 0002	0. 0024	180. 0	0. 0000
172	0.431	0.501	05. 24	0. 6237	0. 006	0. 0649	-0. 6273	-0. 0102	0. 3889	0. 0019	05. 24	-10. 04	0. 0000	0. 0003	0. 0026	180. 0	-0. 0001
173	0.431	0.501	04. 20	0. 5357	0. 0043	0. 0564	-0. 5385	-0. 0154	0. 2868	0. 0017	04. 20	-10. 04	0. 0000	0. 0003	0. 0030	180. 0	-0. 0009

TABLE 2  
CLEAN AIRCRAFT  
ETR = 0° EXTRACTS = -4/2°

SER NO.	REV.	MACH.	LIFT	PITCH	DRAE	NORMAL	AXIAL	CLSA.	BASE.	INC.	SLIP.	CROSS.	VAN	ROLL M.	RANG.	SIDE F
174	0.	0.431	0.300	02.17	0.4428	0.0070	0.0471	-0.4448	-0.0209	0.1939	0.0027	03.17	-10.04	0.0008	0.0003	0.0026
175	0.	0.431	0.300	01.10	0.2429	0.0146	0.0371	-0.2437	-0.0309	0.0589	0.0016	01.10	10.00	0.0015	0.0019	180.0
176	0.	0.431	0.499	-00.98	0.0262	0.0256	0.0340	-0.0358	"0.0332	0.0012	0.0014	-00.98	(10.00)	0.0020	0.0021	180.0
177	0.	0.431	0.301	-03.05	-0.1664	0.0392	0.0372	0.1681	-0.0271	0.0275	0.0013	-03.05	(10.00)	0.0022	0.0021	180.0

TABLE 2  
CLEAR AIRCRAFT I  
ETA = 0° EFACT = -1/2"

SER. NO.	REYN.	WIND.	INCID.	LIFT.	PITCH.	DRAE	NORML.	AXIL.	CL.SQ.	BASE.	FLNC.	SLIP.	CROSS.	VAN. M.	ROLL M.	RANG.	SIDE F.	
180	0.434	0.859	00.08	0.2106	-0.0445	0.0470	-0.2106	-0.0485	0.0442	0.0017	00.08	013.99	-0.0511	-0.0067	0.0015	-0.00	0.0476	
181	0.425	0.901	00.08	0.2053	-0.0395	0.0506	-0.2054	-0.0492	0.0420	0.0020	00.08	012.00	-0.0252	-0.0035	0.0024	-0.00	0.0231	
182	0.434	0.859	00.09	0.2057	-0.0380	0.0519	-0.2059	-0.0496	0.0422	0.0020	00.09	011.00	-0.0125	-0.0020	0.0016	-0.00	0.0115	
183	0.434	0.900	00.09	0.2079	-0.0381	0.0508	-0.2081	-0.0486	0.0431	0.0019	00.09	010.00	-0.0006	-0.0008	0.0014	-0.00	0.0005	
184	0.429	0.861	00.10	0.2289	-0.0325	0.0450	-0.2291	-0.0462	0.0523	0.0016	00.10	013.99	-0.0505	-0.0069	0.0040	-0.00	0.0471	
185	0.429	0.860	00.11	0.2287	-0.0230	0.0470	-0.2289	-0.0456	0.0522	0.0016	00.11	012.00	-0.0244	-0.0034	0.0027	-0.00	0.0227	
186	0.429	0.861	00.11	0.2286	-0.0235	0.0484	-0.2290	-0.0463	0.0522	0.0016	00.11	011.00	-0.0122	-0.0020	0.0019	-0.00	0.0113	
187	0.427	0.879	00.11	0.2304	-0.0217	0.0479	-0.2306	-0.0457	0.0530	0.0016	00.11	010.00	-0.0003	-0.0007	0.0013	-0.00	0.0002	
188	0.439	0.860	00.12	0.2294	-0.0108	0.0399	-0.2295	-0.0412	0.0525	0.0017	00.12	013.99	-0.0501	-0.0066	0.0047	-0.00	0.0471	
189	0.439	0.861	00.12	0.2301	-0.0073	0.0437	-0.2304	-0.0420	0.0528	0.0020	00.12	011.99	-0.0241	-0.0033	0.0029	-0.00	0.0225	
190	0.439	0.859	00.13	0.2297	-0.0033	0.0438	-0.2299	-0.0416	0.0526	0.0018	00.13	011.00	-0.0120	-0.0019	0.0022	-0.00	0.0111	
191	0.439	0.861	00.12	0.2317	-0.0058	0.0444	-0.2319	-0.0421	0.0536	0.0018	00.12	010.00	-0.0004	-0.0006	0.0014	-0.00	0.0003	
192	0.439	0.859	00.13	0.2317	-0.012	0.0274	-0.2169	-0.0387	0.0468	0.0015	00.12	011.99	-0.0492	-0.0065	0.0049	-0.00	0.0161	
193	0.439	0.861	00.12	0.2166	0.0018	0.0274	-0.2168	-0.0387	0.0468	0.0015	00.12	011.99	-0.0242	-0.0034	0.0030	-0.00	0.0227	
194	0.434	0.842	00.12	0.2166	0.0018	0.0062	0.0404	-0.2171	-0.0387	0.0469	0.0017	00.13	011.99	-0.0122	-0.0020	0.0022	-0.00	0.0114
195	0.431	0.841	00.13	0.2169	0.0018	0.0062	0.0404	-0.2192	-0.0384	0.0479	0.0017	00.13	011.00	-0.0122	-0.0020	0.0022	-0.00	0.0114
196	0.434	0.840	00.13	0.2190	0.0018	0.0066	0.0404	-0.2192	-0.0384	0.0479	0.0017	00.13	011.00	-0.0122	-0.0020	0.0022	-0.00	0.0114
197	0.431	0.841	00.13	0.2202	0.0061	0.0420	-0.2204	-0.0388	0.0484	0.0017	00.13	010.00	-0.0003	-0.0007	0.0013	-0.00	0.0004	
198	0.438	0.804	00.12	0.1978	0.0085	0.0382	-0.1980	-0.0368	0.0390	0.0013	00.12	013.99	-0.0492	-0.0063	0.0046	-0.00	0.0465	
199	0.435	0.804	00.12	0.1972	0.0128	0.0374	-0.1974	-0.0362	0.0386	0.0015	00.12	011.99	-0.0239	-0.0023	0.0029	-0.00	0.0225	
200	0.438	0.804	00.12	0.2004	0.0132	0.0281	-0.2006	-0.0363	0.0400	0.0016	00.12	011.00	-0.0123	-0.0020	0.0022	-0.00	0.0116	
201	0.435	0.804	00.12	0.2004	0.0132	0.0281	-0.2006	-0.0363	0.0400	0.0016	00.12	011.00	-0.0123	-0.0020	0.0022	-0.00	0.0116	
202	0.438	0.803	00.13	0.2014	0.0137	0.0286	-0.2016	-0.0365	0.0404	0.0017	00.13	010.00	-0.0003	-0.0007	0.0014	-0.00	0.0004	
203	0.456	0.804	00.13	0.2014	0.0137	0.0286	-0.2016	-0.0365	0.0404	0.0017	00.13	010.00	-0.0003	-0.0007	0.0014	-0.00	0.0004	
204	0.429	0.751	00.14	0.1841	0.0108	0.0340	-0.1843	-0.0356	0.0338	0.0014	00.14	013.99	-0.0483	-0.0060	0.0045	-0.00	0.0437	
205	0.440	0.751	00.11	0.2841	0.0448	0.0363	-0.1843	-0.0353	0.0338	0.0013	00.11	011.99	-0.0240	-0.0023	0.0026	-0.00	0.0227	
206	0.437	0.749	00.11	0.1841	0.0197	0.0379	-0.1843	-0.0362	0.0346	0.0015	00.11	011.99	-0.0241	-0.0020	0.0021	-0.00	0.0114	
207	0.437	0.748	00.11	0.1849	0.0162	0.0387	-0.1851	-0.0369	0.0348	0.0015	00.11	010.00	-0.0006	-0.0007	0.0014	-0.00	0.0005	
208	0.416	0.704	00.09	0.1729	0.0121	0.0332	-0.1740	-0.0148	0.0304	0.0014	00.09	013.99	-0.0479	-0.0059	0.0043	-0.00	0.0153	
209	0.416	0.702	00.10	0.1747	0.0198	0.0363	-0.1749	-0.0155	0.0305	0.0016	00.10	014.39	-0.0240	-0.0032	0.0028	-0.00	0.0226	
210	0.416	0.704	00.10	0.1744	0.0165	0.0366	-0.1746	-0.0150	0.0304	0.0016	00.10	011.00	-0.0123	-0.0020	0.0020	-0.00	0.0116	
211	0.437	0.700	00.10	0.1743	0.0169	0.0365	-0.1745	-0.0148	0.0302	0.0016	00.10	010.00	-0.0003	-0.0008	0.0013	-0.00	0.0002	
212	0.434	0.500	00.06	0.1534	0.0440	0.0323	-0.1535	-0.0135	0.0234	0.0018	00.06	013.99	-0.0465	-0.0084	0.0041	-0.00	0.0440	
213	0.434	0.499	00.06	0.1526	0.0472	0.0341	-0.1527	-0.0134	0.0232	0.0017	00.06	011.99	-0.0232	-0.0034	0.0027	-0.00	0.0219	
214	0.437	0.502	00.07	0.1535	0.0417	0.0348	-0.1537	-0.0134	0.0235	0.0017	00.07	014.00	-0.0147	-0.0018	0.0022	-0.00	0.0110	
215	0.434	0.502	00.07	0.1527	0.0484	0.0349	-0.1528	-0.0134	0.0235	0.0017	00.07	010.00	-0.0004	-0.0009	0.0015	-0.00	0.0003	

**TABLE 3**  
**Tabulated Results**  
**Flare Pack On**

TABLE 3(A)  
FLARE PACK ON  
ETA = 0°(EXTRACT) = -1/2°

SER. NO.	REYN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	ANGL.	CLSQ.	BASE.	ANIC.	SLIP.	CROSS.	VAN H.	ROLL H.	RANG.	SIZE F
002	0.442	0.900	07.43	0.8694	-0.1174	0.1724	-0.8845	-0.0555	0.7557	0.0030	07.43	00.00	-0.0018	-0.0008	-0.0010	179.9	0.0014
003	0.442	0.899	06.91	0.8151	-0.1132	0.1569	-0.8480	-0.0522	0.6973	0.0034	06.91	00.00	-0.0011	-0.0006	-0.0006	179.9	0.0010
004	0.440	0.901	06.39	0.7954	-0.1095	0.1443	-0.8067	-0.0517	0.6326	0.0032	06.39	00.00	-0.0010	-0.0012	-0.0009	179.9	0.0008
005	0.440	0.901	05.87	0.7567	-0.1038	0.1329	-0.7664	-0.0517	0.5725	0.0032	05.87	00.00	-0.0013	-0.0007	-0.0007	179.9	0.0014
006	0.442	0.899	05.35	0.7142	-0.0932	0.1202	-0.7224	-0.0499	0.5104	0.0034	05.35	00.00	-0.0007	-0.0010	-0.0008	179.9	0.0006
007	0.442	0.899	04.34	0.6358	-0.0825	0.1019	-0.6418	-0.0508	0.4042	0.0029	04.34	00.00	-0.0010	-0.0008	-0.0003	180.0	0.0009
008	0.440	0.901	02.26	0.5422	-0.0219	0.0841	-0.5462	-0.0502	0.2936	0.0029	02.26	00.00	-0.0003	-0.0004	-0.0010	180.0	-0.0004
009	0.440	0.901	01.14	0.3169	-0.0493	0.0608	-0.3182	-0.0516	0.1003	0.0028	01.14	00.01	-0.0006	-0.0004	-0.0016	180.0	-0.0007
010	0.442	0.899	-00.59	0.0469	-0.0111	0.0499	-0.0462	-0.0480	0.0221	0.0028	00.59	00.01	-0.0018	-0.0003	-0.0018	180.0	-0.0019
011	0.440	0.901	-03.13	-0.2616	0.0383	0.0608	-0.2644	-0.0438	0.0682	0.0026	-03.13	00.01	-0.0029	0.0004	0.0021	180.0	-0.0030
012	0.434	0.879	07.41	0.8344	-0.1016	0.1598	-0.8481	-0.0477	0.6961	0.0031	07.41	00.00	-0.0016	-0.0003	-0.0024	180.0	0.0015
013	0.434	0.880	06.91	0.8221	-0.1017	0.1509	-0.8344	-0.0478	0.6758	0.0034	06.91	00.00	-0.0016	-0.0010	-0.0013	179.9	0.0015
014	0.437	0.882	06.39	0.7890	-0.0963	0.1286	-0.7396	-0.0467	0.6224	0.0032	06.39	00.00	-0.0012	-0.0014	-0.0012	179.9	0.0011
015	0.434	0.881	05.87	0.7507	-0.0873	0.1261	-0.7598	-0.0454	0.5671	0.0032	05.87	00.00	-0.0015	-0.0012	-0.0007	179.9	0.0014
016	0.435	0.879	05.36	0.7096	-0.0779	0.1146	-0.7173	-0.0446	0.5035	0.0032	05.36	00.00	-0.0013	-0.0010	-0.0004	179.9	0.0011
017	0.434	0.881	04.32	0.6134	-0.0641	0.0950	-0.6288	-0.0440	0.404C	0.0029	04.32	00.00	-0.0005	-0.0007	-0.0001	179.9	0.0004
018	0.434	0.879	03.28	0.5474	-0.0527	0.0786	-0.5511	-0.0444	0.2994	0.0029	02.28	00.00	-0.0006	-0.0003	-0.0003	180.0	-0.0003
019	0.434	0.880	01.17	0.3387	-0.0329	0.0563	-0.3398	-0.0466	0.1148	0.0028	01.17	00.01	-0.0013	0.0003	0.0014	180.0	-0.0014
020	0.435	0.878	00.97	0.0673	-0.0045	0.0442	-0.0667	-0.0426	0.0849	0.0026	00.97	00.01	-0.0011	0.0002	0.0018	180.0	-0.0012
021	0.434	0.881	-03.13	-0.2544	0.0358	0.0526	-0.2568	-0.0372	0.0645	0.0026	-03.13	00.01	-0.0032	0.0006	0.0019	180.0	-0.0023
024	0.444	0.860	07.40	0.8146	-0.1024	0.1529	-0.8276	-0.0438	0.6675	0.0029	02.40	00.00	-0.0003	-0.0012	-0.0026	180.0	0.0002
025	0.444	0.861	06.90	0.8119	-0.0972	0.1432	-0.8231	-0.0415	0.6392	0.0030	06.90	00.00	-0.0017	-0.0012	-0.0020	180.0	0.0016
026	0.444	0.860	06.40	0.7911	-0.0819	0.1234	-0.8013	-0.0413	0.6296	0.0029	06.40	00.00	-0.0013	-0.0010	-0.0010	179.9	0.0012
027	0.444	0.860	05.99	0.7577	-0.0761	0.1211	-0.7662	-0.0398	0.5740	0.0029	05.99	00.00	-0.0004	-0.0010	-0.0007	179.9	0.0003
028	0.444	0.860	05.37	0.7180	-0.0666	0.1101	-0.7203	-0.0329	0.5195	0.0029	05.37	00.00	-0.0007	-0.0010	0.0000	179.9	0.0006
029	0.444	0.860	04.34	0.6434	-0.0508	0.0698	-0.6481	-0.0380	0.4134	0.0029	04.34	00.00	-0.0013	-0.0006	-0.0003	180.0	0.0012
030	0.444	0.861	03.30	0.5595	-0.0401	0.0761	-0.5631	-0.0410	0.3129	0.0028	02.30	00.00	-0.0006	-0.0003	0.0011	180.0	-0.0007
031	0.442	0.859	01.19	0.3507	-0.0163	0.0517	-0.3518	-0.0418	0.1229	0.0026	01.19	-00.01	-0.0014	0.0004	0.0018	180.0	-0.0015
032	0.444	0.860	-00.96	0.0671	0.0095	0.0413	-0.0665	-0.0406	0.0044	0.0025	-00.96	-00.01	-0.0017	0.0003	0.0018	180.0	-0.0016
033	0.442	0.859	-03.12	-0.2417	0.0112	0.0475	-0.2429	-0.0321	0.0503	0.0023	-03.12	-00.01	-0.0012	0.0004	0.0021	180.0	-0.0023
034	0.434	0.839	07.40	0.8163	-0.0974	0.1498	-0.8289	-0.0408	0.6662	0.0027	07.40	00.00	-0.0007	-0.0010	-0.0022	180.0	0.0006
035	0.434	0.838	06.90	0.7886	-0.0714	0.1287	-0.7996	-0.0403	0.6218	0.0025	06.90	00.00	-0.0005	-0.0009	0.0021	180.0	0.0004
036	0.437	0.840	06.41	0.7679	-0.0595	0.1266	-0.7772	-0.0374	0.5894	0.0026	06.41	00.00	-0.0008	-0.0008	0.0004	180.0	0.0007
037	0.437	0.829	05.89	0.7669	-0.0668	0.1183	-0.7751	-0.0363	0.5800	0.0026	05.89	00.00	-0.0021	-0.0008	-0.0005	179.9	0.0020
038	0.437	0.841	05.38	0.7108	-0.0572	0.1053	-0.7376	-0.0347	0.5340	0.0026	05.38	00.00	-0.0013	-0.0009	0.0000	179.9	0.0012

TABLE 3(a)  
FLARE PACK ON  
ETA = 0° ETACT = -1/2°

SER	REYN.	INCH.	INCID.	LIFT.	PITCH.	DRAE.	NORMA.	AXIAL.	CLOS.	BASE.	RING.	SLIP.	CROSS.	YAN M.	ROLL M.	RANG.	SIDE F
002	0.444	0.904	00.08	0.2077	-0.0416	0.0478	-0.2078	-0.0490	0.0430	0.0048	00.08	04.00	-0.0324	-0.0074	0.0036	-0.000	0.0489
003	0.447	0.900	00.08	0.2034	-0.0389	0.0524	-0.2033	-0.0510	0.0411	0.0023	00.08	02.00	-0.0262	-0.0019	0.0024	-0.000	0.0442
004	0.444	0.899	00.09	0.2071	-0.0365	0.0521	-0.2072	-0.0500	0.0428	0.0020	00.09	01.00	-0.0229	-0.0022	0.0018	-0.000	0.0419
005	0.442	0.898	00.09	0.2119	-0.0352	0.0517	-0.2121	-0.0495	0.0449	0.0018	00.09	00.00	-0.0004	-0.0007	0.0012	-0.000	0.0403
006	0.437	0.880	00.11	0.2299	-0.0258	0.0443	-0.2304	-0.0457	0.0527	0.0017	00.11	04.00	-0.0537	-0.0073	0.0042	-0.000	0.0484
007	0.437	0.884	00.11	0.2281	-0.0251	0.0450	-0.2282	-0.0474	0.0519	0.0020	00.11	02.00	-0.0257	-0.0037	0.0026	-0.000	0.0428
008	0.437	0.881	00.11	0.2287	-0.0236	0.0483	-0.2289	-0.0463	0.0522	0.0020	00.11	01.00	-0.0129	-0.0021	0.0019	-0.000	0.0420
009	0.440	0.880	00.12	0.2203	-0.0216	0.0493	-0.2203	-0.0470	0.0529	0.0019	00.12	00.00	-0.0005	-0.0007	0.0012	-0.000	0.0404
010	0.444	0.860	00.13	0.2273	-0.0074	0.0395	-0.2275	-0.0408	0.0516	0.0016	00.13	04.00	-0.0512	-0.0072	0.0047	-0.000	0.0483
011	0.444	0.861	00.13	0.2297	-0.0048	0.0441	-0.2299	-0.0426	0.0526	0.0016	00.13	02.00	-0.0247	-0.0036	0.0029	-0.000	0.0424
012	0.444	0.861	00.13	0.2312	-0.0046	0.0455	-0.2314	-0.0434	0.0533	0.0018	00.13	04.00	-0.0426	-0.0021	0.0022	-0.000	0.0417
013	0.444	0.861	00.13	0.2319	-0.0049	0.0447	-0.2322	-0.0424	0.0537	0.0017	00.13	00.00	-0.0004	-0.0006	0.0014	-0.000	0.0403
014	0.444	0.861	00.13	0.2319	-0.0049	0.0447	-0.2322	-0.0424	0.0537	0.0017	00.13	00.00	-0.0004	-0.0006	0.0014	-0.000	0.0403
015	0.444	0.861	00.13	0.2319	-0.0049	0.0447	-0.2322	-0.0424	0.0537	0.0017	00.13	00.00	-0.0004	-0.0006	0.0014	-0.000	0.0403
016	0.437	0.841	00.13	0.2142	-0.0027	0.0384	-0.2174	-0.0397	0.0471	0.0019	00.13	03.99	-0.0513	-0.0020	0.0048	-0.000	0.0484
017	0.437	0.841	00.13	0.2180	-0.0072	0.0406	-0.2182	-0.0393	0.0474	0.0017	00.13	02.00	-0.0249	-0.0035	0.0029	-0.000	0.0234
018	0.437	0.839	00.13	0.2186	-0.0080	0.0421	-0.2190	-0.0402	0.0478	0.0017	00.13	01.00	-0.0130	-0.0021	0.0020	-0.000	0.0122
019	0.437	0.842	00.13	0.2198	-0.0074	0.0417	-0.2200	-0.0395	0.0463	0.0016	00.13	00.00	-0.0008	-0.0007	0.0012	-0.000	0.0007
020	0.453	0.800	00.12	0.3974	0.0101	0.0350	-0.1976	-0.0263	0.0388	0.0015	00.12	03.99	-0.0501	-0.0068	0.0047	-0.000	0.0474
021	0.450	0.800	00.12	0.3976	0.0141	0.0378	-0.1978	-0.0266	0.0389	0.0015	00.12	02.00	-0.0244	-0.0020	0.0048	-0.000	0.0230
022	0.453	0.800	00.12	0.3976	0.0141	0.0378	-0.1978	-0.0266	0.0389	0.0015	00.12	01.00	-0.0124	-0.0021	0.0021	-0.000	0.0130
023	0.453	0.800	00.12	0.3976	0.0141	0.0378	-0.1978	-0.0266	0.0389	0.0015	00.12	01.00	-0.0125	-0.0022	0.0022	-0.000	0.0116
024	0.453	0.800	00.12	0.3976	0.0141	0.0378	-0.1978	-0.0266	0.0389	0.0015	00.12	01.00	-0.0125	-0.0022	0.0022	-0.000	0.0116
025	0.450	0.800	00.12	0.3986	0.0147	0.0396	-0.1990	-0.0377	0.0395	0.0015	00.13	00.00	-0.0004	-0.0007	0.0014	-0.000	0.0003
026	0.431	0.750	00.11	0.1840	0.0426	0.0354	-0.1842	-0.0370	0.0338	0.0015	00.11	03.99	-0.0497	-0.0065	0.0044	-0.000	0.0470
027	0.431	0.751	00.11	0.1849	0.0462	0.0370	-0.1851	-0.0359	0.0341	0.0016	00.11	02.00	-0.0244	-0.0023	0.0028	-0.000	0.0230
028	0.431	0.752	00.11	0.1862	0.0169	0.0392	-0.1861	-0.0373	0.0346	0.0016	00.11	02.00	-0.0244	-0.0021	0.0021	-0.000	0.0230
029	0.431	0.752	00.11	0.1860	0.0170	0.0393	-0.1861	-0.0368	0.0345	0.0016	00.11	02.00	-0.0246	-0.0020	0.0021	-0.000	0.0230
030	0.411	0.699	00.09	0.1727	0.0136	0.0338	-0.1729	-0.0355	0.0297	0.0013	00.09	03.99	-0.0486	-0.0062	0.0043	-0.000	0.0460
031	0.411	0.698	00.09	0.1737	0.0170	0.0364	-0.1739	-0.0355	0.0301	0.0015	00.09	01.99	-0.0244	-0.0034	0.0028	-0.000	0.0230
032	0.411	0.699	00.09	0.1746	0.0173	0.0371	-0.1750	-0.0354	0.0305	0.0016	00.09	01.00	-0.0125	-0.0020	0.0020	-0.000	0.0118
033	0.411	0.700	00.10	0.1764	0.0178	0.0389	-0.1765	-0.0369	0.0310	0.0017	00.10	00.00	-0.0007	-0.0007	0.0013	-0.000	0.006
034	0.411	0.698	00.07	0.1529	0.0181	0.0347	-0.1534	-0.0334	0.0232	0.0016	00.06	03.99	-0.0471	-0.0058	0.0042	-0.000	0.0447
035	0.431	0.500	00.07	0.1829	0.0188	0.0349	-0.1834	-0.0333	0.0232	0.0017	00.07	01.00	-0.0244	-0.0034	0.0020	-0.000	0.0114
036	0.431	0.501	00.07	0.1935	0.0190	0.0357	-0.1836	-0.0339	0.0234	0.0017	00.07	00.00	-0.0004	-0.0007	0.0016	-0.000	0.0003

TABLE 3(A)  
FLARE PACK ON  
ETA = 0° EXTRACT = -1/2"

SER REVN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLES.	BASE.	RINC.	SLIP.	CROSS.	VAN H.	ROLL H.	RANG.	SIDE F		
042	0.437	0.840	-0.05	0.0610	0.0161	0.0403	-0.0604	0.0391	0.0026	-0.95	-0.01	0.0014	0.0002	0.0015	180.0	-0.0015		
043	0.437	0.840	-0.12	-0.2276	0.0284	0.0458	0.2296	0.0313	0.0517	-0.12	-0.01	0.0026	0.0004	0.0019	180.0	-0.0027		
046	0.450	0.799	0.71	0.8230	-0.0822	0.1442	-0.8348	0.0342	0.6772	0.0028	0.41	0.00	-0.0007	-0.0011	0.0009	180.0	0.0006	
047	0.453	0.804	0.69	0.8009	-0.0755	0.1316	-0.8111	0.0320	0.6413	0.0025	0.90	0.00	-0.0004	-0.0007	0.0017	180.0	0.0003	
048	0.450	0.799	0.64	0.7813	-0.0621	0.1206	-0.7920	0.0299	0.6135	0.0023	0.40	0.00	0.0004	-0.0004	0.0021	180.0	-0.0005	
049	0.450	0.800	0.59	0.7690	-0.0386	0.1103	-0.7764	0.0279	0.5914	0.0025	0.92	0.00	0.0009	-0.0005	0.0024	180.0	-0.0010	
050	0.450	0.799	0.54	0.7558	-0.0354	0.1002	-0.7620	0.0261	0.5712	0.0024	0.41	0.00	-0.0007	-0.0006	0.0006	180.0	0.0006	
051	0.450	0.801	0.43	0.6842	-0.0133	0.0812	-0.6885	0.0261	0.4680	0.0025	0.39	0.00	-0.0003	-0.0004	0.0016	180.0	0.0002	
052	0.450	0.804	0.35	0.5842	0.0014	0.0630	-0.5970	0.0264	0.3412	0.0023	0.35	0.00	0.0006	0.0000	0.0022	180.0	-0.0007	
053	0.450	0.800	0.20	0.3183	0.0123	0.0441	-0.3193	0.0253	0.1012	0.0022	0.20	-0.01	0.0012	0.0002	0.0019	180.0	-0.0013	
054	0.453	0.800	-0.05	0.0535	0.0213	0.0387	-0.0530	0.0277	0.0027	0.0159	-0.95	0.00	0.0018	0.0004	0.0016	180.0	-0.0019	
055	0.450	0.799	-0.10	-0.2110	0.0316	0.0414	0.2129	-0.0283	0.0443	0.0017	-0.10	-0.01	0.0031	0.0004	0.0021	180.0	-0.0032	
056	0.434	0.751	0.74	0.8162	-0.0468	0.1367	-0.8271	0.0281	0.6660	0.0022	0.40	0.00	-0.0002	-0.0006	0.0020	180.0	0.0001	
057	0.434	0.749	0.69	0.8090	-0.0520	0.1259	-0.8183	0.0258	0.6543	0.0024	0.89	0.00	-0.0001	-0.0005	0.0028	180.0	0.0000	
058	0.434	0.751	0.59	0.7874	-0.0412	0.1141	-0.7953	0.0236	0.6195	0.0024	0.45	0.00	-0.0045	-0.0002	0.0033	180.0	-0.0016	
059	0.434	0.750	0.59	0.7662	-0.0274	0.1014	-0.7727	0.0200	0.5870	0.0022	0.89	0.00	0.0014	-0.0001	0.0037	180.0	-0.0012	
060	0.434	0.749	0.39	0.7399	-0.0093	0.0901	-0.7452	0.0181	0.5474	0.0021	0.39	0.00	0.0010	-0.0003	0.0026	180.0	-0.0011	
061	0.434	0.749	0.37	0.6650	0.0112	0.0710	-0.6696	0.0181	0.4421	0.0020	0.37	0.00	0.0012	-0.0002	0.0021	180.0	-0.0012	
062	0.434	0.749	0.20	0.5397	0.0125	0.0563	-0.5421	0.0234	0.2911	0.0020	0.20	0.00	0.0010	0.0002	0.0021	180.0	-0.0011	
063	0.434	0.752	0.17	0.2934	0.0143	0.0425	-0.2943	0.0246	0.0860	0.0020	0.17	0.00	0.0012	0.0001	0.0015	180.0	-0.0013	
064	0.434	0.750	-0.06	0.036	0.0492	0.0225	0.0386	-0.0487	0.0357	0.0023	0.016	-0.96	-0.10	0.0015	0.0003	0.0015	180.0	-0.0016
065	0.434	0.750	-0.02	0.036	0.0465	0.0263	0.0406	0.1997	-0.0281	0.0386	0.0016	-0.01	0.0026	0.0003	0.0020	180.0	-0.0027	
066	0.434	0.749	0.07	0.6155	-0.0826	0.1227	-0.8209	0.0294	0.6649	0.0020	0.26	0.00	0.0008	-0.0003	0.0024	180.0	-0.0009	
067	0.434	0.751	0.06	0.7985	-0.0286	0.1179	-0.8063	0.0196	0.6375	0.0020	0.86	0.00	0.0019	0.0001	0.0027	180.0	-0.0020	
068	0.434	0.751	0.06	0.7834	-0.0232	0.1068	-0.7903	0.0172	0.6135	0.0020	0.36	0.00	0.0011	0.0003	0.0018	180.0	-0.0012	
069	0.434	0.750	0.06	0.7819	-0.0025	0.0936	-0.7873	0.0143	0.5647	0.0020	0.86	0.00	0.0008	0.0002	0.0025	180.0	-0.0009	
070	0.434	0.750	0.05	0.7152	0.0006	0.0833	-0.7201	0.0142	0.5116	0.0020	0.35	0.00	0.0004	0.0000	0.0026	180.0	-0.0005	
071	0.434	0.751	0.04	0.6173	0.0114	0.0651	-0.6205	0.0165	0.3910	0.0020	0.31	0.00	0.0007	0.0003	0.0027	180.0	-0.0006	
072	0.434	0.751	0.04	0.6053	0.0118	0.0526	-0.6076	0.0219	0.2532	0.0020	0.25	0.00	0.0013	0.0005	0.0026	180.0	-0.0014	
073	0.434	0.750	0.04	0.2763	0.0149	0.0404	-0.2761	0.0332	0.0787	0.0018	0.14	-0.01	0.0009	0.0004	0.0015	180.0	-0.0010	
074	0.434	0.750	0.03	0.0449	0.0247	0.0267	-0.0440	0.0266	0.0018	0.0147	-0.96	-0.01	0.0016	0.0002	0.0016	180.0	-0.0017	
075	0.434	0.750	-0.03	0.0446	0.0277	0.0266	-0.0463	0.0268	0.0043	0.016	-0.07	-0.01	0.0027	0.0004	0.0019	180.0	-0.0028	
076	0.434	0.750	0.02	0.7787	-0.0214	0.1172	-0.7874	0.0154	0.6063	0.0026	0.28	0.00	0.0003	0.0003	0.0030	180.0	-0.0004	
077	0.434	0.750	0.02	0.7501	-0.0183	0.1033	-0.7571	0.0119	0.5625	0.0022	0.77	-0.01	0.0003	0.0005	0.0025	180.0	-0.0004	
078	0.434	0.750	0.02	0.7164	-0.0062	0.0910	-0.7222	0.0101	0.5132	0.0021	0.27	-0.01	0.0003	0.0004	0.0021	180.0	0.0002	
079	0.434	0.750	0.02	0.6728	-0.0015	0.0794	-0.6771	0.0095	0.4521	0.0021	0.75	-0.01	-0.0004	0.0004	0.0022	180.0	0.0003	
080	0.434	0.750	0.02	0.6394	0.0024	0.0703	-0.6294	0.0107	0.3899	0.0023	0.23	-0.01	0.0004	0.0005	0.0024	180.0	-0.0005	
081	0.434	0.750	0.02	0.5326	0.0055	0.0563	-0.5356	0.0150	0.2838	0.0028	0.20	-0.01	0.0012	0.0012	0.0029	180.0	-0.0013	

TABLE 3(R)  
FLARE PACK ON  
ETA = 0° ETACT = -1/2°

SER REYN.	IRCH.	INCID.	LIFT.	PITCH.	DREG	NORML	AXIAL.	CLSQ.	BASE.	INC.	SLIP.	CROSS.	VAN H.	ROLL H.	RNG.	SIDE F
064	0.434	0.503	03.17	0.4397	0.0080	0.0477	-0.4447	0.0243	0.1934	0.0024	03.17	-00.04	0.0014	0.0005	0.0024	180.0 -0.0015
065	0.431	0.499	01.10	0.2420	0.0135	0.0378	-0.2428	0.0212	0.0585	0.0020	01.10	-00.04	0.0017	0.0004	0.0020	180.0 -0.0016
066	0.431	0.499	-00.97	0.0394	0.0262	0.0348	-0.0386	0.0238	0.0014	0.0019	-00.97	-00.04	0.0017	0.0002	0.0019	180.0 -0.0016
067	0.431	0.499	-03.04	-0.1636	0.0396	0.0378	0.1632	-0.0276	0.0266	0.0016	-03.04	-00.04	0.0025	0.0002	0.0022	180.0 -0.0025

TABLE 3(B)  
FLARE PACK ON  
ETA = 0° (ETACT) = -2 1/2°

SER	REVR.	MACH.	INCID.	LIFT.	PITCH.	DRAO.	NORMAL	AXIAL.	CLSQ.	BASE.	RINC.	SLIP.	CROSS.	YAW H.	ROLL H.	RAND.	SIDE F.								
0002	0	0.452	0	0.899	07.45	0	0.8454	-0	0.0768	0.1690	-0	0.8602	-0	0.0558	0.7145	0.0021	07.45	-00.01	-0.0056	-0	0.0008	-0	0.0004	179.9	0.0055
0003	0	0.455	0	0.900	06.94	0	0.8111	-0	0.0724	0.1565	-0	0.8242	-0	0.0553	0.6578	0.0021	06.94	-00.01	-0.0054	-0	0.0009	-0	0.0007	179.9	0.0053
0004	0	0.452	0	0.899	06.42	0	0.7690	-0	0.0647	0.1424	-0	0.7803	-0	0.0534	0.5914	0.0022	06.42	00.00	-0.0040	-0	0.0011	-0	0.0015	179.9	0.0039
0005	0	0.453	0	0.902	05.90	0	0.7337	-0	0.0605	0.1313	-0	0.7434	-0	0.0528	0.5382	0.0023	05.90	00.00	-0.0045	-0	0.0010	-0	0.0010	179.9	0.0044
0006	0	0.455	0	0.900	05.38	0	0.6939	-0	0.0500	0.1212	-0	0.7023	-0	0.0534	0.4814	0.0021	05.38	00.00	-0.0046	-0	0.0010	-0	0.0008	179.9	0.0045
0007	0	0.455	0	0.899	04.34	0	0.6120	-0	0.0362	0.1017	-0	0.6180	-0	0.0531	0.3744	0.0020	04.34	00.00	-0.0041	-0	0.0007	-0	0.0006	180.0	0.0040
0008	0	0.455	0	0.900	03.30	0	0.5183	-0	0.0262	0.0852	-0	0.5225	-0	0.0512	0.2685	0.0021	03.30	00.01	-0.0026	-0	0.0003	-0	0.0010	180.0	0.0025
0009	0	0.455	0	0.901	01.17	0	0.2930	-0	0.0058	0.0627	-0	0.2943	-0	0.0458	0.0857	0.0020	01.17	00.01	-0.0014	0	0.0000	0	0.0015	180.0	0.0013
0110	0	0.455	0	0.898	-00.96	0	0.0331	0	0.0255	0.0505	-0	0.0323	-0	0.0496	0.0010	0.0015	-00.96	00.01	0.0011	0	0.0006	0	0.0016	180.0	-0.0012
0111	0	0.455	0	0.900	-03.10	-0	0.2710	0	0.0728	0.0629	0	0.2739	-0	0.0467	0.0733	0.0015	-03.10	00.01	0.0027	0	0.0005	0	0.0018	180.0	-0.0028
0114	0	0.453	0	0.878	07.40	0	0.6131	-0	0.0281	0.1584	-0	0.8269	-0	0.0492	0.6610	0.0020	07.48	00.00	-0.0054	-0	0.0009	0	0.0019	180.0	0.0053
0115	0	0.453	0	0.879	06.94	0	0.8037	-0	0.0379	0.1489	-0	0.8159	-0	0.0488	0.6459	0.0019	06.94	00.00	-0.0040	-0	0.0008	0	0.0000	179.9	0.0039
0116	0	0.453	0	0.880	06.43	0	0.7662	-0	0.0526	0.1358	-0	0.7767	-0	0.0470	0.5870	0.0021	06.43	00.00	-0.0045	-0	0.0008	-0	0.0009	179.9	0.0044
0117	0	0.457	0	0.880	05.91	0	0.7257	-0	0.0431	0.1240	-0	0.7347	-0	0.0465	0.5266	0.0021	05.91	00.00	-0.0035	-0	0.0011	-0	0.0010	179.9	0.0034
0118	0	0.457	0	0.860	05.39	0	0.6874	-0	0.0347	0.1137	-0	0.6952	-0	0.0466	0.4724	0.0021	05.39	00.00	-0.0042	-0	0.0010	-0	0.0003	179.9	0.0041
0119	0	0.453	0	0.879	04.36	0	0.6111	-0	0.0187	0.0932	-0	0.6167	-0	0.0464	0.3733	0.0020	04.36	00.00	-0.0039	-0	0.0009	0	0.0001	180.0	0.0053
0120	0	0.455	0	0.880	03.32	0	0.5228	-0	0.0081	0.0792	-0	0.5266	-0	0.0470	0.2731	0.0019	03.32	00.00	-0.0027	-0	0.0004	0	0.0011	180.0	0.0026
0121	0	0.457	0	0.880	04.20	0	0.2169	-0	0.0087	0.0570	-0	0.2482	-0	0.0486	0.1004	0.0019	01.20	00.01	-0.0008	0	0.0001	0	0.0013	180.0	0.0007
0122	0	0.453	0	0.879	-00.94	0	0.0542	0	0.0324	0.0472	-0	0.0525	-0	0.0465	0.0028	0.0017	-00.94	00.01	0.0006	0	0.0002	0	0.0016	180.0	-0.0007
0123	0	0.455	0	0.883	-03.10	-0	0.2639	0	0.0699	0.0563	0	0.2665	-0	0.0405	0.0695	0.0015	-03.10	00.01	0.0029	0	0.0005	0	0.0014	180.0	-0.0030
0226	0	0.452	0	0.860	07.45	0	0.7967	-0	0.0411	0.1517	-0	0.8098	-0	0.0453	0.6347	0.0017	07.45	00.00	-0.0039	-0	0.0011	0	0.0045	180.0	0.0038
0227	0	0.452	0	0.864	06.95	0	0.7892	-0	0.0339	0.1407	-0	0.8005	-0	0.0439	0.5940	0.0020	06.95	00.01	-0.0043	-0	0.0009	0	0.0015	180.0	0.0054
0228	0	0.452	0	0.860	04.43	0	0.7708	-0	0.0425	0.1331	-0	0.7809	-0	0.0439	0.5394	0.0020	06.43	00.00	-0.0043	-0	0.0009	0	0.0006	179.9	0.0044
0229	0	0.455	0	0.864	05.92	0	0.7343	-0	0.0340	0.1196	-0	0.7428	-0	0.0413	0.5391	0.0020	05.92	00.00	-0.0039	-0	0.0008	-0	0.0005	179.9	0.0038
0330	0	0.452	0	0.860	05.40	0	0.6936	-0	0.0237	0.1036	-0	0.7010	-0	0.0418	0.4810	0.0020	05.40	00.01	-0.0045	-0	0.0007	-0	0.0002	179.9	0.0044
0331	0	0.452	0	0.860	04.37	0	0.6196	-0	0.0074	0.0913	-0	0.6248	-0	0.0418	0.3818	0.0020	04.37	00.00	-0.0035	-0	0.0008	0	0.0003	180.0	0.0034
0332	0	0.452	0	0.860	03.23	0	0.5350	0	0.0064	0.0746	-0	0.5385	-0	0.0416	0.2860	0.0018	02.23	00.00	-0.0024	-0	0.0005	0	0.0011	180.0	0.0023
0333	0	0.455	0	0.864	01.22	0	0.3299	0	0.0234	0.0536	-0	0.3310	-0	0.0448	0.1087	0.0018	01.22	00.01	-0.0006	0	0.0005	0	0.0015	180.0	0.0005
0334	0	0.455	0	0.861	-00.93	0	0.0529	0	0.0443	0.0438	-0	0.0523	-0	0.0431	0.0027	0.0016	-00.93	00.01	0.0008	0	0.0005	0	0.0015	180.0	-0.0004
0335	0	0.452	0	0.860	-03.10	-0	0.2505	0	0.0649	0.0501	0	0.2528	-0	0.0354	0.0626	0.0012	-03.10	00.01	0.0020	0	0.0004	0	0.0016	180.0	-0.0021
0336	0	0.450	0	0.840	07.42	0	0.7971	-0	0.0610	0.1491	-0	0.8098	-0	0.0430	0.6352	0.0019	07.42	00.00	-0.0041	-0	0.0014	0	0.0045	180.0	0.0040
0339	0	0.450	0	0.842	06.93	0	0.7868	-0	0.0524	0.1383	-0	0.7979	-0	0.0406	0.6190	0.0018	06.93	00.00	-0.0036	-0	0.0007	0	0.0032	180.0	0.0035
0400	0	0.450	0	0.842	06.44	0	0.7583	-0	0.0461	0.1276	-0	0.7680	-0	0.0398	0.5750	0.0019	06.44	00.00	-0.0048	-0	0.0009	0	0.0005	180.0	0.0047
0411	0	0.449	0	0.839	05.93	0	0.7477	-0	0.0250	0.1174	-0	0.7560	-0	0.0376	0.5590	0.0019	05.93	00.00	-0.0042	-0	0.0008	-0	0.0004	179.9	0.0044
0422	0	0.450	0	0.840	05.41	0	0.7091	-0	0.0431	0.1056	-0	0.7161	-0	0.0363	0.5027	0.0019	05.41	00.00	-0.0033	-0	0.0008	-0	0.0001	180.0	0.0032
0433	0	0.450	0	0.839	04.38	0	0.6345	0	0.0048	0.0870	-0	0.6394	-0	0.0364	0.4025	0.0019	04.38	00.00	-0.0036	-0	0.0007	0	0.0007	180.0	0.0035

TABLE 3(B)  
FLARE PACK ON  
 $\text{ETA} = 0^\circ$  ETACT = -2 1/2°

SER. REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAG	NORMAL	AXIAL.	CLSQ.	BASE.	RINC.	SLIP.	CROSS.	VAN M.	ROLL M.	RANG.	SIDE F	
044	0.450	0.640	03.35	0.5491	0.0188	0.0719	-0.5524	-0.0380	0.0017	0.315	(0.00	-0.0017	-0.0002	0.0013	180.0	0.0016	
045	0.453	0.642	01.23	0.3278	0.0176	0.0496	-0.3289	-0.0409	0.1073	0.0048	0.1.23	-0.01	0.0004	0.0016	180.0	-0.0002	
046	0.450	0.639	-0.93	0.0467	0.0526	0.0412	-0.0462	-0.0406	0.0021	0.0014	-0.93	-0.01	0.0004	0.0003	180.0	-0.0005	
047	0.450	0.640	-03.09	-0.2372	0.0625	0.0476	0.2193	-0.0335	0.0561	0.0013	-0.09	-0.01	0.0018	0.0003	0.0017	180.0	-0.0019
052	0.450	0.601	07.43	0.8045	-0.0459	0.1427	-0.8163	-0.0356	0.6471	0.0020	0.7.43	0.00	-0.0032	-0.0011	0.0014	180.0	0.0031
053	0.450	0.801	06.92	0.7829	-0.0193	0.1312	-0.7934	-0.0342	0.6128	0.0017	0.6.92	(0.00	-0.0020	-0.0005	0.0032	180.0	0.0019
054	0.450	0.801	06.44	0.7500	0.0038	0.1190	-0.7588	-0.0324	0.5625	0.0017	0.6.44	(0.00	-0.0029	-0.0003	0.0034	180.0	0.0026
055	0.450	0.802	05.95	0.7229	0.0327	0.1072	-0.7292	-0.0299	0.5229	0.0018	0.5.95	(0.00	-0.0017	-0.0004	0.0026	180.0	0.0016
056	0.450	0.802	05.43	0.7327	0.0059	0.1012	-0.7394	-0.0294	0.5368	0.0019	0.5.43	-0.01	-0.0038	-0.0005	0.0007	180.0	0.0037
057	0.450	0.800	04.41	0.6605	0.0301	0.0800	-0.6649	-0.0279	0.4361	0.0018	0.4.41	(0.00	-0.0028	-0.0005	0.0014	180.0	0.0027
058	0.450	0.800	03.37	0.5596	0.0457	0.0645	-0.5623	-0.0297	0.3130	0.0018	0.3.37	-0.01	-0.0011	0.0000	0.0020	180.0	0.0010
059	0.450	0.801	04.22	0.2975	0.0459	0.0454	-0.2985	-0.0375	0.0884	0.0016	0.4.22	(0.01	-0.0010	0.0003	0.0015	180.0	0.0008
060	0.450	0.798	-00.93	0.0382	0.0605	0.0405	-0.0376	-0.0376	0.0043	0.0013	-0.93	(0.01	0.0001	0.0003	0.0014	180.0	-0.0002
061	0.450	0.800	-03.07	-0.2206	0.0666	0.0448	-0.2226	-0.0218	0.0485	0.0016	-0.07	-0.01	0.0023	0.0004	0.0016	180.0	-0.0024
064	0.445	0.750	07.41	0.6033	-0.0329	0.1350	-0.8141	-0.0284	0.6452	0.0018	0.7.41	(0.00	-0.0028	-0.0003	0.0023	180.0	0.0027
065	0.445	0.750	06.92	0.7859	-0.0155	0.1252	-0.7954	-0.0280	0.6176	0.0016	0.6.92	(0.00	-0.0029	-0.0004	0.0022	180.0	0.0028
066	0.447	0.752	06.42	0.7672	-0.0010	0.1427	-0.7752	-0.0244	0.5887	0.0017	0.6.42	(0.00	-0.0010	-0.0003	0.0038	180.0	0.0009
067	0.445	0.750	05.92	0.7465	0.0122	0.1017	-0.7531	-0.0222	0.5571	0.0018	0.5.92	(0.00	-0.0002	0.0002	0.0044	180.0	0.0001
068	0.449	0.751	05.43	0.7153	0.0449	0.0883	-0.7206	-0.0184	0.5116	0.0017	0.5.43	(0.01	-0.0014	0.0001	0.0032	180.0	0.0013
069	0.447	0.746	04.39	0.6394	0.0544	0.0704	-0.6430	-0.0197	0.4086	0.0015	0.4.39	(0.01	-0.0033	-0.0001	0.0020	180.0	0.0032
070	0.445	0.749	03.32	0.5476	0.0566	0.0568	-0.5201	-0.0248	0.2677	0.0018	0.3.32	-0.01	-0.0020	0.0002	0.0022	180.0	0.0019
071	0.449	0.750	04.20	0.2741	0.0364	0.0434	-0.2751	-0.0362	0.0750	0.0015	0.4.20	(0.01	-0.0010	-0.0003	0.0016	180.0	0.0008
072	0.449	0.752	-00.93	0.0317	0.0635	0.0389	-0.0312	-0.0380	0.0039	0.0014	-0.93	(0.01	0.0001	0.0002	0.0015	180.0	-0.0002
073	0.449	0.749	-03.06	-0.2074	0.0726	0.0437	-0.2094	-0.0315	0.0429	0.0012	-0.06	(0.01	0.0011	0.0004	0.0016	180.0	-0.0012
076	0.440	0.701	07.40	0.7942	-0.0113	0.1310	-0.8046	-0.0259	0.6207	0.0017	0.7.40	(0.01	-0.0032	0.0003	0.0037	180.0	0.0031
077	0.440	0.701	06.90	0.7827	0.0000	0.1180	-0.7913	-0.0215	0.6125	0.0016	0.6.90	(0.01	-0.0013	0.0004	0.0041	180.0	0.0012
078	0.439	0.701	06.40	0.7639	0.0163	0.1062	-0.7711	-0.0186	0.5815	0.0017	0.6.40	(0.01	-0.0012	0.0004	0.0040	180.0	0.0008
079	0.437	0.699	05.90	0.7344	0.0314	0.0528	-0.7403	-0.0160	0.5294	0.0018	0.5.90	(0.01	-0.0017	0.0004	0.0028	180.0	0.0016
080	0.440	0.701	05.39	0.6975	0.0413	0.0873	-0.7024	-0.0158	0.4864	0.0018	0.5.39	(0.01	-0.0016	0.0003	0.0031	180.0	0.0015
081	0.440	0.700	04.35	0.5994	0.0525	0.0643	-0.6026	-0.0170	0.7592	0.0017	0.4.35	(0.01	-0.0013	0.0003	0.0028	180.0	0.0012
082	0.440	0.699	03.29	0.4860	0.0534	0.0539	-0.4904	-0.0240	0.2380	0.0018	0.3.29	(0.01	-0.0011	0.0006	0.0027	180.0	0.0010
083	0.440	0.700	01.17	0.2694	0.0561	0.0419	-0.2603	-0.0250	0.0672	0.0015	0.17	-0.01	-0.0011	0.0004	0.0017	180.0	0.0010
084	0.440	0.698	-0.34	0.0264	0.0652	0.0294	-0.0279	-0.0284	0.0007	0.0016	-0.34	-0.01	0.0001	0.0016	180.0	-0.0002	
085	0.439	0.702	-03.05	-0.1993	0.0753	0.0431	-0.2012	-0.0312	0.0396	0.0015	-0.05	-0.01	0.0004	0.0014	0.0019	180.0	-0.0021
086	0.452	0.502	07.30	0.7984	0.0029	0.1194	-0.7646	-0.0206	0.5706	0.0019	0.7.30	(0.01	-0.0028	0.0007	0.0028	180.0	0.0027
089	0.452	0.601	08.80	0.7342	0.0493	0.1030	-0.7433	-0.0136	0.5390	0.0018	0.80	-0.01	-0.0027	0.0006	0.0024	180.0	0.0026

TABLE 3(B)  
FLARE PACK ON  
ETA = 0° ET(1) = -2 1/2°

SER REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORMA	AXIAL.	CLSQ.	BASE.	RINC.	SLIP.	CROSS.	YAW M.	ROLL M.	RANG.	SIDE F
004	0.449	0.500	06.29	0.6993	0.0296	0.0904	-0.7051	-0.0114	0.4889	0.0018	06.29	-00.01	-0.0028	0.0005	0.0022	180.0
005	0.450	0.500	05.77	0.6533	0.0169	0.0790	-0.6581	-0.0112	0.4268	0.0017	05.77	-00.01	-0.0028	0.0005	0.0021	180.0
006	0.449	0.500	05.26	0.6085	0.0407	0.0702	-0.6125	-0.0122	0.3702	0.0019	05.26	-00.01	-0.0022	0.0005	0.0022	180.0
007	0.450	0.500	04.22	0.5160	0.0443	0.0571	-0.5189	-0.0170	0.2661	0.0019	04.22	-00.01	-0.0014	0.0007	0.0020	180.0
008	0.452	0.501	03.19	0.4224	0.0472	0.0484	-0.4243	-0.0230	0.1780	0.0016	03.19	-00.01	-0.0007	0.0006	0.0024	180.0
009	0.452	0.501	01.12	0.2260	0.0547	0.0393	-0.2268	-0.0131	0.0510	0.0018	01.12	-00.01	-0.0005	0.0003	0.0018	180.0
010	0.452	0.501	-00.96	0.0216	0.0646	0.0168	-0.0211	-0.0155	0.0004	0.0016	-00.96	-00.01	0.0008	0.0002	0.0017	180.0
011	0.452	0.501	-03.03	-0.1816	0.0785	0.0409	0.1834	-0.0304	0.0329	0.0013	-03.03	-00.01	0.0021	0.0002	0.0019	180.0

TABLE 3(C)  
FLARE PACK: ON  
ETA = 0° ETAC(1) = -1 1/2°

SER	REYN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSL.	BASE.	ANIC.	SLIP.	CROSS.	YAN M.	ROLL M.	ROLL N.	SIDE F
002	0.452	0.898	07.45	0.8446	-0.0797	0.1676	-0.8693	-0.0545	0.7132	0.0022	0.15	-00.02	0.0002	0.0020	-0.0005	179.9	-0.0002
003	0.452	0.899	06.92	0.8127	-0.0819	0.1551	-0.8286	-0.0539	0.6604	0.0021	0.92	-00.02	0.0006	0.0018	-0.0010	179.9	-0.0007
004	0.455	0.901	06.40	0.7758	-0.0856	0.1417	-0.7869	-0.0520	0.6019	0.0023	0.40	-00.01	0.0011	0.0015	-0.0022	179.9	-0.0011
005	0.453	0.902	05.88	0.7403	-0.0822	0.1309	-0.7499	-0.0519	0.5179	0.0024	0.88	-00.01	0.0001	0.0013	-0.0014	179.9	-0.0002
006	0.455	0.901	05.36	0.7011	-0.0748	0.1194	-0.7093	-0.0510	0.4914	0.0024	0.36	-00.01	0.0002	0.0016	-0.0013	179.9	-0.0003
007	0.455	0.901	04.32	0.6220	-0.0645	0.1007	-0.6289	-0.0511	0.3881	0.0024	0.32	-00.01	0.0016	0.0016	-0.0001	180.0	-0.0017
008	0.455	0.901	03.27	0.5288	-0.0517	0.0836	-0.5128	-0.0511	0.2794	0.0021	0.27	-00.02	0.0013	0.0024	-0.0005	180.0	-0.0024
009	0.455	0.901	01.15	0.3042	-0.0316	0.0620	-0.3055	-0.0528	0.0924	0.0021	0.15	-00.02	0.0038	0.0025	-0.0008	180.0	-0.0035
010	0.455	0.900	-00.98	0.0400	0.0047	0.0503	-0.0393	-0.0493	0.0115	0.0017	0.98	-00.02	0.0056	0.0028	0.0010	180.0	-0.0057
011	0.455	0.901	-03.12	-0.2626	0.0307	0.0614	0.2655	-0.0454	0.0689	0.0016	0.12	-00.02	0.0075	0.0031	0.0012	180.0	-0.0076
014	0.453	0.880	07.47	0.8213	-0.0375	0.1580	-0.8380	-0.0479	0.6745	0.0020	0.47	-00.01	-0.0013	0.0014	0.0008	180.0	0.0012
015	0.453	0.881	06.92	0.8094	-0.0292	0.1498	-0.8216	-0.0489	0.6550	0.0022	0.92	-00.02	0.0003	0.0018	-0.0007	179.9	-0.0004
015	0.453	0.881	06.92	0.8094	-0.0292	0.1498	-0.8216	-0.0489	0.6550	0.0022	0.92	-00.02	0.0003	0.0018	-0.0007	179.9	-0.0004
017	0.455	0.881	06.41	0.7743	-0.0739	0.1369	-0.7849	-0.0475	0.5995	0.0021	0.41	-00.01	0.0008	0.0016	-0.0014	179.9	-0.0009
018	0.453	0.879	05.89	0.7352	-0.0664	0.1229	-0.7440	-0.0447	0.5403	0.0021	0.89	-00.01	0.0015	0.0016	-0.0012	179.9	-0.0019
019	0.453	0.879	05.37	0.6985	-0.0600	0.1126	-0.7061	-0.0445	0.4878	0.0022	0.37	-00.01	0.0008	0.0016	-0.0008	179.9	-0.0009
020	0.453	0.878	04.34	0.6247	-0.0432	0.0945	-0.6272	-0.0449	0.3864	0.0022	0.34	-00.01	0.0008	0.0017	-0.0004	179.9	-0.0008
021	0.453	0.880	03.20	0.5251	-0.0340	0.0784	-0.5388	-0.0453	0.2861	0.0021	0.30	-00.01	0.0023	0.0020	0.0006	180.0	-0.0024
022	0.455	0.881	01.19	0.2262	-0.0192	0.0571	-0.3274	-0.0484	0.1062	0.0020	0.18	-00.02	0.0041	0.0027	0.0008	180.0	-0.0042
023	0.453	0.879	-00.95	0.0610	0.0134	0.0439	-0.0604	-0.0434	0.0336	0.0016	0.95	-00.02	0.0050	0.0028	0.0011	180.0	-0.0051
024	0.455	0.880	-03.12	-0.2352	0.0478	0.0535	0.2076	-0.0380	0.0649	0.0016	0.12	-00.02	0.0072	0.0031	0.0010	180.0	-0.0073
027	0.455	0.860	07.42	0.7950	-0.0778	0.1521	-0.8121	-0.1455	0.6284	0.0022	0.42	-00.02	0.0022	0.0013	0.0045	180.0	-0.0023
029	0.452	0.859	06.95	0.7849	-0.0742	0.1390	-0.7984	-0.1409	0.6180	0.0020	0.95	-00.01	0.0009	0.0018	0.0013	180.0	-0.0009
029	0.455	0.860	06.41	0.7800	-0.0665	0.1221	-0.7900	-0.0422	0.6083	0.0020	0.41	-00.01	0.0009	0.0017	-0.0007	179.9	-0.0010
030	0.455	0.861	05.90	0.7449	-0.0587	0.1210	-0.7535	-0.0417	0.5546	0.0020	0.90	-00.01	0.0007	0.0016	-0.0006	179.9	-0.0008
031	0.455	0.860	05.39	0.7057	-0.0485	0.1092	-0.7429	-0.0403	0.4978	0.0022	0.39	-00.01	0.0021	0.0017	-0.0006	179.9	-0.0022
032	0.455	0.860	04.25	0.6312	-0.0325	0.0913	-0.6364	-0.0411	0.3983	0.0020	0.35	-00.01	0.0014	0.0023	-0.0002	179.9	-0.0015
033	0.455	0.859	03.31	0.5473	-0.0198	0.0748	-0.5609	-0.0410	0.2994	0.0020	0.31	-00.01	0.0013	0.0026	0.0006	180.0	-0.0020
034	0.455	0.859	01.24	0.2396	0.0009	0.0519	-0.3407	-0.0428	0.1432	0.0020	0.21	-00.02	0.0046	0.0029	0.0011	180.0	-0.0047
026	0.485	0.859	-00.94	0.0887	0.0264	0.0417	-0.0881	-0.0409	0.0633	0.0016	0.94	-00.02	0.0048	0.0029	0.0010	180.0	-0.0049
026	0.455	0.859	-03.11	-0.2406	0.0420	0.0478	0.2427	-0.0233	0.0577	0.0014	0.11	-00.02	0.0064	0.0030	0.0010	180.0	-0.0064
029	0.453	0.841	07.41	0.8046	-0.0807	0.1450	-0.8172	-0.0428	0.6472	0.0020	0.41	-00.01	0.0011	0.0028	0.0010	180.0	-0.0012
040	0.450	0.839	06.90	0.7922	-0.0731	0.1370	-0.7931	-0.0400	0.6317	0.0020	0.90	-00.01	0.0016	0.0027	0.0011	180.0	-0.0027
041	0.450	0.839	06.40	0.7726	-0.0637	0.1266	-0.7820	-0.0377	0.5968	0.0018	0.46	-00.01	0.0020	0.0020	0.0018	180.0	-0.0020
042	0.450	0.840	05.91	0.7556	-0.0494	0.1472	-0.7626	-0.0366	0.5710	0.0023	0.91	-00.01	0.0009	0.0016	-0.0007	179.9	-0.0010
043	0.450	0.838	05.40	0.7206	-0.0372	0.1053	-0.7278	-0.0353	0.5192	0.0019	0.40	-00.01	0.0011	0.0017	-0.0003	179.9	-0.0012

TABLE 3(C)  
FLARE PACK ON  
 $\Theta_{TA} = 0^\circ$ ,  $\Theta_{TACT} = -1^\circ 1/2^\circ$

SER	REVN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORML.	AXIAL.	CLSR.	BASE.	NING.	SLIP.	CROSS.	VAN W.	ROLL W.	RANG.	SIDE F.
044	0. 453	0. 841	04. 37	0. 6426	-0. 0212	0. 0878	-0. 6498	-0. 0364	0. 4141	0. 0024	04. 37	-0. 01	0. 0024	0. 0015	0. 0002	180. 0	-0. 0022
045	0. 453	0. 840	03. 33	0. 5605	-0. 0074	0. 0710	-0. 5628	-0. 0263	0. 3140	0. 0020	01. 33	-0. 02	0. 0022	0. 0007	180. 0	-0. 0023	
046	0. 453	0. 842	04. 24	0. 3349	0. 0143	0. 0492	-0. 3360	-0. 0402	0. 1124	0. 0019	01. 21	-0. 02	0. 0043	0. 0028	0. 0011	180. 0	-0. 0044
047	0. 453	0. 840	-00. 94	0. 0554	0. 0329	0. 0403	-0. 0548	-0. 0396	0. 0030	0. 0017	-0. 94	-0. 02	0. 0050	0. 0029	0. 0009	180. 0	-0. 0051
048	0. 453	0. 841	-03. 11	-0. 2314	0. 0414	0. 0463	0. 2332	-0. 0123	0. 0533	0. 0015	-01. 11	-0. 02	0. 0060	0. 0030	0. 0011	180. 0	-0. 0061
051	0. 467	0. 800	02. 42	0. 8161	-0. 0681	0. 1426	-0. 8279	-0. 0349	0. 6659	0. 0020	07. 42	-0. 01	0. 0013	0. 0014	0. 0010	180. 0	-0. 0014
052	0. 467	0. 799	06. 91	0. 7922	-0. 0584	0. 1310	-0. 8023	-0. 0328	0. 6275	0. 0018	06. 91	-0. 01	0. 0014	0. 0018	0. 0007	180. 0	-0. 0011
053	0. 467	0. 799	06. 13	0. 7960	-0. 0217	0. 1179	-0. 7646	-0. 0107	0. 5715	0. 0018	06. 13	-0. 01	0. 0023	0. 0019	0. 0029	180. 0	-0. 0024
054	0. 470	0. 800	05. 92	0. 7491	-0. 0126	0. 1081	-0. 7554	-0. 0285	0. 5596	0. 0017	05. 92	-0. 01	0. 0029	0. 0020	0. 0025	180. 0	-0. 0020
055	0. 467	0. 799	05. 42	0. 7448	-0. 0176	0. 0996	-0. 7509	-0. 0270	0. 5546	0. 0018	05. 42	-0. 01	0. 0009	0. 0018	0. 0004	180. 0	-0. 0010
056	0. 467	0. 799	04. 40	0. 6721	0. 0060	0. 0793	-0. 6763	-0. 0257	0. 4516	0. 0020	04. 40	-0. 01	0. 0018	0. 0020	0. 0009	180. 0	-0. 0016
057	0. 470	0. 799	02. 25	0. 5672	0. 0209	0. 0623	-0. 5701	-0. 0283	0. 3216	0. 0017	03. 35	-0. 01	0. 0013	0. 0013	0. 0017	180. 0	-0. 0014
058	0. 470	0. 801	04. 24	0. 2058	0. 0304	0. 0447	-0. 3065	-0. 0368	0. 0932	0. 0016	01. 21	-0. 02	0. 0043	0. 0028	0. 0013	180. 0	-0. 0044
059	0. 467	0. 801	-00. 94	0. 0463	0. 0380	0. 0390	-0. 0459	-0. 0363	0. 0202	0. 0015	-00. 94	-0. 02	0. 0048	0. 0028	0. 0011	180. 0	-0. 0049
060	0. 467	0. 801	-03. 09	-0. 2134	0. 0453	0. 0433	0. 2150	-0. 0306	0. 0453	0. 0012	-03. 09	-0. 02	0. 0062	0. 0029	0. 0014	180. 0	-0. 0063
063	0. 457	0. 746	02. 40	0. 8154	-0. 0516	0. 1256	-0. 8259	-0. 0277	0. 6643	0. 0017	07. 40	-0. 01	0. 0016	0. 0019	0. 0014	180. 0	-0. 0017
064	0. 460	0. 751	06. 93	0. 7940	-0. 0104	0. 1239	-0. 7933	-0. 0263	0. 6146	0. 0017	06. 93	-0. 01	0. 0016	0. 0024	0. 0017	180. 0	-0. 0017
065	0. 460	0. 750	06. 41	0. 7739	-0. 0165	0. 1126	-0. 7716	-0. 0239	0. 5986	0. 0017	06. 41	-0. 01	0. 0013	0. 0022	0. 0011	180. 0	-0. 0023
066	0. 460	0. 751	05. 94	0. 7556	-0. 0132	0. 1025	-0. 7623	-0. 0223	0. 5709	0. 0018	05. 91	-0. 01	0. 0044	0. 0026	0. 0012	180. 0	-0. 0045
067	0. 460	0. 750	05. 41	0. 7291	0. 0052	0. 0507	-0. 7345	-0. 0198	0. 5319	0. 0018	05. 41	-0. 01	0. 0042	0. 0025	0. 0010	180. 0	-0. 0042
068	0. 460	0. 751	04. 29	0. 6544	0. 0282	0. 0769	-0. 6580	-0. 0187	0. 4281	0. 0019	04. 29	-0. 01	0. 0017	0. 0016	0. 0014	180. 0	-0. 0017
069	0. 460	0. 749	02. 32	0. 3290	0. 0306	0. 0563	-0. 5313	-0. 0219	0. 2798	0. 0017	03. 32	-0. 01	0. 0016	0. 0022	0. 0018	180. 0	-0. 0017
070	0. 460	0. 749	01. 18	0. 2819	0. 0213	0. 0422	-0. 2826	-0. 0348	0. 0794	0. 0017	01. 18	-0. 02	0. 0040	0. 0026	0. 0012	180. 0	-0. 0041
071	0. 460	0. 750	-00. 95	0. 0419	0. 0401	0. 0378	-0. 0414	-0. 0372	0. 0017	0. 0013	-00. 95	-0. 02	0. 0056	0. 0027	0. 0010	180. 0	-0. 0057
072	0. 460	0. 750	-03. 07	-0. 1987	0. 0516	0. 0426	0. 2007	-0. 0307	0. 0394	0. 0013	-03. 07	-0. 02	0. 0061	0. 0029	0. 0013	180. 0	-0. 0062
073	0. 445	0. 699	02. 40	0. 7980	-0. 0079	0. 1314	-0. 8084	-0. 0258	0. 6367	0. 0017	07. 40	-0. 01	0. 0030	0. 0025	0. 0014	180. 0	-0. 0024
074	0. 445	0. 699	02. 40	0. 7927	-0. 0199	0. 1176	-0. 8042	-0. 0198	0. 6282	0. 0017	06. 93	-0. 01	0. 0022	0. 0026	0. 0012	180. 0	-0. 0023
075	0. 445	0. 699	06. 39	0. 7741	-0. 0066	0. 1057	-0. 7814	-0. 0170	0. 5991	0. 0019	06. 39	-0. 01	0. 0025	0. 0025	0. 0014	180. 0	-0. 0026
076	0. 445	0. 699	06. 89	0. 7450	0. 0167	0. 0944	-0. 7509	-0. 0153	0. 5550	0. 0019	05. 89	-0. 01	0. 0040	0. 0025	0. 0013	180. 0	-0. 0041
077	0. 445	0. 699	05. 37	0. 7073	0. 0169	0. 0834	-0. 7122	-0. 0150	0. 5002	0. 0017	05. 37	-0. 01	0. 0031	0. 0025	0. 0012	180. 0	-0. 0032
078	0. 445	0. 700	04. 24	0. 6129	0. 0281	0. 0657	-0. 6172	-0. 0172	0. 3768	0. 0018	04. 34	-0. 01	0. 0025	0. 0026	0. 0014	180. 0	-0. 0040
079	0. 445	0. 702	03. 27	0. 4970	0. 0283	0. 0527	-0. 4993	-0. 0225	0. 2469	0. 0017	03. 27	-0. 01	0. 0040	0. 0027	0. 0013	180. 0	-0. 0041
080	0. 445	0. 700	01. 16	0. 2702	0. 0216	0. 0409	-0. 2711	-0. 0139	0. 0729	0. 0016	01. 16	-0. 01	0. 0019	0. 0022	0. 0010	180. 0	-0. 0040
081	0. 445	0. 698	00. 95	0. 0290	0. 0410	0. 0375	-0. 0384	-0. 0369	0. 0014	0. 0014	-00. 95	-0. 01	0. 0051	0. 0026	0. 0010	180. 0	-0. 0032
082	0. 445	0. 699	-03. 06	-0. 1904	0. 0537	0. 0423	0. 1921	-0. 0308	0. 0361	0. 0013	-03. 06	-0. 01	0. 0062	0. 0026	0. 0011	180. 0	-0. 0063
087	0. 444	0. 500	07. 26	0. 7691	-0. 0190	0. 1212	-0. 7786	-0. 0206	0. 5916	0. 0019	07. 26	-0. 01	0. 0023	0. 0022	0. 0014	180. 0	-0. 0024

TABLE 3(C)  
FLARE PACK ON  
ETA = 0° ETAT(T) = -1 1/2°

SER REVN.	MACH.	INCID.	LIFT.	PITCH.	DRAG	NORMAL	AXIAL.	CLSQ.	BASE.	RINC.	SLIP.	CROSS.	VAN H.	ROLL H.	RAND.	SIDE F.
068	0.444	0.501	06.78	0.7413	-0.0035	0.1021	-0.7494	-0.0129	0.9495	0.0020	0.78	-00.01	0.0023	0.0027	0.0019	180.0 -0.0024
089	0.444	0.502	06.27	0.7079	0.0076	0.0910	-0.7137	-0.0111	0.5009	0.0021	0.27	-00.01	0.0026	0.0016	0.0016	180.0 -0.0021
090	0.444	0.500	05.76	0.8631	0.0146	0.0792	-0.6678	-0.0104	0.4295	0.0019	0.76	-00.01	0.0025	0.0026	0.0016	180.0 -0.0026
091	0.444	0.499	05.24	0.6346	0.0182	0.0701	-0.6188	-0.0116	0.3776	0.0021	0.24	-00.01	0.0021	0.0026	0.0018	180.0 -0.0032
092	0.444	0.500	04.21	0.5258	0.0212	0.0573	-0.5287	-0.0166	0.2763	0.0020	0.21	-00.01	0.0033	0.0026	0.0022	180.0 -0.0034
093	0.444	0.499	03.17	0.4305	0.0244	0.0484	-0.4327	-0.0226	0.1852	0.0020	0.17	-00.01	0.0039	0.0026	0.0019	180.0 -0.0040
094	0.444	0.501	01.10	0.22351	0.0313	0.0183	-0.2259	-0.0221	0.0551	0.0017	0.10	-00.01	0.0044	0.0024	0.0013	180.0 -0.0049
095	0.444	0.499	-00.97	0.0351	0.0418	0.0361	-0.0306	-0.0351	0.0009	0.0016	0.97	-00.01	0.0055	0.0026	0.0013	180.0 -0.0056
096	0.444	0.500	-03.04	-0.1706	0.0551	0.0291	0.1723	-0.0287	0.0289	0.0014	-03.04	-00.01	0.0069	0.0030	0.0014	180.0 -0.0070

TABLE 3(D)  
FLARE PACK ON  
ETA = 0°(EXTRACT) = +1/2°

SER REVN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	ANIML.	CLSA.	BASE.	RINC.	SLIP.	CROSS.	YAW H.	ROLL H.	RAND.	SIDE F
002 0. 447 0. 904 02. 43 0. 8535 -0. 0994 0. 1685 -0. 0683 -0. 0544 0. 7284 0. 0023 07. 43 -00. 01 -0. 0035 -0. 0003 -0. 0001 179. 9 0. 0034																
003 0. 447 0. 904 06. 90 0. 8204 -0. 1031 0. 1583 -0. 0333 -0. 0532 0. 6730 0. 0024 06. 90 -00. 01 -0. 0030 -0. 0005 -0. 0002 179. 9 0. 0029																
004 0. 447 0. 900 06. 38 0. 7823 -0. 1060 0. 1419 -0. 7933 -0. 0517 0. 6119 0. 0024 06. 38 -00. 00 -0. 0045 -0. 0015 -0. 0013 179. 9 0. 0044																
005 0. 447 0. 904 05. 85 0. 7511 -0. 1111 0. 1313 -0. 7607 -0. 0513 0. 5641 0. 0027 05. 85 -00. 00 -0. 0046 -0. 0011 -0. 0006 179. 9 0. 0055																
006 0. 449 0. 904 05. 33 0. 7160 -0. 1094 0. 1207 -0. 7243 -0. 0512 0. 5126 0. 0024 05. 33 -00. 00 -0. 0046 -0. 0011 -0. 0006 179. 9 0. 0048																
007 0. 447 0. 900 04. 29 0. 6354 -0. 0980 0. 1009 -0. 6412 -0. 0503 0. 4036 0. 0024 04. 29 -00. 00 -0. 0036 -0. 0003 0. 0004 180. 0 0. 0035																
008 0. 449 0. 904 03. 24 0. 5424 -0. 0901 0. 0841 -0. 5464 -0. 0510 0. 2940 0. 0025 03. 24 -00. 00 -0. 0030 -0. 0008 0. 0010 180. 0 0. 0029																
009 0. 447 0. 900 03. 13 0. 3207 -0. 0682 0. 0601 -0. 3219 -0. 0519 0. 1027 0. 0023 03. 13 -00. 00 -0. 0017 -0. 0004 0. 0014 180. 0 0. 0016																
010 0. 447 0. 900 04. 01 0. 0537 -0. 0348 0. 0499 -0. 0530 -0. 0488 0. 0028 0. 0024 -01. 01 -00. 01 0. 0002 0. 0000 0. 0017 180. 0 0. 0003																
011 0. 447 0. 899 03. 15 -0. 2488 0. 0427 0. 0587 0. 2516 -0. 0431 0. 6618 0. 0019 -03. 15 -00. 01 0. 0025 0. 0004 0. 0019 180. 0 0. 0026																
014 0. 453 0. 879 07. 47 0. 8120 -0. 0363 0. 1562 -0. 8285 -0. 0472 0. 6592 0. 0024 07. 47 -00. 00 -0. 0063 -0. 0019 0. 0023 180. 0 0. 0067																
015 0. 453 0. 881 06. 91 0. 8140 -0. 0966 0. 1494 -0. 8262 -0. 0462 0. 6626 0. 0022 06. 91 -00. 00 -0. 0056 -0. 0012 -0. 0002 179. 9 0. 0055																
016 0. 453 0. 881 06. 39 0. 7819 -0. 0970 0. 1372 -0. 7923 -0. 0459 0. 6113 0. 0025 06. 39 -00. 00 -0. 0041 -0. 0007 -0. 0007 179. 9 0. 0040																
017 0. 453 0. 880 05. 87 0. 7478 -0. 0981 0. 1244 -0. 7567 -0. 0449 0. 5591 0. 0024 05. 87 -00. 00 -0. 0035 -0. 0011 -0. 0009 179. 9 0. 0034																
018 0. 453 0. 883 05. 35 0. 7100 -0. 0948 0. 1143 -0. 7177 -0. 0452 0. 3040 0. 0025 05. 35 -00. 00 -0. 0051 -0. 0012 -0. 0003 179. 9 0. 0050																
019 0. 450 0. 880 04. 31 0. 6369 -0. 0810 0. 0951 -0. 6424 -0. 0445 0. 4056 0. 0024 04. 31 -00. 00 -0. 0034 -0. 0009 0. 0002 180. 0 0. 0023																
020 0. 453 0. 881 03. 27 0. 5480 -0. 0703 0. 0798 -0. 5517 -0. 0463 0. 3001 0. 0022 03. 27 -00. 00 -0. 0030 -0. 0005 0. 0013 180. 0 0. 0029																
021 0. 450 0. 880 04. 16 0. 3419 -0. 0518 0. 0584 -0. 3431 -0. 0463 0. 1168 0. 0022 01. 16 -00. 01 -0. 0018 0. 0000 0. 0014 180. 0 0. 0017																
022 0. 453 0. 881 00. 98 0. 0739 -0. 0248 0. 0432 -0. 0732 -0. 0426 0. 0053 0. 0019 -00. 98 -00. 01 0. 0007 0. 0001 0. 0017 180. 0 0. 0006																
023 0. 450 0. 879 -03. 15 -0. 2442 0. 0109 0. 0517 0. 2465 -0. 0363 0. 0395 0. 0019 -03. 15 -00. 01 0. 0022 0. 0002 0. 0016 180. 0 0. 0021																
026 0. 452 0. 860 07. 40 0. 8110 -0. 1012 0. 1539 -0. 8242 -0. 0460 0. 6576 0. 0022 07. 40 -00. 00 -0. 0019 -0. 0012 0. 0053 180. 0 0. 0038																
027 0. 452 0. 859 06. 94 0. 7874 -0. 0403 0. 1393 -0. 7986 -0. 0409 0. 6200 0. 0022 06. 94 -00. 00 -0. 0064 -0. 0015 0. 0021 180. 0 0. 0062																
028 0. 452 0. 864 06. 29 0. 7908 -0. 0952 0. 1229 -0. 8008 -0. 0419 0. 6252 0. 0024 06. 29 -00. 00 -0. 0052 -0. 0011 -0. 0003 179. 9 0. 0054																
029 0. 452 0. 861 05. 86 0. 7576 -0. 0900 0. 1208 -0. 7661 -0. 0404 0. 5739 0. 0022 05. 86 -00. 00 -0. 0037 -0. 0011 -0. 0003 179. 9 0. 0036																
030 0. 452 0. 860 05. 36 0. 7197 -0. 0820 0. 1089 -0. 7268 -0. 0390 0. 5176 0. 0022 05. 36 -00. 00 -0. 0019 -0. 0003 -0. 0003 179. 9 0. 0028																
031 0. 455 0. 864 04. 84 0. 6822 -0. 0756 0. 0997 -0. 6883 -0. 0394 0. 4653 0. 0023 04. 84 -00. 00 -0. 0041 -0. 0010 0. 0002 180. 0 0. 0040																
032 0. 452 0. 860 04. 33 0. 6444 -0. 0666 0. 0908 -0. 6495 -0. 0398 0. 4151 0. 0024 04. 33 -00. 00 -0. 0018 -0. 0007 0. 0004 180. 0 0. 0027																
033 0. 452 0. 860 03. 29 0. 5607 -0. 0540 0. 0741 -0. 5641 -0. 0396 0. 3142 0. 0022 03. 29 -00. 00 -0. 0026 -0. 0005 0. 0011 180. 0 0. 0025																
034 0. 452 0. 859 04. 18 0. 3527 -0. 0349 0. 0527 -0. 3538 -0. 0432 0. 1243 0. 0022 01. 18 -00. 01 -0. 0016 0. 0001 0. 0016 180. 0 0. 0015																
035 0. 452 0. 858 -00. 97 0. 0712 -0. 0108 0. 0406 -0. 0706 -0. 0399 0. 0050 0. 0019 -00. 97 -00. 01 0. 0007 0. 0002 0. 0015 180. 0 -0. 0008																
036 0. 452 0. 859 -03. 14 -0. 2314 0. 0074 0. 0478 0. 2336 -0. 0323 0. 0534 0. 0020 -03. 14 -00. 01 0. 0013 0. 0002 0. 0017 180. 0 -0. 0016																
037 0. 450 0. 842 07. 39 0. 8109 -0. 1032 0. 1484 -0. 8234 -0. 0408 0. 6374 0. 0020 07. 39 -00. 00 -0. 0013 -0. 0011 0. 0041 180. 0 0. 0038																
038 0. 450 0. 849 06. 69 0. 7911 -0. 0866 0. 1378 -0. 8021 -0. 0396 0. 6258 0. 0021 06. 69 -00. 00 -0. 0029 -0. 0008 0. 0034 180. 0 0. 0028																
039 0. 450 0. 841 06. 42 0. 7819 -0. 0566 0. 1252 -0. 7914 -0. 0349 0. 6112 0. 0024 06. 42 -00. 00 -0. 0051 -0. 0012 0. 0016 180. 0 0. 0050																
040 0. 450 0. 840 05. 88 0. 7678 -0. 0812 0. 1479 -0. 7760 -0. 0364 0. 5894 0. 0022 05. 88 -00. 00 -0. 0019 -0. 0010 -0. 0004 179. 9 0. 0038																
043 0. 450 0. 841 05. 37 0. 7319 -0. 0718 0. 1053 -0. 7386 -0. 0341 0. 5356 0. 0024 05. 37 -00. 00 -0. 0043 -0. 0010 0. 0002 180. 0 0. 0042																

TABLE 3(D)  
FLARE PACK DM  
ETA = 0° ETACT = +1/2°

SER	REVN.	INCH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLOSE.	BASE.	ALNG.	SLIP.	CROSS.	VAN M.	ROLL M.	RANG.	SIDE F
044	0.450	0.844	04.	.34	0.6582	-0.0551	0.0658	-0.6629	-0.0345	0.4331	0.0028	04.	.34	00.	00.	-0.0037	0.0007
045	0.449	0.819	03.	.30	0.5756	-0.0403	0.0741	-0.5788	-0.0357	0.3311	0.0024	03.	.30	-00.	01	-0.0021	0.0002
046	0.450	0.819	01.	.19	0.3465	-0.0187	0.0480	-0.3478	-0.0387	0.1200	0.0021	01.	.19	-00.	01	-0.0011	0.0015
047	0.450	0.840	-00.	.97	0.0679	-0.0042	0.2388	-0.0673	-0.0381	0.0045	0.0019	-00.	.97	-00.	14	-0.0001	0.0015
048	0.450	0.839	-03.	.13	-0.2174	0.0054	0.0447	0.2194	-0.0341	0.0471	0.0016	-03.	.13	00.	00.	0.0013	0.0017
051	0.447	0.804	07.	.39	0.8255	-0.0921	0.1436	-0.8372	-0.0341	0.6813	0.0020	07.	.39	00.	00.	-0.0046	0.0012
052	0.447	0.793	06.	.80	0.8023	-0.0859	0.1312	-0.8124	-0.0322	0.6436	0.0019	06.	.88	00.	00.	-0.0038	0.0022
053	0.450	0.800	06.	.40	0.7234	-0.0554	0.1201	-0.7821	-0.0311	0.5961	0.0024	06.	.40	00.	00.	-0.0024	0.0040
054	0.447	0.804	05.	.89	0.7626	-0.0568	0.1078	-0.7697	-0.0270	0.5815	0.0020	05.	.89	00.	00.	-0.0027	0.0027
055	0.450	0.800	05.	.39	0.7555	-0.0495	0.1003	-0.7617	-0.0268	0.5707	0.0024	05.	.39	00.	00.	-0.0040	0.0006
056	0.447	0.799	04.	.37	0.6851	-0.0271	0.0297	-0.6893	-0.0250	0.4693	0.0028	04.	.37	-00.	01	-0.0045	0.0004
057	0.447	0.793	03.	.32	0.5804	-0.0131	0.0624	-0.5821	-0.0267	0.3368	0.0019	03.	.32	-00.	01	-0.0032	0.0002
058	0.450	0.804	01.	.18	0.3201	-0.0054	0.0445	-0.3214	-0.0237	0.1039	0.0020	04.	.18	-00.	01	-0.0003	0.0003
059	0.450	0.800	-00.	.97	0.0596	0.0019	0.3385	-0.0590	-0.0378	0.0034	0.0018	00.	.97	-00.	01	0.0000	0.0014
060	0.447	0.799	-02.	.11	-0.1986	0.0096	0.0408	0.2005	-0.0285	0.0393	0.0015	-03.	.11	00.	00.	-0.0010	0.0017
063	0.468	0.751	07.	.39	0.8286	-0.0815	0.1371	-0.8195	-0.0272	0.6865	0.0024	07.	.39	00.	00.	-0.0047	0.0005
064	0.468	0.749	06.	.91	0.7979	-0.0443	0.1231	-0.8070	-0.0242	0.6366	0.0020	06.	.91	00.	00.	-0.0048	0.0007
065	0.468	0.748	06.	.40	0.7935	-0.0537	0.1119	-0.8012	-0.0209	0.6236	0.0018	06.	.40	00.	00.	-0.0028	0.0003
066	0.468	0.750	05.	.89	0.7693	-0.0431	0.1021	-0.7756	-0.0205	0.5918	0.0024	05.	.99	00.	00.	-0.0019	0.0002
067	0.468	0.748	05.	.41	0.7266	0.0058	0.0879	-0.7317	-0.0169	0.5279	0.0020	05.	.41	00.	00.	-0.0014	0.0000
068	0.468	0.750	04.	.89	0.7490	-0.0111	0.0799	-0.7233	-0.0161	0.5169	0.0028	04.	.89	00.	00.	-0.0034	0.0004
069	0.468	0.751	04.	.37	0.6629	-0.0051	0.0705	-0.6624	-0.0177	0.4405	0.0020	04.	.37	-00.	01	-0.0025	0.0003
070	0.465	0.749	03.	.30	0.5419	-0.0022	0.0558	-0.5443	-0.0225	0.2935	0.0020	03.	.30	00.	00.	-0.0018	0.0002
071	0.468	0.751	04.	.16	0.2972	-0.0021	0.0416	-0.2982	-0.0337	0.0882	0.0019	04.	.16	-00.	01	-0.0008	0.0003
072	0.470	0.752	-00.	.97	0.0547	0.0047	0.0368	-0.0542	-0.0264	0.029	0.0017	-00.	.97	-00.	01	0.0003	0.0018
073	0.468	0.749	-02.	.10	-0.1846	0.0152	0.0418	-0.1865	-0.0202	0.0339	0.0017	-02.	.10	-00.	01	-0.0042	0.0003
076	0.449	0.699	07.	.39	0.8020	-0.0275	0.1213	-0.8123	-0.0250	0.6432	0.0020	07.	.39	-00.	01	-0.0028	0.0004
077	0.449	0.700	06.	.88	0.8009	-0.0379	0.1184	-0.8094	-0.0196	0.6113	0.0024	06.	.88	-00.	01	-0.0020	0.0002
078	0.449	0.699	06.	.37	0.7847	-0.0357	0.1053	-0.7916	-0.0156	0.6156	0.0019	05.	.37	-00.	01	-0.0015	0.0002
079	0.445	0.698	05.	.87	0.7576	-0.0246	0.0943	-0.7634	-0.0144	0.5738	0.0019	05.	.87	-00.	01	-0.0016	0.0003
080	0.449	0.700	05.	.36	0.7198	-0.0153	0.0832	-0.7244	-0.0137	0.5178	0.0020	05.	.36	-00.	01	-0.0035	0.0000
081	0.449	0.700	04.	.31	0.6216	-0.0050	0.0661	-0.6249	-0.0170	0.3862	0.0028	04.	.31	-00.	01	-0.0020	0.0002
082	0.449	0.700	03.	.25	0.6091	-0.0045	0.0527	-0.6114	-0.0217	0.2590	0.0020	03.	.25	-00.	01	-0.0017	0.0003
083	0.449	0.700	04.	.14	0.2826	-0.0015	0.0415	-0.2836	-0.0341	0.0798	0.0019	04.	.14	-00.	01	-0.0012	0.0001
084	0.449	0.699	-00.	.97	0.0518	0.0070	0.0364	-0.0543	-0.0156	0.0026	0.0017	-00.	.97	-00.	01	0.0002	0.0001
085	0.449	0.699	-03.	.08	-0.1764	0.0185	0.0398	-0.1782	-0.0287	0.0309	0.0017	-03.	.08	-00.	01	0.0007	0.0000

TABLE 3(D)  
FLARE PACK ON  
ETA = 0° EXTRACT = +1/2"

SER	REYN.	MACH.	LIFT.	PITCH.	DRAE	NORMA	AXIAL.	CLSR.	BASE.	RING.	SLIP.	CROSS.	YAN M.	ROLL M	RAND.	SIDE F
088	0.444	0.499	.07. 27	0. 7794	-0. 0472	0. 1183	-0. 7982	-0. 0168	0. 6074	0. 0020	.07. 27	-10. 04	-0. 0032	0. 0005	0. 0028	180. 0
089	0.444	0.500	.06. 76	0. 7534	-0. 0229	0. 1042	-0. 7606	-0. 0124	0. 5676	0. 0023	.06. 76	-10. 04	-0. 0032	0. 0005	0. 0026	180. 0
090	0.444	0.501	.06. 26	0. 7448	-0. 0229	0. 0904	-0. 7205	-0. 0098	0. 5109	0. 0021	.06. 26	-10. 04	-0. 0032	0. 0003	0. 0022	180. 0
091	0.444	0.501	.05. 74	0. 6757	-0. 0466	0. 0799	-0. 6805	-0. 0097	0. 4565	0. 0024	.05. 74	-10. 04	-0. 0033	0. 0003	0. 0021	180. 0
092	0.444	0.500	.05. 23	0. 6282	-0. 0127	0. 0708	-0. 6321	-0. 0110	0. 3945	0. 0023	.05. 23	-10. 04	-0. 0022	0. 0004	0. 0023	180. 0
093	0.444	0.499	.04. 19	0. 5382	-0. 0092	0. 0575	-0. 5410	-0. 0158	0. 2893	0. 0022	.04. 19	-10. 04	-0. 0021	0. 0002	0. 0028	180. 0
094	0.444	0.498	.03. 16	0. 4414	-0. 0064	0. 0483	-0. 4438	-0. 0218	0. 1947	0. 0021	.03. 16	-10. 04	-0. 0011	0. 0004	0. 0024	180. 0
095	0.444	0.500	.04. 09	0. 2486	0. 0003	0. 0386	-0. 2494	-0. 0319	0. 0617	0. 0024	.04. 09	-10. 04	-0. 0005	0. 0003	0. 0018	180. 0
096	0.444	0.501	.00. 98	0. 0449	0. 0100	0. 0357	-0. 0444	-0. 0345	0. 0020	0. 0020	.00. 98	-10. 04	0. 0004	0. 0002	0. 0017	180. 0
097	0.444	0.500	-.03. 05	-0. 1566	0. 0224	0. 0378	0. 1583	-0. 0277	0. 0244	0. 0016	-.03. 05	-10. 04	0. 0012	0. 0001	0. 0019	180. 0

TABLE 3(E)  
FLARE PACK ON  
ETA = 0°(ETRAT) = +1 1/2°

SER	REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORML.	AXIAL.	CLSQ.	BASE.	MINC.	SLIP.	CROSS.	VAN H.	ROLL H.	RANS.	SIDE F
002	0.455	0.900	02. 42	0. 8529	-0. 1082	0. 1699	-0. 0675	-0. 0360	0. 7274	0. 0023	07. 42	-00. 01	-0. 0010	0. 0011	-0. 0004	179. 9	0. 0010
003	0.455	0.899	06. 90	0. 8209	-0. 1144	0. 1582	-0. 0337	-0. 0533	0. 6737	0. 0026	06. 90	-00. 01	-0. 0001	0. 0012	-0. 0006	179. 9	0. 0006
004	0.455	0.893	06. 37	0. 7865	-0. 1200	0. 1442	-0. 0326	-0. 0520	0. 6180	0. 0024	06. 37	-00. 01	-0. 0002	0. 0011	-0. 0019	179. 9	0. 0001
005	0.455	0.893	05. 84	0. 7528	-0. 1222	0. 1202	-0. 0302	-0. 0504	0. 5667	0. 0025	05. 84	-00. 01	-0. 0021	0. 0017	-0. 0012	179. 9	-0. 0022
006	0.455	0.893	05. 32	0. 7203	-0. 1255	0. 1196	-0. 0283	-0. 0499	0. 5187	0. 0024	05. 32	-00. 01	-0. 0004	0. 0014	-0. 0010	179. 9	-0. 0005
007	0.455	0.895	04. 27	0. 6459	-0. 1262	0. 1006	-0. 0517	-0. 0500	0. 4121	0. 0022	04. 27	-00. 01	-0. 0005	0. 0014	-0. 0001	180. 0	-0. 0006
008	0.455	0.899	03. 22	0. 5934	-0. 1169	0. 0845	-0. 0574	-0. 0511	0. 3061	0. 0021	01. 22	-00. 01	-0. 0010	0. 0006	-0. 0006	180. 0	-0. 0011
009	0.457	0.900	01. 11	0. 2300	-0. 0960	0. 0601	-0. 2312	-0. 0516	0. 1087	0. 0021	01. 11	-00. 01	-0. 0037	0. 0022	-0. 0006	180. 0	-0. 0016
010	0.457	0.900	00. 92	0. 0604	-0. 0586	0. 0484	-0. 0537	-0. 0474	0. 0026	0. 0021	01. 03	-00. 01	-0. 0051	0. 0023	-0. 0012	180. 0	-0. 0052
011	0.457	0.904	00. 16	-0. 2415	-0. 0074	0. 0570	0. 2442	-0. 0418	0. 0391	0. 0018	-01. 16	-00. 02	-0. 0067	0. 0026	-0. 0014	180. 0	-0. 0068
012	0.457	0.873	07. 44	0. 8277	-0. 0706	0. 1562	-0. 0411	-0. 0435	0. 6850	0. 0024	07. 44	-00. 01	-0. 0015	0. 0013	-0. 0012	180. 0	0. 0018
013	0.457	0.875	06. 90	0. 8191	-0. 1095	0. 1482	-0. 0340	-0. 0465	0. 6707	0. 0023	06. 90	-00. 01	-0. 0008	0. 0014	-0. 0005	179. 9	0. 0007
014	0.455	0.890	06. 38	0. 7864	-0. 1128	0. 1283	-0. 7867	-0. 0448	0. 6183	0. 0023	06. 38	-00. 01	-0. 0005	0. 0012	-0. 0006	180. 0	-0. 0025
015	0.458	0.884	05. 85	0. 7831	-0. 1167	0. 1240	-0. 7620	-0. 0442	0. 5672	0. 0024	05. 85	-00. 01	-0. 0010	0. 0015	-0. 0012	179. 9	-0. 0011
016	0.456	0.861	05. 32	0. 7202	-0. 1193	0. 1126	-0. 7277	-0. 0439	0. 5186	0. 0023	05. 32	-00. 01	-0. 0004	0. 0011	-0. 0005	179. 9	-0. 0006
017	0.459	0.880	04. 29	0. 6460	-0. 1084	0. 0558	-0. 6310	-0. 0451	0. 4172	0. 0024	04. 29	-00. 01	-0. 0019	0. 0013	-0. 0012	180. 0	0. 0019
018	0.458	0.860	03. 25	0. 5598	-0. 0986	0. 0793	-0. 5635	-0. 0453	0. 3132	0. 0023	03. 25	-00. 02	-0. 0025	0. 0021	-0. 0008	180. 0	-0. 0026
019	0.458	0.873	03. 14	0. 3010	-0. 0784	0. 0549	-0. 2821	-0. 0462	0. 1230	0. 0018	01. 14	-00. 02	-0. 0024	0. 0028	-0. 0010	180. 0	-0. 0025
020	0.458	0.873	03. 00	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0026
021	0.458	0.873	02. 86	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0027
022	0.458	0.873	02. 71	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0027
023	0.459	0.879	02. 16	-0. 2338	-0. 0124	0. 0511	0. 2261	-0. 0362	0. 0048	0. 0029	-01. 16	-00. 02	-0. 0072	0. 0023	-0. 0011	180. 0	-0. 0072
024	0.460	0.861	02. 40	0. 6125	-0. 2014	0. 1520	-0. 8235	-0. 0446	0. 6601	0. 0023	07. 40	-00. 01	-0. 0082	0. 0008	-0. 0005	180. 0	0. 0001
025	0.460	0.861	02. 25	0. 7984	-0. 0556	0. 1378	-0. 8073	-0. 0382	0. 6374	0. 0022	06. 53	-00. 01	-0. 0085	0. 0013	-0. 0007	180. 0	0. 0008
026	0.460	0.861	02. 14	0. 3010	-0. 0784	0. 0549	-0. 2821	-0. 0462	0. 1230	0. 0018	01. 14	-00. 02	-0. 0024	0. 0028	-0. 0010	180. 0	-0. 0028
027	0.460	0.861	02. 00	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
028	0.460	0.861	01. 96	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
029	0.460	0.861	01. 91	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
030	0.460	0.861	01. 86	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
031	0.460	0.861	01. 81	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
032	0.460	0.861	01. 77	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
033	0.460	0.861	01. 72	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
034	0.460	0.861	01. 67	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
035	0.460	0.861	01. 62	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
036	0.460	0.861	01. 57	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
037	0.460	0.861	01. 52	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
038	0.460	0.861	01. 47	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
039	0.460	0.861	01. 42	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
040	0.460	0.861	01. 37	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0029
041	0.462	0.819	01. 32	0. 7986	-0. 0556	0. 1378	-0. 8073	-0. 0382	0. 6271	0. 0019	06. 91	-00. 01	-0. 0019	0. 0019	-0. 0028	180. 0	-0. 0028
042	0.462	0.819	01. 27	0. 3010	-0. 0784	0. 0549	-0. 2821	-0. 0462	0. 1230	0. 0019	06. 91	-00. 01	-0. 0020	0. 0020	-0. 0028	180. 0	-0. 0028
043	0.462	0.819	01. 22	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
044	0.462	0.819	01. 17	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
045	0.462	0.819	01. 12	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
046	0.462	0.819	01. 07	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
047	0.462	0.819	01. 02	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
048	0.462	0.819	00. 97	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
049	0.462	0.819	00. 92	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
050	0.462	0.819	00. 87	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
051	0.462	0.819	00. 82	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
052	0.462	0.819	00. 77	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024	-0. 0011	179. 9	-0. 0028
053	0.462	0.819	00. 72	0. 0624	-0. 0502	0. 0426	-0. 0827	-0. 0433	0. 0020	0. 0019	01. 00	-00. 02	-0. 0046	0. 0024			

TABLE 3(E)  
FLARE PACK ON  
ETA = 0° EXTRACT = +1 1/2"

SER	REVN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	BASE.	AINC.	SLIP.	CROSS.	YAN N.	ROLL N.	RANG.	SIDE F	
046	0	462	0	041	04.	.32	0.	.6683	-0.	.0617	0.	.0664	-0.	.6730	-0.	.0336	0.	.0002
047	0	462	0	040	03.	.29	0.	.5855	-0.	.0677	0.	.0656	-0.	.5866	-0.	.0343	0.	.0003
048	0	462	0	040	04.	.17	0.	.3588	-0.	.0459	0.	.0491	-0.	.3598	-0.	.0388	0.	.0019
049	0	462	0	040	-00.	.29	0.	.0764	-0.	.0294	0.	.0389	-0.	.0758	-0.	.0384	0.	.0032
050	0	462	0	039	-03.	.15	-0.	.2079	-0.	.0183	0.	.0430	0.	.2093	-0.	.0299	0.	.0043
053	0	455	0	001	02.	.38	0.	.8129	-0.	.1134	0.	.1450	-0.	.0447	-0.	.0149	0.	.0003
054	0	455	0	799	06.	.87	0.	.8112	-0.	.1077	0.	.1319	-0.	.8213	-0.	.0317	0.	.0002
055	0	457	0	798	06.	.37	0.	.7919	-0.	.0990	0.	.1214	-0.	.8006	-0.	.0308	0.	.0020
056	0	458	0	802	05.	.87	0.	.7846	-0.	.0908	0.	.1108	-0.	.7919	-0.	.0278	0.	.0011
057	0	457	0	798	05.	.37	0.	.7663	-0.	.0755	0.	.0955	-0.	.7726	-0.	.0252	0.	.0014
058	0	455	0	799	04.	.35	0.	.6358	-0.	.0540	0.	.0802	-0.	.7000	-0.	.0250	0.	.0018
059	0	455	0	800	03.	.30	0.	.5901	-0.	.0394	0.	.0626	-0.	.5928	-0.	.0267	0.	.0016
060	0	455	0	800	03.	.16	0.	.3295	-0.	.0301	0.	.0429	-0.	.3304	-0.	.0345	0.	.0015
061	0	455	0	799	-00.	.99	0.	.0708	-0.	.0228	0.	.0365	-0.	.0702	-0.	.0359	0.	.0014
062	0	458	0	799	-03.	.13	-0.	.1898	-0.	.0147	0.	.0402	0.	.1916	-0.	.0282	0.	.0012
062	0	473	0	751	02.	.38	0.	.8361	-0.	.1030	0.	.1377	-0.	.8469	-0.	.0272	0.	.0017
063	0	473	0	750	06.	.88	0.	.8173	-0.	.0837	0.	.1254	-0.	.8265	-0.	.0245	0.	.0016
064	0	473	0	750	06.	.39	0.	.7944	-0.	.0608	0.	.1137	-0.	.8023	-0.	.0225	0.	.0015
065	0	473	0	750	05.	.88	0.	.7817	-0.	.0688	0.	.1022	-0.	.7892	-0.	.0196	0.	.0014
066	0	473	0	751	05.	.40	0.	.7379	-0.	.0151	0.	.0865	-0.	.7430	-0.	.0168	0.	.0013
067	0	473	0	751	04.	.25	0.	.6777	-0.	.0308	0.	.0719	-0.	.6812	-0.	.0179	0.	.0012
068	0	473	0	749	02.	.28	0.	.5530	-0.	.0286	0.	.0569	-0.	.5555	-0.	.0230	0.	.0011
069	0	473	0	749	01.	.14	0.	.3074	-0.	.0270	0.	.0420	-0.	.3080	-0.	.0342	0.	.0010
070	0	473	0	750	-00.	.99	0.	.0660	-0.	.0191	0.	.0370	-0.	.0654	-0.	.0365	0.	.0009
071	0	473	0	750	-03.	.14	-0.	.1759	-0.	.0084	0.	.0407	0.	.1777	-0.	.0295	0.	.0008
072	0	455	0	704	07.	.36	0.	.8242	-0.	.0790	0.	.1330	-0.	.8346	-0.	.0245	0.	.0007
073	0	455	0	699	06.	.85	0.	.8175	-0.	.0794	0.	.1202	-0.	.8261	-0.	.0197	0.	.0006
074	0	455	0	699	06.	.36	0.	.7963	-0.	.0651	0.	.1063	-0.	.8032	-0.	.0157	0.	.0005
075	0	453	0	704	05.	.85	0.	.7701	-0.	.0520	0.	.0951	-0.	.7759	-0.	.0140	0.	.0004
076	0	453	0	700	05.	.34	0.	.7313	-0.	.0410	0.	.0840	-0.	.7360	-0.	.0137	0.	.0003
077	0	455	0	704	04.	.30	0.	.6364	-0.	.0302	0.	.0652	-0.	.6397	-0.	.0162	0.	.0002
078	0	455	0	700	03.	.24	0.	.5227	-0.	.0294	0.	.0533	-0.	.5250	-0.	.0158	0.	.0001
079	0	455	0	699	04.	.13	0.	.2927	-0.	.0293	0.	.0401	-0.	.2937	-0.	.0238	0.	.0001
080	0	457	0	700	-00.	.99	0.	.0627	-0.	.0169	0.	.0357	-0.	.6391	-0.	.0113	0.	.0000
081	0	455	0	699	-03.	.10	-0.	.1668	-0.	.0052	0.	.0397	-0.	.1686	-0.	.0292	0.	.0013
082	0	455	0	699	-03.	.15	-0.	.1668	-0.	.0052	0.	.0397	-0.	.0277	-0.	.0013	0.	.0003
083	0	450	0	499	07.	.26	0.	.7928	-0.	.0704	0.	.1231	-0.	.8021	-0.	.0197	0.	.0024
084	0	450	0	499	07.	.26	0.	.6284	-0.	.0022	0.	.0015	-0.	.6284	-0.	.0026	0.	.0004

TABLE 3(E)  
FLARE PACK ON  
ETRA = 0° EXTRACT = +1 1/2°

SER	REVN.	HACH.	INCID.	LIFT.	PITCH.	DRES.	MORHEL.	AXIAL.	CL.SQ.	BASE.	INC.	SLIP.	CROSS.	VAN H.	ROLL H.	RANS.	SIDE F
027	0.449	0.501	06.76	0.7689	-0.0582	0.1047	-0.7760	-0.0115	0.5941	0.0020	06.76	-10.04	0.0018	0.0020	180.0	-0.0018	
028	0.449	0.502	06.25	0.7321	-0.0471	0.0916	-0.7378	-0.0091	0.5359	0.0022	06.25	-10.04	0.0013	0.0024	180.0	-0.0014	
029	0.450	0.500	05.74	0.6903	-0.0405	0.0809	-0.6950	-0.0092	0.4763	0.0022	05.74	-10.04	0.0016	0.0024	180.0	-0.0017	
030	0.452	0.501	05.22	0.6439	-0.0365	0.0712	-0.6478	-0.0102	0.4145	0.0024	05.22	-10.04	0.0020	0.0024	180.0	-0.0021	
031	0.452	0.501	04.19	0.5516	-0.0326	0.0576	-0.5545	-0.0152	0.3042	0.0020	04.19	-10.04	0.0030	0.0024	180.0	-0.0031	
032	0.452	0.500	03.15	0.4551	-0.0294	0.0462	-0.4571	-0.0211	0.2070	0.0021	03.15	-10.04	0.0035	0.0023	180.0	-0.0036	
033	0.450	0.499	01.08	0.2607	-0.0217	0.0377	-0.2615	-0.0310	0.1678	0.0019	01.08	-10.04	0.0038	0.0023	180.0	-0.0039	
034	0.471	0.502	-00.99	0.0568	-0.0122	0.0345	-0.0560	-0.0136	0.0031	0.0019	00.99	-10.04	0.0042	0.0022	180.0	-0.0042	
035	0.471	0.499	-03.06	-0.1464	0.0001	0.0264	0.1481	-0.0268	0.0213	0.0018	-03.06	-10.04	0.0050	0.0022	180.0	-0.0051	

TABLE 3(F)  
FLARE PACK ON  
ETA = 0° ET(A1) = +2 1/2°

SER. NO.	REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAF.	NORML.	AXIAL.	CLSB.	BASE.	RINC.	SLIP.	CROSS.	VAN N.	ROLL M.	RANG.	SIDE F.
008	0.444	0.901	07.42	0.8538	-0.1115	0.1714	-0.8689	-0.0573	0.7288	0.0025	07.42	-10.04	-0.0042	-0.0004	0.0001	180.0	0.0041
019	0.444	0.898	06.89	0.8213	-0.1185	0.1552	-0.8241	-0.0532	0.6744	0.0023	06.89	-10.04	-0.0047	-0.0005	0.0001	179.9	0.0046
010	0.444	0.901	06.36	0.7861	-0.1240	0.1433	-0.7972	-0.0528	0.6179	0.0025	06.36	-10.00	-0.0043	-0.0009	0.0010	179.9	0.0042
011	0.444	0.902	05.84	0.7539	-0.1297	0.1320	-0.7635	-0.0519	0.5683	0.0027	05.84	-10.00	-0.0040	-0.0009	0.0005	179.9	0.0039
012	0.444	0.899	05.31	0.7223	-0.1383	0.1197	-0.7304	-0.0497	0.5217	0.0026	05.31	-10.04	-0.0035	-0.0005	0.0003	179.9	0.0034
013	0.444	0.899	04.26	0.6512	-0.1406	0.1008	-0.6370	-0.0497	0.4239	0.0024	04.26	-10.00	-0.0021	-0.0004	0.0008	180.0	0.0020
014	0.444	0.901	03.24	0.5603	-0.1372	0.0841	-0.5642	-0.0501	0.3139	0.0026	03.21	-10.01	-0.0006	0.0001	0.0013	180.0	0.0003
015	0.444	0.901	04.09	0.5367	-0.1142	0.0609	-0.3379	-0.0520	0.1133	0.0025	01.09	-10.01	-0.0007	0.0002	0.0019	180.0	0.0008
016	0.444	0.900	-01.04	0.0684	-0.0766	0.0494	-0.0676	-0.0483	0.0045	0.0024	-01.04	0.0022	0.0003	0.0022	0.0022	180.0	0.0023
017	0.444	0.901	-01.18	-0.2154	-0.0252	0.0586	0.2382	-0.0434	0.0532	0.0025	-01.18	0.0044	0.0005	0.0025	0.0025	180.0	0.0042
020	0.450	0.881	07.41	0.8300	-0.1079	0.1602	-0.8433	-0.0493	0.6886	0.0023	07.41	-10.00	-0.0058	-0.0010	0.0013	180.0	0.0037
021	0.450	0.880	06.89	0.7891	-0.1218	0.1356	-0.7995	-0.0479	0.6683	0.0023	06.89	-10.00	-0.0046	-0.0009	0.0005	180.0	0.0045
022	0.450	0.879	06.37	0.7691	-0.1269	0.1247	-0.7644	-0.0466	0.6227	0.0023	06.37	-10.01	-0.0028	-0.0004	0.0005	179.9	0.0027
023	0.450	0.880	05.84	0.7355	-0.1269	0.1247	-0.7644	-0.0466	0.5707	0.0026	05.84	-10.00	-0.0036	-0.0009	0.0006	179.9	0.0025
024	0.450	0.880	05.32	0.7209	-0.1277	0.1141	-0.7285	-0.0443	0.5196	0.0024	05.32	-10.00	-0.0038	-0.0007	0.0000	179.9	0.0037
025	0.453	0.881	04.28	0.6538	-0.1271	0.0962	-0.6593	-0.0448	0.4274	0.0024	04.28	-10.00	-0.0026	-0.0006	0.0003	180.0	0.0025
026	0.453	0.881	03.23	0.5664	-0.1164	0.0786	-0.5700	-0.0443	0.3206	0.0024	03.23	-10.00	-0.0015	-0.0002	0.0013	180.0	0.0014
027	0.453	0.880	01.32	0.3574	-0.0958	0.0561	-0.3866	-0.0468	0.1276	0.0023	01.32	-10.01	-0.0014	0.0002	0.0015	180.0	0.0010
028	0.453	0.881	-01.02	0.0870	-0.0870	0.0478	-0.0863	-0.0432	0.0074	0.0024	-01.02	-10.01	-0.0009	0.0004	0.0016	180.0	0.0010
029	0.449	0.876	-03.16	-0.2292	-0.0297	0.0498	0.2215	-0.0352	0.0523	0.0016	-03.16	-10.01	-0.0037	0.0006	0.0018	180.0	-0.0036
032	0.463	0.861	07.39	0.8202	-0.1173	0.1547	-0.8233	-0.0455	0.6726	0.0024	07.39	-10.00	-0.0037	-0.0005	0.0030	180.0	0.0036
033	0.463	0.861	06.92	0.8107	-0.0836	0.1437	-0.8282	-0.0428	0.6571	0.0022	06.92	-10.00	-0.0062	-0.0012	0.0007	180.0	0.0061
034	0.463	0.860	06.37	0.7293	-0.1216	0.1129	-0.8092	-0.0410	0.6387	0.0023	06.37	-10.00	-0.0037	-0.0007	0.0002	179.9	0.0036
035	0.463	0.859	05.85	0.7638	-0.1236	0.1198	-0.7781	-0.0395	0.5925	0.0022	05.85	-10.00	-0.0035	-0.0007	0.0002	179.9	0.0034
036	0.463	0.859	05.33	0.7341	-0.1234	0.1089	-0.7413	-0.0390	0.5389	0.0023	05.33	-10.00	-0.0039	-0.0008	0.0002	180.0	0.0036
037	0.463	0.860	04.29	0.6624	-0.1121	0.0899	-0.6674	-0.0379	0.4386	0.0022	04.29	-10.00	-0.0034	-0.0006	0.0006	180.0	0.0033
038	0.463	0.859	03.26	0.5800	-0.0973	0.0739	-0.5823	-0.0387	0.3363	0.0021	03.24	-10.00	-0.0022	-0.0004	0.0014	180.0	0.0021
040	0.462	0.862	01.14	0.2684	-0.0806	0.0523	-0.2690	-0.0437	0.1356	0.0023	01.14	-10.01	-0.0007	0.0004	0.0018	180.0	0.0006
040	0.462	0.862	01.14	0.2684	-0.0806	0.0533	-0.2695	-0.0437	0.1356	0.0023	01.14	-10.01	-0.0007	0.0004	0.0018	180.0	0.0006
041	0.463	0.860	-01.00	0.0868	-0.0534	0.0406	-0.0862	-0.0401	0.0074	0.0024	-01.00	-10.01	-0.0009	0.0003	0.0018	180.0	-0.0010
042	0.463	0.859	-03.17	-0.2171	-0.0322	0.0464	0.2193	-0.0326	0.0470	0.0019	-03.17	-10.01	-0.0029	-0.0004	0.0020	180.0	-0.0026
045	0.468	0.840	07.39	0.8228	-0.1164	0.1496	-0.8363	-0.0404	0.6785	0.0021	07.39	-10.00	-0.0035	-0.0007	0.0029	180.0	0.0034
046	0.458	0.841	06.88	0.8068	-0.1159	0.1374	-0.8176	-0.0376	0.6508	0.0022	06.88	-10.00	-0.0036	-0.0009	0.0030	180.0	0.0038
047	0.458	0.839	06.39	0.7930	-0.0954	0.1283	-0.8025	-0.0371	0.6288	0.0023	06.39	-10.00	-0.0043	-0.0008	0.0040	180.0	0.0040
048	0.458	0.829	05.86	0.7846	-0.1188	0.1182	-0.7926	-0.0353	0.6154	0.0028	05.86	-10.00	-0.0043	-0.0009	0.0032	179.9	0.0042
049	0.458	0.840	05.34	0.7484	-0.1147	0.1057	-0.7662	-0.0343	0.5616	0.0023	05.34	-10.00	-0.0043	-0.0006	0.0004	180.0	0.0042

TABLE 3(F)  
FLARE PACK DN  
ETA = 0°(ETRACT) = +2 1/2°

SER	REYN.	HACH.	INCID.	LIFT.	PITCH.	DRAG	NORMAL	AXIAL.	CLSQ.	BASE.	HTIC.	SLIP.	CROSS.	VAN H.	ROLL H.	RNG.	SIDE F
050	0. 458	0. 839	04. 34	0. 6763	-0. 0976	0. 0869	-0. 6910	-0. 0336	0. 4573	0. 0023	04. 31	00. 00	-0. 0024	-0. 0006	0. 0007	180. 0	0. 0023
051	0. 458	0. 639	03. 27	0. 5928	-0. 0840	0. 0711	-0. 5960	-0. 0350	0. 3513	0. 0021	03. 27	-00. 01	-0. 0020	-0. 0002	0. 0014	180. 0	0. 0019
052	0. 458	0. 840	04. 16	0. 3643	-0. 0628	0. 0486	-0. 3653	-0. 0390	0. 1325	0. 0022	01. 16	00. 00	-0. 0006	-0. 0001	0. 0016	180. 0	0. 0005
053	0. 458	0. 840	-04. 00	0. 0817	-0. 0458	0. 0387	-0. 0811	-0. 0383	0. 0066	0. 0019	-01. 00	-00. 01	0. 0003	0. 0002	0. 0015	180. 0	-0. 0004
054	0. 462	0. 844	-03. 16	-0. 2044	-0. 0352	0. 0434	0. 2064	-0. 0302	0. 0415	0. 0019	-03. 16	-00. 01	0. 0018	0. 0002	0. 0018	180. 0	-0. 0019
057	0. 458	0. 800	07. 36	0. 8270	-0. 1146	0. 1442	-0. 8388	-0. 0347	0. 6839	0. 0021	07. 36	00. 00	-0. 0023	-0. 0007	0. 0021	180. 0	0. 0022
058	0. 455	0. 801	06. 87	0. 8436	-0. 1087	0. 1316	-0. 8236	-0. 0310	0. 6618	0. 0023	06. 87	00. 00	-0. 0027	-0. 0004	0. 0021	180. 0	0. 0026
059	0. 458	0. 802	06. 37	0. 8003	-0. 1008	0. 1203	-0. 8088	-0. 0287	0. 6404	0. 0020	06. 37	00. 00	-0. 0032	-0. 0006	0. 0021	180. 0	0. 0031
060	0. 455	0. 799	05. 86	0. 7903	-0. 1034	0. 1101	-0. 7975	-0. 0268	0. 6245	0. 0020	05. 86	00. 00	-0. 0026	-0. 0007	0. 0021	180. 0	0. 0028
061	0. 458	0. 801	05. 36	0. 7754	-0. 0920	0. 1012	-0. 7816	-0. 0260	0. 6012	0. 0022	05. 36	00. 00	-0. 0036	-0. 0003	0. 0008	180. 0	0. 0028
062	0. 455	0. 801	04. 34	0. 7026	-0. 0712	0. 0808	-0. 7068	-0. 0251	0. 4934	0. 0022	04. 34	-00. 01	-0. 0036	-0. 0004	0. 0012	180. 0	0. 0025
063	0. 458	0. 800	03. 29	0. 5974	-0. 0537	0. 0621	-0. 6001	-0. 0237	0. 3568	0. 0020	01. 29	-00. 01	-0. 0019	-0. 0000	0. 0022	180. 0	0. 0018
064	0. 455	0. 799	04. 15	0. 3357	-0. 0460	0. 0438	-0. 3366	-0. 0351	0. 1126	0. 0020	01. 15	-00. 01	-0. 0010	-0. 0003	0. 0016	180. 0	0. 0010
065	0. 458	0. 801	-04. 00	0. 0754	-0. 0289	0. 0380	-0. 0745	-0. 0374	0. 0056	0. 0020	-01. 00	-00. 01	-0. 0008	0. 0001	0. 0014	180. 0	-0. 0009
066	0. 458	0. 801	-03. 14	-0. 1845	-0. 0310	0. 0401	0. 1863	-0. 0282	0. 0339	0. 0018	-03. 14	-00. 01	0. 0026	0. 0003	0. 0018	180. 0	-0. 0026
069	0. 453	0. 751	07. 36	0. 8337	-0. 1172	0. 1364	-0. 8444	-0. 0265	0. 6949	0. 0020	07. 36	00. 00	-0. 0034	-0. 0003	0. 0027	180. 0	0. 0023
070	0. 453	0. 750	06. 86	0. 8070	-0. 0927	0. 1236	-0. 8161	-0. 0242	0. 6512	0. 0020	06. 86	00. 00	-0. 0036	-0. 0003	0. 0019	180. 0	0. 0025
071	0. 453	0. 751	06. 37	0. 7970	-0. 0772	0. 1138	-0. 8048	-0. 0225	0. 6351	0. 0021	06. 37	00. 00	-0. 0014	-0. 0002	0. 0044	180. 0	0. 0013
072	0. 453	0. 750	05. 86	0. 7899	-0. 0827	0. 1039	-0. 7965	-0. 0207	0. 6239	0. 0020	05. 86	00. 00	-0. 0013	-0. 0001	0. 0041	180. 0	0. 0012
073	0. 453	0. 749	05. 36	0. 7648	-0. 0679	0. 0914	-0. 7701	-0. 0267	0. 5848	0. 0020	05. 36	00. 00	-0. 0019	-0. 0002	0. 0029	180. 0	0. 0018
074	0. 453	0. 749	04. 33	0. 6838	-0. 0466	0. 0717	-0. 6873	-0. 0179	0. 4674	0. 0020	04. 33	00. 00	-0. 0026	-0. 0005	0. 0020	180. 0	0. 0025
075	0. 453	0. 749	03. 27	0. 5589	-0. 0440	0. 0561	-0. 5613	-0. 0222	0. 3122	0. 0020	03. 27	00. 00	-0. 0016	-0. 0002	0. 0021	180. 0	0. 0015
076	0. 453	0. 748	04. 13	0. 3118	-0. 0420	0. 0412	-0. 3126	-0. 0334	0. 0971	0. 0019	01. 13	-00. 01	-0. 0006	0. 0002	0. 0044	180. 0	0. 0005
077	0. 453	0. 750	-04. 00	0. 0719	-0. 0341	0. 0360	-0. 0714	-0. 0355	0. 0051	0. 0017	-01. 00	-00. 01	-0. 0002	0. 0001	0. 0015	180. 0	0. 0004
078	0. 453	0. 750	-03. 12	-0. 1691	-0. 0238	0. 0395	0. 1709	-0. 0287	0. 0284	0. 0017	-03. 12	-00. 01	0. 0017	0. 0003	0. 0016	180. 0	-0. 0018
081	0. 444	0. 699	07. 35	0. 8321	-0. 0943	0. 1369	-0. 8429	-0. 0273	0. 6922	0. 0020	07. 35	-00. 01	-0. 0029	0. 0002	0. 0028	180. 0	0. 0028
082	0. 445	0. 700	06. 84	0. 8195	-0. 0944	0. 1243	-0. 8286	-0. 0237	0. 6715	0. 0020	06. 84	-00. 01	-0. 0025	0. 0002	0. 0047	180. 0	0. 0025
083	0. 445	0. 701	06. 36	0. 7832	-0. 0478	0. 1079	-0. 7905	-0. 0184	0. 6134	0. 0020	06. 36	00. 00	-0. 0010	0. 0003	0. 0047	180. 0	0. 0029
084	0. 445	0. 700	05. 84	0. 7734	-0. 0671	0. 0997	-0. 7796	-0. 0183	0. 5981	0. 0021	05. 84	00. 00	-0. 0010	0. 0004	0. 0043	180. 0	0. 0029
085	0. 445	0. 700	05. 33	0. 7372	-0. 0562	0. 0801	-0. 7423	-0. 0172	0. 5434	0. 0020	05. 33	-00. 01	-0. 0019	0. 0002	0. 0031	180. 0	0. 0018
086	0. 445	0. 699	04. 29	0. 6405	-0. 0458	0. 0692	-0. 6440	-0. 0191	0. 4101	0. 0020	04. 29	-00. 01	-0. 0020	0. 0005	0. 0027	180. 0	0. 0019
087	0. 445	0. 699	03. 23	0. 5258	-0. 0444	0. 0578	-0. 5284	-0. 0261	0. 2763	0. 0019	02. 23	-00. 01	-0. 0009	0. 0004	0. 0027	180. 0	0. 0008
088	0. 445	0. 700	04. 12	0. 2977	-0. 0401	0. 0449	-0. 2987	-0. 0367	0. 0885	0. 0020	04. 12	-00. 01	-0. 0008	0. 0003	0. 0016	180. 0	0. 0007
089	0. 445	0. 699	-04. 00	0. 0678	-0. 0312	0. 0380	-0. 0672	-0. 0374	0. 0045	0. 0018	-01. 00	-00. 01	-0. 0008	0. 0003	0. 0015	180. 0	-0. 0004
090	0. 445	0. 699	-03. 10	-0. 1610	-0. 0203	0. 0413	0. 1629	-0. 0310	0. 0258	0. 0017	-03. 10	-00. 01	-0. 0012	0. 0001	0. 0017	180. 0	-0. 0013
092	0. 445	0. 802	07. 26	0. 7980	-0. 0834	0. 1213	-0. 8070	-0. 0176	0. 6366	0. 0022	07. 26	-00. 01	-0. 0024	0. 0005	0. 0027	180. 0	0. 0023

TABLE 3(F)  
FLARE PACK ON  
ETR = 0° ETRACT = +2 1/2°

SER NO.	REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	BASE.	RING.	SLIP.	CROSS.	YAW M.	ROLL M.	RAND.	SIDE F
094	0.444	0.438	06.75	0.7727	-0.0694	0.1019	-0.7798	-0.0113	0.3969	0.0024	06.75	-00.01	-0.0030	0.0005	0.0024	180.0	0.0029
095	0.444	0.439	06.24	0.7781	-0.0594	0.0922	-0.7441	-0.0091	0.5451	0.0023	06.24	-00.01	-0.0032	0.0004	0.0023	180.0	0.0031
096	0.445	0.435	05.72	0.6919	-0.0538	0.0804	-0.6966	-0.0068	0.4787	0.0024	05.73	-00.01	-0.0030	0.0004	0.0022	180.0	0.0029
097	0.444	0.500	05.21	0.6489	-0.0507	0.0718	-0.6528	-0.0104	0.4210	0.0022	05.21	-00.01	-0.0020	0.0005	0.0025	180.0	0.0019
098	0.444	0.501	04.16	0.5581	-0.0467	0.0580	-0.5609	-0.0151	0.3113	0.0024	04.16	-00.01	-0.0014	0.0005	0.0027	180.0	0.0013
099	0.444	0.500	03.15	0.4629	-0.0432	0.0487	-0.4680	-0.0213	0.2142	0.0020	03.15	-00.01	-0.0011	0.0007	0.0024	180.0	0.0010
100	0.444	0.501	04.08	0.2654	-0.0350	0.0381	-0.2662	-0.0211	0.0703	0.0021	04.08	-00.01	-0.0006	0.0004	0.0018	180.0	0.0005
101	0.445	0.502	-04.00	0.0626	-0.0254	0.0344	-0.0621	-0.0336	0.0038	0.0019	-01.00	-00.01	-0.0005	0.0002	0.0016	180.0	-0.0013
102	0.444	0.499	-03.07	-0.1397	-0.0134	0.0364	0.1413	-0.0270	0.0194	0.0019	-03.07	-00.01	0.0012	0.0004	0.0018	180.0	-0.0018

TABLE 3(G)  
FLARE PACK OH  
ETA = 0° EXTRACT = +3 1/2°

SER. REVN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	BASE.	RING.	SLIP.	CROSS.	VAN H.	ROLL H.	RANG.	SIDE F	
002	0.457	0.900	07.42	0.8516	-0.1158	0.1698	-0.8663	0.7251	0.0024	07.42	-10.01	-0.0006	0.0012	-0.0003	179.9	0.0005	
003	0.457	0.901	06.89	0.8185	-0.1213	0.1560	-0.8214	0.6941	0.0025	06.89	-10.01	0.0000	0.0013	-0.0002	179.9	0.0000	
004	0.457	0.901	06.36	0.7837	-0.1282	0.1420	-0.7947	0.515	0.0027	06.36	-10.01	-0.0007	0.0009	-0.0012	179.9	0.0006	
005	0.457	0.900	05.93	0.7512	-0.1344	0.1305	-0.7607	0.511	0.0024	05.93	-10.01	-0.0008	0.0008	-0.0008	179.9	0.0007	
006	0.457	0.900	05.34	0.7224	-0.1399	0.1210	-0.7306	0.510	0.0025	05.34	-10.01	-0.0003	0.0014	-0.0007	179.9	0.0002	
007	0.457	0.900	04.25	0.6528	-0.1527	0.1020	-0.6587	0.510	0.4260	0.0023	04.25	-10.01	0.0005	0.0017	0.0003	180.0	-0.0006
008	0.457	0.899	03.19	0.5690	-0.1645	0.0854	-0.5229	0.514	0.3235	0.0022	03.19	-10.01	0.0024	0.0021	0.0007	180.0	-0.0025
009	0.457	0.900	04.06	0.3451	-0.1496	0.0611	-0.3462	0.524	0.1190	0.0024	01.06	-10.02	0.0038	0.0024	0.0009	180.0	-0.0039
010	0.457	0.900	01.07	0.0768	-0.1096	0.0498	-0.0759	0.489	0.0058	0.0024	-01.07	-10.02	0.0048	0.0028	0.0012	180.0	-0.0048
011	0.457	0.901	03.21	-0.2259	-0.0568	0.0589	0.2298	0.441	0.0039	0.0022	-03.21	-10.02	0.0063	0.0025	0.0013	180.0	-0.0064
014	0.458	0.881	07.41	0.8244	-0.1077	0.1574	-0.8379	0.474	0.6793	0.0024	07.41	-10.01	-0.0047	0.0002	0.0006	180.0	0.0046
015	0.457	0.880	06.89	0.8178	-0.1184	0.1513	-0.8302	0.497	0.6687	0.0025	06.89	-10.01	-0.0006	0.0014	0.0000	179.9	0.0005
016	0.457	0.879	06.37	0.7880	-0.1247	0.1379	-0.7986	0.473	0.6210	0.0023	06.37	-10.01	-0.0011	0.0009	0.0007	179.9	0.0010
017	0.457	0.880	05.84	0.7542	-0.1311	0.1237	-0.7632	0.458	0.5687	0.0025	05.84	-10.01	0.0000	0.0012	0.0008	179.9	-0.0001
018	0.457	0.878	05.34	0.7225	-0.1371	0.1146	-0.7302	0.449	0.5220	0.0023	05.34	-10.01	0.0004	0.0013	0.0003	179.9	-0.0003
019	0.457	0.878	04.26	0.6601	-0.1461	0.0958	-0.6655	0.444	0.4356	0.0024	04.26	-10.01	0.0005	0.0017	-0.0002	179.9	-0.0005
020	0.457	0.879	03.21	0.5784	-0.1485	0.0804	-0.5821	0.456	0.3343	0.0023	03.21	-10.02	0.0028	0.0025	0.0008	180.0	-0.0039
021	0.458	0.882	04.09	0.3666	-0.1325	0.0568	-0.3677	0.474	0.1343	0.0024	01.09	-10.02	0.0041	0.0026	0.0009	180.0	-0.0042
022	0.458	0.880	-01.04	0.0997	-0.1001	0.0428	-0.0990	0.425	0.0098	0.0024	-01.04	-10.02	0.0053	0.0031	0.0012	180.0	-0.0054
023	0.458	0.880	-03.20	-0.2206	-0.0587	0.0511	0.2230	-0.368	0.485	0.0020	-03.20	-10.02	0.0072	0.0028	0.0011	180.0	-0.0073
026	0.460	0.860	07.39	0.8210	-0.1227	0.1560	-0.8143	0.468	0.6739	0.0023	07.39	-10.01	-0.0003	0.0014	0.0030	180.0	0.0002
027	0.503	0.861	06.88	0.8053	-0.1145	0.1427	-0.8169	0.428	0.6487	0.0023	06.88	-10.01	-0.0011	0.0015	0.0010	180.0	0.0010
028	0.460	0.863	06.37	0.7961	-0.1252	0.1387	-0.8064	0.441	0.6338	0.0025	06.37	-10.01	-0.0006	0.0009	-0.0007	179.9	0.0005
029	0.457	0.860	05.84	0.7704	-0.1315	0.1227	-0.7790	0.412	0.5935	0.0023	05.84	-10.01	0.0005	0.0015	-0.0005	179.9	-0.0007
030	0.457	0.860	05.32	0.7369	-0.1358	0.1105	-0.7441	0.395	0.5429	0.0022	05.32	-10.01	0.0003	0.0014	-0.0002	179.9	-0.0003
031	0.458	0.858	04.27	0.6739	-0.1402	0.0914	-0.6790	0.368	0.4020	0.0020	04.27	-10.01	0.0007	0.0017	-0.0004	179.9	-0.0007
032	0.460	0.864	03.23	0.5903	-0.1343	0.0761	-0.5937	0.405	0.3483	0.0023	03.23	-10.02	0.0037	0.0032	0.0007	180.0	-0.0037
033	0.460	0.861	01.12	0.3804	-0.1115	0.0524	-0.3814	0.428	0.1446	0.0021	01.12	-10.02	0.0041	0.0027	0.0010	180.0	-0.0042
034	0.458	0.858	-01.03	0.0988	-0.0828	0.0406	-0.0981	0.404	0.0097	0.0020	-01.03	-10.02	0.0048	0.0028	0.0010	180.0	-0.0049
035	0.460	0.861	-03.20	-0.2079	-0.0619	0.0460	0.2100	-0.324	0.431	0.0020	-03.20	-10.02	0.0067	0.0033	0.0011	180.0	-0.0058
036	0.457	0.840	07.37	0.8249	-0.1289	0.1529	-0.8377	0.425	0.6803	0.0023	07.37	-10.01	0.0002	0.0016	0.0025	180.0	-0.0003
037	0.460	0.839	06.86	0.8078	-0.1305	0.1405	-0.8189	0.409	0.6525	0.0020	06.86	-10.01	0.0011	0.0010	0.0006	180.0	0.0010
038	0.455	0.840	06.39	0.7982	-0.0910	0.1292	-0.8077	0.373	0.6370	0.0022	06.39	-10.01	0.0011	0.0012	0.0018	179.9	-0.0013
039	0.453	0.838	05.84	0.7881	-0.1338	0.1183	-0.7961	0.352	0.6210	0.0022	05.84	-10.01	0.0012	0.0018	0.0004	179.9	-0.0013
040	0.455	0.841	05.32	0.7537	-0.1327	0.1081	-0.7605	0.3354	0.5679	0.0022	05.32	-10.02	0.0019	0.0021	0.0000	180.0	-0.0020
043	0.457	0.840	04.29	0.6873	-0.1291	0.0874	-0.6921	0.3117	0.4723	0.0024	04.29	-10.01	0.0014	0.0017	0.0002	180.0	-0.0013

TABLE 3(G)  
FLARE PACK ON  
ETA = 0° (EXACT) = +3 1/2°

SER	REVN.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSR.	BASE.	RINC.	SLIP.	CROSS.	WVN H.	ROLL H.	PROM.	SIDE F
044	0.455	0.840	03. 25	0. 6045	-0. 1191	0. 0733	-0. 6078	-0. 0366	0. 3653	0. 0022	03. 25	-10. 02	0. 0026	0. 0023	0. 0008	180. 0 -0. 0027
045	0.455	0.841	04. 13	0. 3767	-0. 0945	0. 0495	-0. 3777	-0. 0390	0. 1417	0. 0021	01. 13	-10. 02	0. 0036	0. 0025	0. 0010	180. 0 -0. 0037
046	0.455	0.840	-04. 02	0. 0952	-0. 0756	0. 0378	-0. 0946	-0. 0376	0. 0090	0. 0019	-01. 02	-10. 02	0. 0040	0. 0028	0. 0009	180. 0 -0. 0041
047	0.457	0.840	-03. 19	-0. 1945	-0. 0638	0. 0428	0. 1965	-0. 0301	0. 0377	0. 0019	-02. 19	-10. 02	0. 0062	0. 0027	0. 0013	180. 0 -0. 0042
050	0.453	0.799	07. 36	0. 8351	-0. 1203	0. 1486	-0. 8474	-0. 0382	0. 6973	0. 0021	07. 36	-10. 01	0. 0012	0. 0014	0. 0024	180. 0 -0. 0013
051	0.455	0.801	06. 86	0. 6229	-0. 1279	0. 1373	-0. 8335	-0. 0258	0. 6770	0. 0022	06. 86	-10. 01	0. 0014	0. 0018	0. 0022	180. 0 -0. 0015
053	0.455	0.800	06. 25	0. 8110	-0. 1226	0. 1255	-0. 8200	-0. 0330	0. 6576	0. 0020	06. 35	-10. 01	0. 0013	0. 0018	0. 0020	180. 0 -0. 0014
053	0.455	0.800	06. 35	0. 8110	-0. 1226	0. 1255	-0. 8200	-0. 0330	0. 6576	0. 0020	06. 35	-10. 01	0. 0013	0. 0018	0. 0020	180. 0 -0. 0014
054	0.455	0.801	05. 86	0. 7988	-0. 1167	0. 1146	-0. 8064	-0. 0303	0. 6380	0. 0022	05. 86	-10. 01	0. 0008	0. 0016	0. 0016	180. 0 -0. 0016
055	0.455	0.800	05. 34	0. 7848	-0. 1221	0. 1050	-0. 7913	-0. 0294	0. 6159	0. 0020	05. 34	-10. 01	0. 0014	0. 0019	0. 0004	180. 0 -0. 0015
056	0.453	0.799	04. 32	0. 7135	-0. 1017	0. 0815	-0. 7139	-0. 0274	0. 5119	0. 0020	04. 32	-10. 01	0. 0014	0. 0020	0. 0008	180. 0 -0. 0013
057	0.455	0.800	03. 27	0. 6112	-0. 0870	0. 0672	-0. 6142	-0. 0300	0. 3734	0. 0021	02. 27	-10. 02	0. 0029	0. 0024	0. 0015	180. 0 -0. 0020
058	0.455	0.801	04. 13	0. 3497	-0. 0756	0. 0470	-0. 3807	-0. 0381	0. 1222	0. 0020	01. 13	-10. 02	0. 0043	0. 0034	0. 0011	180. 0 -0. 0014
059	0.455	0.799	-04. 02	0. 0865	-0. 0668	0. 0403	-0. 0859	-0. 0400	0. 0074	0. 0018	-01. 02	-10. 02	0. 0046	0. 0026	0. 0010	180. 0 -0. 0047
060	0.455	0.800	-03. 16	-0. 1742	-0. 0591	0. 0423	0. 1762	-0. 0109	0. 0102	0. 0018	-03. 16	-10. 02	0. 0058	0. 0027	0. 0013	180. 0 -0. 0039
063	0.465	0.749	07. 35	0. 8162	-0. 1265	0. 1276	-0. 8471	-0. 0273	0. 6991	0. 0024	07. 35	-10. 01	0. 0012	0. 0022	0. 0026	180. 0 -0. 0013
064	0.465	0.749	06. 85	0. 8214	-0. 1196	0. 1239	-0. 8307	-0. 0248	0. 6746	0. 0022	06. 85	-10. 01	0. 0015	0. 0023	0. 0028	180. 0 -0. 0016
065	0.465	0.751	06. 35	0. 8160	-0. 1203	0. 1146	-0. 8238	-0. 0215	0. 6658	0. 0028	06. 35	-10. 01	0. 0036	0. 0022	0. 0042	180. 0 -0. 0016
066	0.465	0.752	05. 85	0. 7975	-0. 0978	0. 1034	-0. 8000	-0. 0196	0. 6296	0. 0022	05. 85	-10. 01	0. 0024	0. 0019	0. 0037	180. 0 -0. 0012
067	0.463	0.749	05. 25	0. 7749	-0. 0807	0. 0921	-0. 7802	-0. 0172	0. 6003	0. 0022	05. 25	-10. 01	0. 0023	0. 0021	0. 0025	180. 0 -0. 0024
068	0.463	0.748	04. 34	0. 6938	-0. 0757	0. 0715	-0. 6973	-0. 0169	0. 4812	0. 0021	04. 34	-10. 01	0. 0014	0. 0047	0. 0016	180. 0 -0. 0015
069	0.465	0.751	03. 25	0. 5707	-0. 0730	0. 0566	-0. 5731	-0. 0220	0. 3255	0. 0021	03. 25	-10. 01	0. 0042	0. 0024	0. 0019	180. 0 -0. 0013
070	0.465	0.751	01. 14	0. 3222	-0. 0698	0. 0424	-0. 3830	-0. 0340	0. 1037	0. 0021	01. 14	-10. 02	0. 0046	0. 0028	0. 0012	180. 0 -0. 0014
071	0.465	0.751	-01. 01	0. 0821	-0. 0610	0. 0367	-0. 0815	-0. 0362	0. 0066	0. 0020	-01. 01	-10. 02	0. 0048	0. 0026	0. 0012	180. 0 -0. 0013
072	0.465	0.750	-01. 14	-0. 1593	-0. 0504	0. 0396	0. 1612	-0. 0290	0. 0253	0. 0016	-01. 14	-10. 02	0. 0059	0. 0028	0. 0015	180. 0 -0. 0060
073	0.452	0.700	07. 33	0. 9187	-0. 1219	0. 1342	-0. 8491	-0. 0239	0. 7033	0. 0021	07. 33	-10. 01	0. 0020	0. 0023	0. 0035	180. 0 -0. 0024
076	0.452	0.701	06. 83	0. 8306	-0. 1213	0. 1215	-0. 8392	-0. 0196	0. 6897	0. 0022	06. 83	-10. 01	0. 0026	0. 0024	0. 0041	180. 0 -0. 0027
077	0.449	0.699	06. 34	0. 8010	-0. 0835	0. 1087	-0. 8082	-0. 0174	0. 6415	0. 0028	06. 34	-10. 01	0. 0038	0. 0025	0. 0044	180. 0 -0. 0029
078	0.452	0.701	05. 82	0. 7841	-0. 0952	0. 0969	-0. 7893	-0. 0146	0. 6147	0. 0022	05. 82	-10. 01	0. 0031	0. 0024	0. 0037	180. 0 -0. 0022
079	0.452	0.701	05. 24	0. 7459	-0. 0849	0. 0848	-0. 7506	-0. 0132	0. 5863	0. 0022	05. 24	-10. 01	0. 0027	0. 0026	0. 0027	180. 0 -0. 0028
080	0.452	0.701	04. 27	0. 6506	-0. 0744	0. 0670	-0. 6539	-0. 0162	0. 4234	0. 0024	04. 27	-10. 01	0. 0024	0. 0024	0. 0022	180. 0 -0. 0025
081	0.452	0.699	03. 24	0. 6370	-0. 0720	0. 0547	-0. 5392	-0. 0226	0. 2862	0. 0019	01. 24	-10. 01	0. 0026	0. 0025	0. 0021	180. 0 -0. 0024
082	0.452	0.701	01. 10	0. 3080	-0. 0672	0. 0408	-0. 3086	-0. 0329	0. 9553	0. 0019	01. 10	-10. 01	0. 0041	0. 0041	0. 0020	180. 0 -0. 0042
083	0.452	0.699	-01. 01	0. 0790	-0. 0576	0. 0363	-0. 0784	-0. 0159	0. 0662	0. 0019	-01. 01	-10. 01	0. 0043	0. 0043	0. 0026	180. 0 -0. 0044
084	0.452	0.699	-01. 12	-0. 1494	-0. 0458	0. 0392	0. 1512	-0. 0293	0. 0222	0. 0017	-01. 12	-10. 01	0. 0055	0. 0027	0. 0013	180. 0 -0. 0055
087	0.447	0.500	07. 24	0. 8077	-0. 1003	0. 1224	-0. 8168	-0. 0173	0. 6524	0. 0023	07. 24	-10. 01	0. 0011	0. 0028	0. 0022	180. 0 -0. 0012

TABLE 3(0)  
FLARE PACK ON  
ETA = 0° EXTRACT = +3 1/2"

SER	REVN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	BASE.	RING.	SLIP.	CROSS.	VAN N.	ROLL N	RANG.	SIDE F
088	0	447	0	.500	.06.74	0.7643	-0.0966	0.1062	-0.7921	-0.0110	0.6160	0.0023	06.74	-00.04	0.0017	0.0019	180.0 -0.0018
089	0	449	0	.501	.06.23	0.7470	-0.0860	0.0928	-0.7529	-0.0088	0.5579	0.0023	06.23	-00.01	0.0021	0.0018	180.0 -0.0022
090	0	444	0	.490	.05.71	0.7048	-0.0812	0.0817	-0.7095	-0.0090	0.4966	0.0024	05.71	-00.01	0.0017	0.0022	0.0018 -0.0018
091	0	447	0	.499	.05.20	0.6601	-0.0779	0.0724	-0.6643	-0.0099	0.4356	0.0023	05.20	-00.01	0.0013	0.0027	0.0019 180.0 -0.0020
092	0	447	0	.499	.04.17	0.5667	-0.0734	0.0587	-0.5696	-0.0151	0.3210	0.0022	04.17	-00.01	0.0032	0.0029	0.0023 180.0 -0.0031
093	0	447	0	.199	.03.13	0.4715	-0.0693	0.0491	-0.4736	-0.0210	0.2221	0.0022	03.13	-00.01	0.0024	0.0024	0.0020 180.0 -0.0020
094	0	449	0	.501	.01.06	0.2767	-0.0503	0.0382	-0.2775	-0.0209	0.0765	0.0025	01.06	-00.01	0.0036	0.0025	0.0013 180.0 -0.0045
095	0	450	0	.500	-.01.04	0.0731	-0.0497	0.0340	-0.0726	-0.0333	0.0052	0.0024	01.04	-00.01	0.0044	0.0025	0.0013 180.0 -0.0037
096	0	450	0	.500	-.01.06	-0.1290	-0.0374	0.0362	0.1306	-0.0274	0.0165	0.0019	-01.08	-00.01	0.0056	0.0026	0.0013 180.0 -0.0037

TABLE 3(H)  
FLARE PACK ON  
ETA = -15° ETAKT = +1/2°

SER	REYN.	MACH.	INCID.	LIFT.	PITCH.	DRAO	NORMAL	AXIAL.	CL SQ.	BASE.	ANG.	SLIP.	CROSS.	VAN H.	ROLL H.	RAND.	SIDE F
002	0.455	0.900	06.53	0.7227	0.0752	0.1483	-0.7349	-0.0625	0.5222	0.0016	06.53	-0.0004	-0.0017	179.9	0.0371		
003	0.455	0.900	04.46	0.5662	0.1043	0.1097	-0.5732	-0.0640	0.3204	0.0014	04.46	-0.0002	0.0002	180.0	0.0293		
004	0.455	0.900	02.36	0.3703	0.1346	0.0809	-0.3740	-0.0644	0.1374	0.0011	02.36	-0.0003	0.0011	0.0019	180.0	0.0178	
005	0.455	0.901	00.25	0.1234	0.1765	0.0662	-0.1238	-0.0648	0.0151	0.0009	00.25	-0.0006	0.0016	0.0017	180.0	0.0065	
006	0.449	0.880	06.52	0.7168	0.0873	0.1422	-0.7284	-0.0583	0.5137	0.0015	06.52	-0.0004	-0.0016	179.9	0.0370		
007	0.449	0.878	04.46	0.5641	0.1228	0.1022	-0.5704	-0.0568	0.3180	0.0012	04.46	-0.0003	0.0003	180.0	0.0289		
008	0.450	0.879	02.38	0.3829	0.1525	0.0751	-0.3868	-0.0582	0.1472	0.0009	02.38	-0.0002	0.0019	180.0	0.0183		
009	0.449	0.881	00.27	0.1478	0.1875	0.0622	-0.1482	-0.0609	0.0218	0.0017	00.27	-0.0002	0.0020	180.0	0.0072		
013	0.457	0.859	06.53	0.7142	0.0977	0.1344	-0.7250	-0.0509	0.5101	0.0013	06.53	-0.0002	-0.0004	179.9	0.0372		
014	0.455	0.858	04.48	0.5759	0.1360	0.0985	-0.6819	-0.0519	0.3314	0.0012	04.48	-0.0003	0.0005	180.0	0.0302		
015	0.457	0.860	02.41	0.3979	0.1679	0.0735	-0.4007	-0.0559	0.1582	0.0009	02.41	-0.0003	0.0020	180.0	0.0184		
016	0.457	0.861	00.28	0.1477	0.2008	0.0580	-0.1481	-0.0564	0.0217	0.0008	00.28	-0.0002	0.0015	180.0	0.0071		
017	0.449	0.838	06.52	0.7116	0.1029	0.1279	-0.7216	-0.0449	0.5064	0.0012	06.52	-0.0003	0.0016	180.0	0.0362		
018	0.450	0.841	04.49	0.5851	0.1469	0.0954	-0.5909	-0.0481	0.3423	0.0011	04.49	-0.0003	0.0006	180.0	0.0296		
019	0.450	0.839	02.42	0.4034	0.1857	0.0689	-0.4060	-0.0510	0.1625	0.0009	02.42	-0.0003	0.0019	180.0	0.0198		
020	0.450	0.839	00.28	0.1352	0.2114	0.0545	-0.1355	-0.0523	0.0182	0.0006	00.28	-0.0002	0.0021	0.0019	180.0	0.0069	
023	0.463	0.800	06.54	0.7140	0.1198	0.1246	-0.7236	-0.0412	0.5097	0.0010	06.54	-0.0003	0.0036	180.0	0.0156		
024	0.455	0.798	04.54	0.6119	0.1829	0.0896	-0.6172	-0.0400	0.3744	0.0009	04.54	-0.0003	0.0019	180.0	0.0108		
025	0.467	0.800	02.42	0.3815	0.2079	0.0614	-0.3829	-0.0445	0.2454	0.0007	02.42	-0.0003	0.0016	180.0	0.0195		
026	0.467	0.800	00.27	0.1454	0.2172	0.0523	-0.1158	-0.0512	0.0132	0.0006	00.27	-0.0002	0.0056	180.0	0.0055		
027	0.445	0.750	06.52	0.7321	0.1501	0.1183	-0.7210	-0.0357	0.5070	0.0010	06.52	-0.0003	0.0029	180.0	0.0329		
028	0.445	0.751	04.50	0.3878	0.2112	0.0794	-0.3924	-0.0320	0.3454	0.0010	04.50	-0.0003	0.0027	180.0	0.0286		
029	0.445	0.751	02.37	0.3408	0.2144	0.0570	-0.3430	-0.0422	0.1161	0.0007	02.37	-0.0003	0.0018	180.0	0.0171		
030	0.445	0.750	00.25	0.0594	0.2229	0.0501	-0.0597	-0.0491	0.0098	0.0006	00.25	-0.0002	0.0047	0.0019	180.0	0.0046	
031	0.426	0.699	06.48	0.6951	0.1738	0.1169	-0.7033	-0.0306	0.4831	0.0014	06.48	-0.0003	0.0022	0.0019	180.0	0.0322	
032	0.426	0.702	04.44	0.5281	0.2182	0.0719	-0.5422	-0.0289	0.2894	0.0016	04.44	-0.0003	0.0027	180.0	0.0257		
033	0.422	0.700	02.32	0.3110	0.2204	0.0549	-0.3131	-0.0416	0.0966	0.0007	02.32	-0.0002	0.0049	180.0	0.0148		
034	0.422	0.700	00.22	0.0823	0.2288	0.0495	-0.0826	-0.0487	0.0067	0.0007	00.22	-0.0002	0.0042	0.0022	180.0	0.0041	
035	0.440	0.501	06.35	0.6704	0.1896	0.0939	-0.6367	-0.0222	0.3969	0.0014	06.35	-0.0003	0.0027	180.0	0.0207		
036	0.440	0.501	04.29	0.4468	0.2077	0.0629	-0.4904	-0.0261	0.1995	0.0012	04.29	-0.0002	0.0024	0.0029	180.0	0.0214	
039	0.427	0.429	02.22	0.2866	0.2167	0.0585	-0.2605	-0.0284	0.0668	0.0018	02.22	-0.0002	0.0120	0.0023	180.0	0.0119	
040	0.440	0.801	00.16	0.6878	0.2878	0.0471	-0.6881	-0.0460	0.0033	0.0009	00.16	-0.0001	0.0020	0.0023	180.0	0.0023	

TABLE 3(1)  
FLARE PACK ON  
ETA = -10° EXTRACT = +1/2"

SER	REVN.	WACH.	INCID.	LIFT.	PITCH.	DRAE	NORML.	AXIAL.	CLSR.	BASE.	MINC.	SLIP.	CROSS.	VAN H.	ROLL H.	RANK.	SIDE F
0012	0.457	0.307	06.47	0.7528	-0.0023	0.1450	-0.7645	-0.0573	0.5667	0.0019	06.47	(10.00	-0.0053	-0.0015	-0.0012	179.3	0.0031
0113	0.455	0.898	04.41	0.5892	0.0406	0.1028	-0.5955	-0.0556	0.3471	0.0016	04.41	(10.00	-0.0033	-0.0042	0.0003	180.0	0.0034
0014	0.457	0.901	02.31	0.3914	0.0643	0.0757	-0.3943	-0.0584	0.1531	0.0015	02.31	(10.00	-0.0016	-0.0006	0.0016	180.0	0.0015
0015	0.457	0.500	00.19	0.1447	0.0991	0.0596	-0.1450	-0.0579	0.0208	0.0013	00.19	(10.00	-0.0010	-0.0004	0.0015	180.0	0.0009
0116	0.456	0.880	06.47	0.7468	0.0084	0.1284	-0.7578	-0.0516	0.5577	0.0017	06.47	(10.00	-0.0050	-0.0014	-0.0011	179.3	0.0049
0017	0.450	0.880	04.41	0.5898	0.0546	0.0993	-0.5958	-0.0518	0.3478	0.0017	04.41	(10.00	-0.0048	-0.0010	0.0001	180.0	0.0047
0118	0.453	0.880	02.31	0.4040	0.0816	0.0712	-0.4067	-0.0533	0.1631	0.0014	02.31	(10.00	-0.0007	-0.0004	0.0016	180.0	0.0006
0119	0.450	0.879	00.21	0.1702	0.1133	0.0541	-0.1705	-0.0524	0.0289	0.0011	00.21	(10.00	-0.0006	-0.0003	0.0016	180.0	0.0008
0120	0.460	0.861	06.48	0.7503	0.0182	0.1337	-0.7607	-0.0465	0.3629	0.0017	06.48	(10.00	-0.0047	-0.0013	-0.0003	179.3	0.0046
0113	0.457	0.859	04.43	0.5997	0.0667	0.0927	-0.6092	-0.0445	0.3595	0.0016	04.43	(10.00	-0.0043	-0.0014	0.0005	180.0	0.0042
0114	0.460	0.861	02.35	0.4194	0.0964	0.0664	-0.4219	-0.0478	0.1757	0.0014	02.35	(10.00	-0.0013	-0.0006	0.0018	180.0	0.0012
0115	0.457	0.860	00.23	0.1744	0.1279	0.0498	-0.1747	-0.0480	0.0304	0.0014	00.23	(10.00	-0.0006	-0.0004	0.0017	180.0	0.0008
0116	0.450	0.840	06.49	0.7350	0.0503	0.1262	-0.7447	-0.0408	0.3402	0.0015	06.49	(10.00	-0.0047	-0.0010	0.0007	180.0	0.0046
0117	0.450	0.839	04.44	0.6125	0.0779	0.0908	-0.6178	-0.0417	0.3750	0.0014	04.44	(10.00	-0.0043	-0.0012	0.0006	180.0	0.0042
0118	0.450	0.839	02.36	0.4293	0.1137	0.0628	-0.4316	-0.0428	0.1841	0.0012	02.36	(10.00	-0.0017	-0.0002	0.0020	180.0	0.0016
0119	0.453	0.840	00.23	0.1609	0.1390	0.0476	-0.1612	-0.0460	0.0258	0.0010	00.23	(10.00	-0.0003	-0.0001	0.0017	180.0	0.0002
0122	0.467	0.798	06.48	0.7520	0.0346	0.1215	-0.7610	-0.0343	0.3694	0.0015	06.48	(10.00	-0.0028	-0.0010	0.0019	180.0	0.0037
0123	0.467	0.800	04.46	0.6407	0.1050	0.0831	-0.6493	-0.0313	0.4104	0.0015	04.46	(10.00	-0.0034	-0.0009	0.0012	180.0	0.0033
0124	0.467	0.800	02.37	0.4084	0.1345	0.0547	-0.4105	-0.0266	0.1667	0.0012	02.37	(10.00	-0.0045	-0.0002	0.0021	180.0	0.0014
0125	0.467	0.800	00.22	0.1423	0.1431	0.0451	-0.1428	-0.0266	0.0202	0.0010	00.22	(10.00	-0.0008	-0.0004	0.0017	180.0	0.0007
0126	0.449	0.749	06.46	0.7495	0.0627	0.1137	-0.7576	-0.0272	0.3617	0.0014	06.46	(10.00	-0.0019	-0.0006	0.0035	180.0	0.0018
0127	0.450	0.753	04.45	0.6227	0.1287	0.0751	-0.6268	-0.0231	0.3877	0.0014	04.45	(10.00	-0.0031	-0.0007	0.0020	180.0	0.0030
0128	0.449	0.750	02.32	0.3716	0.1249	0.0510	-0.3735	-0.0348	0.1380	0.0012	02.32	(10.01	-0.0017	-0.0001	0.0020	180.0	0.0016
0129	0.449	0.749	00.19	0.1272	0.1421	0.0440	-0.1275	-0.0427	0.0161	0.0010	00.19	(10.00	-0.0001	-0.0002	0.0015	180.0	0.0000
0130	0.426	0.700	06.42	0.7351	0.0807	0.1072	-0.7426	-0.0228	0.3402	0.0015	06.42	(10.00	-0.0016	-0.0004	0.0043	180.0	0.0015
0131	0.426	0.700	04.38	0.5745	0.1264	0.0671	-0.5780	-0.0217	0.3298	0.0014	04.38	(10.00	-0.0025	-0.0003	0.0029	180.0	0.0024
0132	0.426	0.700	02.27	0.3452	0.1304	0.0494	-0.3470	-0.0245	0.1191	0.0014	02.27	(10.00	-0.0013	-0.0003	0.0019	180.0	0.0012
0133	0.426	0.700	00.16	0.1177	0.1401	0.0421	-0.1179	-0.0408	0.0138	0.0010	00.16	(10.01	-0.0002	0.0001	0.0016	180.0	0.0001
0136	0.440	0.501	06.21	0.6710	0.0967	0.0906	-0.6770	-0.0147	0.4501	0.0016	06.21	(10.01	-0.0030	0.0000	0.0023	180.0	0.0020
0137	0.440	0.501	04.25	0.4886	0.1125	0.0588	-0.4917	-0.0208	0.2386	0.0017	04.25	(10.01	-0.0018	0.0002	0.0026	180.0	0.0017
0138	0.440	0.499	02.18	0.2979	0.1213	0.0450	-0.2956	-0.0223	0.0886	0.0014	02.18	(10.00	-0.0012	-0.0004	0.0021	180.0	0.0014
0139	0.444	0.501	00.14	0.0958	0.1325	0.0402	-0.0957	-0.0287	0.0090	0.0013	00.14	(10.00	-0.0001	-0.0001	0.0010	180.0	-0.0002

TABLE 3(J)  
FLARE PACK OK  
ETA = -5°ETR(T) = +1/2°

SER	REYN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	BASE.	RING.	SLIP.	CROSS.	YAW H.	ROLL H.	RANG.	SIDE F
002	0.449	0.900	06. 44	0. 7750	-0. 0676	0. 1420	-0. 7962	-0. 0534	0. 6006	0. 0024	06. 41	(0. 00	-0. 0051	-0. 0014	-0. 0013	179. 9	0. 0050
003	0.447	0.901	04. 34	0. 6160	-0. 0375	0. 1011	-0. 6220	-0. 0520	0. 3794	0. 0021	04. 34	(0. 00	-0. 0037	-0. 0010	0. 0004	180. 0	0. 0036
004	0.447	0.900	02. 24	0. 4159	-0. 0126	0. 0705	-0. 4184	-0. 0522	0. 3728	0. 0019	02. 24	(0. 00	-0. 0013	-0. 0005	0. 0016	180. 0	0. 0012
005	0.449	0.902	00. 12	0. 1657	0. 0169	0. 0534	-0. 1659	-0. 0513	0. 0273	0. 0018	00. 12	(0. 00	-0. 0004	-0. 0002	0. 0015	180. 0	0. 0003
006	0.442	0.881	06. 42	0. 7695	-0. 0558	0. 1375	-0. 7802	-0. 0486	0. 5921	0. 0020	06. 42	(0. 00	-0. 0040	-0. 0014	-0. 0007	179. 9	0. 0029
007	0.442	0.881	04. 35	0. 6128	-0. 0192	0. 0957	-0. 6184	-0. 0469	0. 3754	0. 0021	04. 35	(0. 00	-0. 0038	-0. 0010	0. 0003	180. 0	0. 0037
008	0.442	0.880	02. 26	0. 4282	0. 0042	0. 0670	-0. 4206	-0. 0483	0. 1832	0. 0018	02. 26	(0. 00	-0. 0010	-0. 0002	0. 0017	180. 0	0. 0009
009	0.442	0.880	00. 14	0. 1928	0. 0311	0. 0492	-0. 1940	-0. 0472	0. 0375	0. 0015	00. 14	(0. 00	-0. 0002	-0. 0001	0. 0017	180. 0	0. 0004
010	0.450	0.858	06. 43	0. 7724	-0. 0479	0. 1322	-0. 7825	-0. 0430	0. 5965	0. 0019	06. 43	(0. 00	-0. 0045	-0. 0012	-0. 0005	179. 9	0. 0044
011	0.449	0.860	04. 37	0. 6234	-0. 0062	0. 0904	-0. 6286	-0. 0407	0. 3886	0. 0019	04. 37	(0. 00	-0. 0035	-0. 0009	0. 0006	180. 0	0. 0034
012	0.449	0.860	02. 29	0. 4467	0. 0199	0. 0622	-0. 4490	-0. 0425	0. 5994	0. 0018	02. 29	(-0. 01	-0. 0014	0. 0000	0. 0018	180. 0	0. 0013
013	0.449	0.860	00. 16	0. 1940	0. 0491	0. 0469	-0. 1943	-0. 0450	0. 0375	0. 0014	00. 16	(-0. 01	-0. 0004	0. 0000	0. 0017	180. 0	0. 0003
014	0.452	0.859	06. 44	0. 7637	-0. 0148	0. 1275	-0. 7723	-0. 0191	0. 5831	0. 0018	06. 44	(0. 00	-0. 0037	-0. 0012	0. 0003	180. 0	0. 0036
015	0.445	0.841	04. 38	0. 6358	0. 0048	0. 0879	-0. 6418	-0. 0370	0. 4054	0. 0019	04. 38	(0. 00	-0. 0041	-0. 0011	0. 0007	180. 0	0. 0040
016	0.445	0.840	02. 30	0. 4524	0. 0358	0. 0581	-0. 4545	-0. 0383	0. 2045	0. 0017	02. 30	(-0. 01	-0. 0019	-0. 0002	0. 0020	180. 0	0. 0016
017	0.445	0.840	00. 16	0. 1846	0. 0589	0. 0420	-0. 1848	-0. 0410	0. 0339	0. 0015	00. 16	(-0. 01	-0. 0011	-0. 0001	0. 0016	180. 0	0. 0010
018	0.445	0.840	06. 44	0. 7710	-0. 0253	0. 1204	-0. 7798	-0. 0315	0. 3944	0. 0017	06. 44	(0. 00	-0. 0045	-0. 0010	0. 0020	180. 0	0. 0044
019	0.452	0.802	04. 41	0. 6629	0. 0306	0. 0800	-0. 6692	-0. 0266	0. 4027	0. 0019	04. 41	(0. 00	-0. 0029	-0. 0008	0. 0015	180. 0	0. 0028
020	0.452	0.802	02. 31	0. 4322	0. 0568	0. 0514	-0. 4340	-0. 0324	0. 1866	0. 0016	02. 31	(0. 00	-0. 0007	-0. 0001	0. 0021	180. 0	0. 0006
021	0.452	0.791	00. 16	0. 1658	0. 0646	0. 0416	-0. 1660	-0. 0399	0. 0273	0. 0012	00. 16	(-0. 01	-0. 0015	-0. 0002	0. 0017	180. 0	0. 0014
022	0.452	0.802	00. 16	0. 1940	0. 0491	0. 0469	-0. 1943	-0. 0450	0. 0375	0. 0014	00. 16	(-0. 01	-0. 0004	0. 0000	0. 0016	180. 0	0. 0003
023	0.452	0.802	02. 26	0. 4281	0. 0048	0. 0680	-0. 4206	-0. 0483	0. 1832	0. 0018	02. 26	(0. 00	-0. 0006	-0. 0001	0. 0020	180. 0	0. 0007
024	0.452	0.802	00. 14	0. 1928	0. 0311	0. 0492	-0. 1940	-0. 0450	0. 0375	0. 0015	00. 14	(-0. 01	-0. 0013	-0. 0002	0. 0017	180. 0	0. 0012
025	0.452	0.802	00. 16	0. 1658	0. 0646	0. 0416	-0. 1660	-0. 0399	0. 0273	0. 0012	00. 16	(-0. 01	-0. 0015	-0. 0002	0. 0017	180. 0	0. 0014
026	0.440	0.749	06. 42	0. 7697	0. 0000	0. 1128	-0. 7776	-0. 0242	0. 3923	0. 0018	06. 42	(0. 00	-0. 0021	-0. 0005	0. 0041	180. 0	0. 0020
027	0.442	0.750	04. 40	0. 6419	0. 0554	0. 0704	-0. 6456	-0. 0192	0. 4120	0. 0018	04. 40	(0. 00	-0. 0028	-0. 0006	0. 0021	180. 0	0. 0027
028	0.442	0.751	02. 26	0. 3851	0. 0587	0. 0480	-0. 3958	-0. 0208	0. 1960	0. 0016	02. 26	(0. 00	-0. 0008	-0. 0005	0. 0020	180. 0	0. 0007
029	0.442	0.750	00. 14	0. 1516	0. 0651	0. 0394	-0. 1516	-0. 0277	0. 0229	0. 0014	00. 14	(-0. 01	-0. 0013	-0. 0002	0. 0017	180. 0	0. 0012
030	0.449	0.700	06. 39	0. 7606	0. 0148	0. 1052	-0. 7677	-0. 0183	0. 3784	0. 0026	06. 39	(0. 00	-0. 0010	0. 0000	0. 0045	180. 0	0. 0009
031	0.449	0.701	04. 34	0. 6007	0. 0544	0. 0659	-0. 6040	-0. 0185	0. 3606	0. 0019	04. 34	(0. 00	-0. 0015	0. 0000	0. 0028	180. 0	0. 0014
032	0.449	0.701	02. 22	0. 3716	0. 0569	0. 0459	-0. 3732	-0. 0239	0. 1379	0. 0016	02. 22	(0. 00	-0. 0008	-0. 0001	0. 0019	180. 0	0. 0007
033	0.449	0.699	00. 12	0. 1419	0. 0650	0. 0383	-0. 1421	-0. 0269	0. 0203	0. 0013	00. 12	(0. 00	-0. 0002	-0. 0001	0. 0016	180. 0	0. 0004
034	0.449	0.699	06. 29	0. 6951	0. 0329	0. 0900	-0. 7019	-0. 0110	0. 4844	0. 0028	06. 28	(0. 01	-0. 0025	0. 0001	0. 0023	180. 0	0. 0024
035	0.449	0.699	04. 22	0. 5172	0. 0462	0. 0574	-0. 5201	-0. 0372	0. 2673	0. 0020	04. 22	(-0. 01	-0. 0014	0. 0002	0. 0028	180. 0	0. 0013
036	0.437	0.501	02. 15	0. 2223	0. 0529	0. 0427	-0. 2248	-0. 0287	0. 1044	0. 0018	02. 15	(-0. 01	-0. 0008	0. 0000	0. 0021	180. 0	0. 0007
037	0.437	0.501	00. 08	0. 2225	0. 0524	0. 0369	-0. 1937	-0. 0281	0. 0191	0. 0018	00. 08	(0. 00	-0. 0004	-0. 0003	0. 0018	180. 0	0. 0003

TABLE 3(K)  
FLARE PACK ON  
ETA = 0°(TRACT) = +1/2°

SER	REYN.	WIND.	INCID.	LIFT.	PITCH.	DRAE.	NORM.	AXIAL.	C150.	BASE.	RINC.	SLIP.	CROSS.	VAN N.	ROLL N.	PANG.	SIDE F.
0012	0.439	0.900	07.41	0.8590	-0.1195	0.1692	-0.0738	-0.0546	0.7379	0.0024	07.41	10.00	-0.0046	-0.0010	-0.0002	179.9	0.0045
0113	0.437	0.897	06.89	0.8254	-0.1206	0.1541	-0.0380	-0.0519	0.6811	0.0024	06.89	10.00	-0.0040	-0.0010	-0.0006	179.9	0.0039
0014	0.435	0.898	06.37	0.7898	-0.1211	0.1414	-0.0607	-0.0507	0.6237	0.0028	06.37	10.00	-0.0030	-0.0011	-0.0009	179.9	0.0039
0115	0.435	0.900	05.84	0.7340	-0.1215	0.1299	-0.0764	-0.0501	0.5684	0.0023	05.84	10.00	-0.0017	-0.0016	-0.0005	179.9	0.0046
0116	0.439	0.901	05.33	0.7158	-0.1151	0.1206	-0.0724	-0.0512	0.5123	0.0024	05.33	10.00	-0.0041	-0.0014	-0.0008	179.9	0.0040
0017	0.439	0.902	04.29	0.6265	-0.1060	0.0998	-0.6423	-0.0497	0.4050	0.0022	04.29	10.00	-0.0038	-0.0012	-0.0003	180.0	0.0037
0118	0.435	0.900	03.24	0.5444	-0.0940	0.0832	-0.5483	-0.0502	0.2962	0.0021	03.24	10.00	-0.0029	-0.0009	-0.0011	180.0	0.0028
0019	0.439	0.901	04.12	0.3197	-0.0741	0.0607	-0.3209	-0.0525	0.1020	0.0020	01.12	10.00	-0.0019	-0.0004	-0.0014	180.0	0.0018
0110	0.435	0.899	-01.04	0.0540	-0.0361	0.0482	-0.0532	-0.0474	0.0828	0.0018	-01.04	10.00	-0.0005	-0.0005	-0.0017	180.0	0.0005
0111	0.439	0.902	-03.15	-0.2523	0.0161	0.0585	0.2950	-0.0430	0.1635	0.0017	-03.15	10.00	-0.0021	-0.0002	-0.0020	180.0	-0.0022
0114	0.445	0.879	07.44	0.8319	-0.0735	0.1577	-0.8454	-0.0466	0.6920	0.0020	07.44	10.00	-0.0038	-0.0014	-0.0013	180.0	0.0037
0115	0.445	0.881	06.90	0.8206	-0.1112	0.1490	-0.8327	-0.0472	0.6732	0.0021	06.90	10.00	-0.0047	-0.0011	-0.0005	179.9	0.0046
0116	0.445	0.881	06.38	0.7894	-0.1126	0.1362	-0.7995	-0.0454	0.6226	0.0023	06.38	10.00	-0.0051	-0.0015	-0.0007	179.9	0.0050
0117	0.442	0.880	05.86	0.7522	-0.1065	0.1240	-0.7611	-0.0444	0.5658	0.0024	05.86	10.00	-0.0027	-0.0012	-0.0004	179.9	0.0026
0118	0.442	0.880	05.34	0.7137	-0.1010	0.1131	-0.7213	-0.0441	0.5094	0.0024	05.34	10.00	-0.0044	-0.0014	-0.0002	179.9	0.0042
0119	0.445	0.881	04.34	0.6379	-0.0867	0.0947	-0.6433	-0.0443	0.4068	0.0024	04.34	10.00	-0.0028	-0.0014	-0.0013	180.0	0.0037
0220	0.442	0.879	03.26	0.5502	-0.0739	0.0776	-0.5539	-0.0443	0.3026	0.0019	03.26	10.00	-0.0023	-0.0009	-0.0012	180.0	0.0022
0121	0.442	0.879	01.15	0.3440	-0.0550	0.0544	-0.3451	-0.0451	0.1182	0.0018	01.15	10.00	-0.0013	-0.0004	-0.0016	180.0	0.0018
0222	0.445	0.880	-00.99	0.0724	-0.0266	0.0427	-0.0717	-0.0424	0.0551	0.0016	00.99	10.00	-0.0006	-0.0003	-0.0018	180.0	0.0005
0223	0.445	0.880	-03.15	-0.2462	0.0120	0.0506	0.2485	-0.0356	0.0605	0.0015	-03.15	10.00	-0.0028	-0.0002	-0.0017	180.0	-0.0029
0226	0.444	0.860	07.39	0.8144	-0.1078	0.1529	-0.8271	-0.0448	0.6627	0.0020	07.39	10.00	-0.0043	-0.0016	-0.0043	180.0	0.0042
0227	0.447	0.861	06.92	0.8079	-0.0773	0.1430	-0.8194	-0.0424	0.6527	0.0022	06.92	10.00	-0.0052	-0.0014	-0.0009	180.0	0.0051
0228	0.444	0.860	06.29	0.7931	-0.1021	0.1334	-0.8032	-0.0423	0.6290	0.0024	06.39	10.00	-0.0043	-0.0013	-0.0004	179.9	0.0042
0229	0.447	0.861	05.87	0.7607	-0.0976	0.1205	-0.7692	-0.0400	0.5786	0.0020	05.87	10.00	-0.0048	-0.0014	-0.0004	179.9	0.0047
0330	0.444	0.861	05.26	0.7207	-0.0883	0.1099	-0.7280	-0.0399	0.5194	0.0022	05.36	10.00	-0.0027	-0.0012	-0.0001	180.0	0.0036
0331	0.444	0.860	04.32	0.6477	-0.0720	0.0910	-0.6528	-0.0398	0.4193	0.0020	04.32	10.00	-0.0037	-0.0011	-0.0004	180.0	0.0036
0332	0.444	0.860	03.28	0.5627	-0.0595	0.0747	-0.5662	-0.0404	0.3164	0.0019	02.28	10.00	-0.0025	-0.0010	-0.0012	180.0	0.0024
0333	0.444	0.861	01.17	0.3525	-0.0400	0.0517	-0.3846	-0.0426	0.1249	0.0018	01.17	10.00	-0.0009	-0.0005	-0.0017	180.0	0.0008
0334	0.447	0.862	-00.98	0.0705	-0.0151	0.0409	-0.0699	-0.0405	0.0449	0.0017	-00.98	10.00	-0.0011	-0.0004	-0.0017	180.0	0.0010
0335	0.444	0.860	-03.14	-0.2339	0.0074	0.0470	0.2360	-0.0326	0.0546	0.0015	-03.14	10.00	-0.0013	-0.0001	-0.0018	180.0	-0.0014
0336	0.440	0.839	07.29	0.8169	-0.1092	0.1487	-0.8293	-0.0415	0.6671	0.0020	07.29	10.00	-0.0038	-0.0011	-0.0035	180.0	0.0037
0339	0.442	0.841	06.90	0.7978	-0.0809	0.1375	-0.8087	-0.0387	0.6365	0.0019	06.90	10.00	-0.0043	-0.0010	-0.0025	180.0	0.0042
0400	0.440	0.840	06.26	0.7876	-0.0928	0.1264	-0.7969	-0.0361	0.6202	0.0018	06.38	10.00	-0.0021	-0.0009	-0.0009	180.0	0.0020
0411	0.442	0.841	05.86	0.7714	-0.0875	0.1170	-0.7795	-0.0353	0.5950	0.0020	05.86	10.00	-0.0045	-0.0013	-0.002	179.9	0.0043
0422	0.442	0.840	05.37	0.7363	-0.0770	0.1053	-0.7431	-0.0340	0.5421	0.0019	05.37	10.00	-0.0040	-0.0013	-0.0004	180.0	0.0039
0433	0.442	0.840	04.34	0.6611	-0.0394	0.0867	-0.6658	-0.0345	0.4369	0.0020	04.34	10.00	-0.0043	-0.0011	-0.0007	180.0	0.0042

TABLE 3(K)  
FLARE PACK D/N  
ETR = 0°(EXTRACT) = +1/2°

SER. NO.	REVN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	INSTRAL.	ANAL.	CLSR.	BASE.	RINC.	SLIP.	CROSS.	VAN N.	ROLL N.	RANG.	SIDE F
044	0. 440	0. 839	03. 30	0. 9778	-0. 0450	0. 0700	-0. 5810	-0. 0347	0. 3337	0. 0016	03. 30	00. 00	-0. 0024	-0. 0009	0. 0014	180. 0	0. 0023
045	0. 442	0. 841	01. 18	0. 2487	-0. 0245	0. 0481	-0. 2497	-0. 0193	0. 1215	0. 0016	01. 18	00. 00	-0. 0011	-0. 0002	0. 0018	180. 0	0. 0010
046	0. 442	0. 839	-00. 97	0. 0638	-0. 0072	0. 0373	-0. 0633	-0. 0370	0. 0039	0. 0014	-00. 97	10. 00	0. 0001	-0. 0003	0. 0016	180. 0	-0. 0002
047	0. 442	0. 839	-03. 13	-0. 2188	0. 0048	0. 0430	0. 2207	-0. 0297	0. 0478	0. 0013	-03. 13	10. 00	0. 0011	-0. 0004	0. 0019	180. 0	-0. 0012
051	0. 447	0. 799	07. 39	0. 8161	-0. 0907	0. 1416	-0. 8276	-0. 0337	0. 6658	0. 0017	07. 39	10. 00	-0. 0035	-0. 0014	0. 0022	180. 0	0. 0034
052	0. 447	0. 799	06. 89	0. 8070	-0. 0892	0. 1309	-0. 8170	-0. 0343	0. 6512	0. 0019	06. 89	10. 00	-0. 0032	-0. 0011	0. 0027	180. 0	0. 0030
053	0. 447	0. 798	06. 79	0. 7874	-0. 0257	0. 1280	-0. 7959	-0. 0299	0. 6199	0. 0017	06. 79	10. 00	-0. 0036	-0. 0012	0. 0026	180. 0	0. 0035
054	0. 447	0. 799	05. 91	0. 7669	-0. 0375	0. 1101	-0. 7743	-0. 0288	0. 5880	0. 0018	05. 91	10. 00	-0. 0035	-0. 0011	0. 0017	180. 0	0. 0034
055	0. 450	0. 799	05. 39	0. 7595	-0. 0537	0. 0985	-0. 7655	-0. 0251	0. 5768	0. 0016	05. 39	10. 00	-0. 0046	-0. 0012	0. 0008	180. 0	0. 0045
056	0. 450	0. 799	04. 37	0. 6883	-0. 0125	0. 0796	-0. 6930	-0. 0251	0. 4744	0. 0016	04. 37	10. 00	-0. 0035	-0. 0009	0. 0014	180. 0	0. 0034
057	0. 450	0. 799	01. 32	0. 5868	-0. 0182	0. 0615	-0. 5824	-0. 0259	0. 3372	0. 0018	01. 32	10. 00	-0. 0026	-0. 0012	0. 0024	180. 0	0. 0025
058	0. 450	0. 802	01. 17	0. 3206	0. 0096	0. 0438	-0. 3216	-0. 0157	0. 1027	0. 0016	01. 17	10. 00	-0. 0007	-0. 0002	0. 0018	180. 0	0. 0006
059	0. 450	0. 801	00. 97	0. 0574	-0. 0017	0. 0370	-0. 0569	-0. 0366	0. 0031	0. 0014	-0. 97	10. 00	0. 0004	-0. 0003	0. 0018	180. 0	-0. 0005
060	0. 450	0. 800	-03. 12	-0. 2022	0. 0076	0. 0403	0. 2040	-0. 0282	0. 0408	0. 0013	-03. 12	10. 00	0. 0014	-0. 0003	0. 0021	180. 0	-0. 0015
063	0. 449	0. 750	07. 39	0. 8144	-0. 0693	0. 1281	-0. 8252	-0. 0277	0. 6632	0. 0017	07. 39	10. 00	-0. 0020	-0. 0004	0. 0032	180. 0	0. 0019
064	0. 449	0. 750	06. 89	0. 8027	-0. 0572	0. 1247	-0. 8120	-0. 0258	0. 6442	0. 0017	06. 89	10. 00	-0. 0028	-0. 0007	0. 0026	180. 0	0. 0027
065	0. 449	0. 749	06. 40	0. 7839	-0. 0344	0. 1129	-0. 7918	-0. 0239	0. 6144	0. 0018	06. 40	10. 00	-0. 0018	-0. 0006	0. 0042	180. 0	0. 0017
066	0. 449	0. 749	05. 88	0. 7718	-0. 0452	0. 1027	-0. 7784	-0. 0213	0. 5957	0. 0012	06. 88	10. 00	-0. 0019	-0. 0007	0. 0041	180. 0	0. 0014
067	0. 449	0. 751	05. 39	0. 7465	-0. 0226	0. 0922	-0. 7520	-0. 0197	0. 5723	0. 0019	05. 39	10. 00	-0. 0019	-0. 0008	0. 0029	180. 0	0. 0018
068	0. 445	0. 748	04. 35	0. 6629	-0. 0099	0. 0715	-0. 6675	-0. 0192	0. 4406	0. 0017	04. 35	10. 00	-0. 0023	-0. 0007	0. 0021	180. 0	0. 0022
069	0. 449	0. 750	01. 29	0. 8428	-0. 0082	0. 0573	-0. 5463	-0. 0244	0. 2986	0. 0016	01. 29	10. 00	-0. 0011	-0. 0004	0. 0023	180. 0	0. 0010
070	0. 449	0. 750	01. 15	0. 2978	-0. 0073	0. 0426	-0. 2986	-0. 0250	0. 0885	0. 0015	01. 15	10. 00	-0. 0005	-0. 0004	0. 0017	180. 0	0. 0004
071	0. 449	0. 751	00. 97	0. 0544	0. 0016	0. 0285	-0. 0529	-0. 0181	0. 0029	0. 0015	-0. 97	10. 00	-0. 0014	-0. 0006	0. 0017	180. 0	0. 0010
072	0. 449	0. 749	02. 10	-0. 1876	0. 0125	0. 0406	-0. 1894	-0. 0294	0. 0350	0. 0014	-02. 10	10. 00	0. 0011	-0. 0003	0. 0021	180. 0	-0. 0012
073	0. 440	0. 701	07. 38	0. 8129	-0. 0509	0. 1329	-0. 8233	-0. 0256	0. 6607	0. 0017	07. 38	10. 00	-0. 0016	-0. 0002	0. 0029	180. 0	0. 0015
076	0. 440	0. 701	06. 86	0. 8064	-0. 0584	0. 1187	-0. 8150	-0. 0196	0. 6502	0. 0019	06. 86	10. 00	-0. 0027	-0. 0002	0. 0044	180. 0	0. 0026
077	0. 440	0. 701	06. 26	0. 7857	-0. 0448	0. 1061	-0. 7927	-0. 0165	0. 6152	0. 0016	06. 26	10. 00	-0. 0019	-0. 0002	0. 0048	180. 0	0. 0018
078	0. 440	0. 701	05. 86	0. 7592	-0. 0312	0. 0942	-0. 7640	-0. 0145	0. 5746	0. 0017	05. 86	10. 00	-0. 0019	-0. 0003	0. 0041	180. 0	0. 0019
079	0. 440	0. 701	05. 25	0. 7235	-0. 0204	0. 0834	-0. 7282	-0. 0134	0. 5234	0. 0016	05. 25	10. 00	-0. 0020	-0. 0002	0. 0031	180. 0	0. 0019
080	0. 440	0. 699	04. 31	0. 6234	-0. 0099	0. 0632	-0. 6267	-0. 0164	0. 3886	0. 0017	04. 31	10. 00	-0. 0026	-0. 0002	0. 0028	180. 0	0. 0025
081	0. 437	0. 699	03. 25	0. 5108	-0. 0085	0. 0526	-0. 5432	-0. 0220	0. 2609	0. 0016	03. 25	10. 00	-0. 0018	-0. 0002	0. 0027	180. 0	0. 0017
082	0. 437	0. 700	03. 14	0. 2929	-0. 0060	0. 0327	-0. 2923	-0. 0226	0. 0797	0. 0015	01. 14	10. 00	-0. 0010	-0. 0004	0. 0017	180. 0	0. 0009
083	0. 437	0. 700	00. 98	0. 0519	0. 0032	0. 0365	-0. 0509	-0. 0162	0. 0026	0. 0018	-0. 98	10. 00	0. 0008	-0. 0002	0. 0018	180. 0	-0. 0001
084	0. 440	0. 701	03. 06	-0. 1776	0. 0152	0. 0288	-0. 1792	-0. 0280	0. 0214	0. 0012	-03. 06	10. 00	0. 0019	-0. 0008	0. 0020	180. 0	-0. 0009
087	0. 440	0. 498	07. 26	0. 7880	-0. 0518	0. 1498	-0. 7970	-0. 0173	0. 6210	0. 0019	07. 26	-10. 01	-0. 0020	0. 0001	0. 0027	180. 0	0. 0029
088	0. 440	0. 504	06. 77	0. 7897	-0. 0368	0. 1027	-0. 7669	-0. 0146	0. 5771	0. 0019	06. 77	-10. 01	-0. 0033	0. 0001	0. 0024	180. 0	0. 0024
089	0. 437	0. 498	06. 26	0. 7813	-0. 0284	0. 0903	-0. 7270	-0. 0094	0. 5202	0. 0016	06. 26	-10. 01	-0. 0031	0. 0001	0. 0023	180. 0	0. 0020

TABLE 3(K)  
FLARE PACK ON  
ETA = 0° EXTRACT = +1/2"

SER. NO.	REVN.	MACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSQ.	WEAR.	INC.	SLIP.	CROSS.	VAN. N.	ROLL. N.	RAND.	SIDE F.
090	0.440	0.500	05.74	0.6791	-0.0210	0.0794	-0.6937	-0.0094	0.4610	0.0020	05.74	-00.01	-0.0027	0.0004	0.0022	180.0	0.0026
091	0.440	0.500	05.23	0.6342	-0.0177	0.0707	-0.6282	-0.0105	0.4021	0.0021	05.23	-00.01	-0.0027	0.0001	0.0025	180.0	0.0026
092	0.437	0.499	04.19	0.5418	-0.0136	0.0572	-0.5447	-0.0155	0.2934	0.0019	04.19	-00.01	-0.0017	0.0001	0.0028	180.0	0.0016
093	0.440	0.502	03.16	0.4452	-0.0109	0.0461	-0.4473	-0.0213	0.1981	0.0019	03.16	-00.01	-0.0017	0.0000	0.0025	180.0	0.0010
094	0.440	0.500	04.09	0.2499	-0.0036	0.0374	-0.2507	-0.0310	0.0623	0.0017	01.09	00.00	-0.0008	-0.0002	0.0018	180.0	0.0007
095	0.440	0.501	-00.98	0.0455	0.0060	0.0345	-0.0491	-0.0116	0.0020	0.0016	-00.98	00.00	0.0003	-0.0004	0.0020	180.0	-0.0004
096	0.437	0.499	-03.05	-0.1561	0.0190	0.0374	0.1576	-0.0279	0.0243	0.0018	-03.05	00.00	0.0010	-0.0002	0.0022	180.0	-0.0011

TABLE 3(CL)  
FLARE PACK ON  
ETRA = +5° EXTRACT = +1/2"

SER REVN.	HATCH N.	HATCH S.	INCID.	LIFT	PITCH	DRAE	NORMA	AXIAL	C158.	BASE.	WING.	SUP.	CROSS.	VAN N.	ROLL N.	PNG.	SIDE F.
002 0. 444	0. 904	06. 36	0. 7836	-0. 1309	0. 1414	-0. 7946	-0. 0512	0. 6140	0. 0025	06. 36	-10. 04	0. 0014	-0. 0010	179. 9	-0. 0042		
003 0. 442	0. 899	04. 25	0. 6507	-0. 1522	0. 1000	-0. 6564	-0. 0492	0. 4233	0. 0022	04. 25	-10. 04	0. 0012	0. 0001	180. 0	-0. 0032		
004 0. 440	0. 901	02. 13	0. 4616	-0. 1550	0. 0711	-0. 4640	-0. 0517	0. 2129	0. 0022	02. 13	-10. 04	0. 0019	0. 0022	0. 0011	180. 0	-0. 0040	
005 0. 442	0. 899	00. 00	0. 2185	-0. 1276	0. 0511	-0. 2186	-0. 0490	0. 0477	0. 0024	00. 00	-10. 04	0. 0038	0. 0022	0. 0009	180. 0	-0. 0038	
006 0. 437	0. 879	06. 35	0. 7934	-0. 1379	0. 1366	-0. 8037	-0. 0457	0. 6293	0. 0022	06. 35	-10. 04	0. 0010	0. 0013	-0. 0010	179. 9	-0. 0011	
007 0. 437	0. 804	04. 25	0. 6599	-0. 1524	0. 0947	-0. 6652	-0. 0433	0. 4384	0. 0022	04. 25	-10. 04	0. 0008	0. 0015	-0. 0002	179. 9	-0. 0009	
008 0. 437	0. 879	02. 15	0. 4782	-0. 1339	0. 0664	-0. 4805	-0. 0464	0. 2285	0. 0020	02. 16	-10. 04	0. 0036	0. 0022	0. 0012	180. 0	-0. 0037	
009 0. 437	0. 876	00. 04	0. 2406	-0. 1088	0. 0472	-0. 2407	-0. 0450	0. 0576	0. 0024	00. 04	-10. 02	0. 0042	0. 0027	0. 0010	180. 0	-0. 0041	
010 0. 447	0. 860	06. 26	0. 7997	-0. 1317	0. 1334	-0. 8097	-0. 0417	0. 6393	0. 0022	06. 36	-10. 04	0. 0014	0. 0015	-0. 0009	179. 9	-0. 0015	
011 0. 447	0. 861	04. 27	0. 6702	-0. 1365	0. 0905	-0. 6752	-0. 0382	0. 4490	0. 0020	04. 27	-10. 04	0. 0011	0. 0015	0. 0000	180. 0	-0. 0012	
012 0. 444	0. 859	02. 10	0. 4941	-0. 1170	0. 0620	-0. 4962	-0. 0411	0. 2440	0. 0020	02. 18	-10. 04	0. 0045	0. 0024	0. 0013	180. 0	-0. 0046	
013 0. 444	0. 859	00. 06	0. 2399	-0. 0926	0. 0436	-0. 2401	-0. 0413	0. 0575	0. 0024	00. 05	-10. 02	0. 0044	0. 0025	0. 0014	180. 0	-0. 0044	
014 0. 447	0. 861	00. 05	0. 2394	-0. 0996	0. 1195	-0. 8018	-0. 0284	0. 6293	0. 0022	06. 38	-10. 04	0. 0015	0. 0016	0. 0011	180. 0	-0. 0012	
015 0. 440	0. 840	06. 28	0. 8003	-0. 1042	0. 1274	-0. 8096	-0. 0287	0. 6404	0. 0020	06. 38	-10. 04	0. 0009	0. 0004	0. 0011	179. 9	0. 0008	
016 0. 440	0. 840	04. 28	0. 6850	-0. 1272	0. 0879	-0. 6998	-0. 0343	0. 4691	0. 0023	04. 28	-10. 04	0. 0014	0. 0016	0. 0001	180. 0	-0. 0015	
017 0. 440	0. 840	02. 20	0. 4982	-0. 0995	0. 0575	-0. 5002	-0. 0365	0. 2480	0. 0019	02. 20	-10. 04	0. 0043	0. 0024	0. 0014	180. 0	-0. 0044	
018 0. 440	0. 839	00. 05	0. 2291	-0. 0780	0. 0299	-0. 2253	-0. 0179	0. 0506	0. 0019	00. 06	-10. 04	0. 0018	0. 0023	0. 0010	180. 0	-0. 0039	
019 0. 437	0. 837	00. 06	0. 238	-0. 0996	0. 1095	-0. 8018	-0. 0284	0. 6293	0. 0022	06. 38	-10. 04	0. 0011	0. 0015	0. 0009	180. 0	-0. 0014	
020 0. 435	0. 804	04. 33	0. 7129	-0. 1014	0. 0812	-0. 7171	-0. 0252	0. 5082	0. 0020	04. 33	-10. 04	0. 0013	0. 0020	0. 0009	180. 0	-0. 0014	
021 0. 435	0. 800	02. 21	0. 4749	-0. 0774	0. 0510	-0. 4767	-0. 0108	0. 2255	0. 0020	02. 21	-10. 04	0. 0041	0. 0024	0. 0014	180. 0	-0. 0042	
022 0. 435	0. 799	00. 05	0. 2106	-0. 0709	0. 0293	-0. 2107	-0. 0173	0. 0442	0. 0019	00. 05	-10. 04	0. 0044	0. 0024	0. 0011	180. 0	-0. 0045	
023 0. 437	0. 750	06. 34	0. 8426	-0. 1196	0. 1141	-0. 8213	-0. 0216	0. 6617	0. 0020	06. 34	-10. 04	0. 0025	0. 0019	0. 0016	180. 0	-0. 0026	
024 0. 437	0. 749	04. 10	0. 6961	-0. 0738	0. 0710	-0. 6896	-0. 0173	0. 4706	0. 0020	04. 30	-10. 04	0. 0021	0. 0019	0. 0016	180. 0	-0. 0022	
025 0. 437	0. 730	02. 17	0. 4420	-0. 0729	0. 0486	-0. 4427	-0. 0200	0. 1952	0. 0020	02. 17	-10. 04	0. 0028	0. 0019	0. 0014	180. 0	-0. 0023	
026 0. 437	0. 751	00. 05	0. 1964	-0. 0660	0. 0380	-0. 1955	-0. 0160	0. 0384	0. 0019	00. 05	-10. 04	0. 0023	0. 0021	0. 0010	180. 0	-0. 0040	
027 0. 437	0. 749	06. 31	0. 8046	-0. 1046	0. 1074	-0. 8116	-0. 0166	0. 6472	0. 0017	06. 34	-10. 04	0. 0027	0. 0022	0. 0019	180. 0	-0. 0028	
028 0. 437	0. 697	00. 05	0. 2291	-0. 0780	0. 0299	-0. 2253	-0. 0179	0. 0506	0. 0019	00. 06	-10. 04	0. 0028	0. 0023	0. 0016	180. 0	-0. 0029	
029 0. 437	0. 750	06. 34	0. 8426	-0. 1196	0. 1141	-0. 8213	-0. 0216	0. 6617	0. 0020	06. 34	-10. 04	0. 0028	0. 0023	0. 0020	180. 0	-0. 0030	
030 0. 434	0. 697	06. 31	0. 8046	-0. 1046	0. 1074	-0. 8116	-0. 0166	0. 6472	0. 0017	06. 34	-10. 04	0. 0027	0. 0022	0. 0020	180. 0	-0. 0029	
031 0. 434	0. 696	04. 25	0. 6447	-0. 0747	0. 0655	-0. 6480	-0. 0165	0. 4155	0. 0020	04. 25	-10. 04	0. 0028	0. 0023	0. 0022	180. 0	-0. 0029	
032 0. 436	0. 701	02. 15	0. 1120	-0. 0710	0. 0662	-0. 1205	-0. 0285	0. 1735	0. 0019	02. 15	-10. 04	0. 0023	0. 0020	0. 0012	180. 0	-0. 0030	
033 0. 437	0. 700	00. 04	0. 1893	-0. 0620	0. 0173	-0. 1894	-0. 0154	0. 0258	0. 0017	00. 04	-10. 04	0. 0042	0. 0024	0. 0010	180. 0	-0. 0042	
034 0. 437	0. 699	06. 22	0. 7443	-0. 0867	0. 0923	-0. 7500	-0. 0089	0. 5529	0. 0020	06. 23	-10. 04	0. 0032	0. 0023	0. 0016	180. 0	-0. 0013	
035 0. 437	0. 699	04. 17	0. 3629	-0. 0739	0. 0586	-0. 5657	-0. 0154	0. 3166	0. 0028	04. 17	-10. 04	0. 0028	0. 0020	0. 0022	180. 0	-0. 0029	
036 0. 437	0. 699	02. 10	0. 3696	-0. 0652	0. 0423	-0. 3769	-0. 0269	0. 1354	0. 0019	00. 50	-10. 04	0. 0036	0. 0019	0. 0014	180. 0	-0. 0037	
037 0. 437	0. 699	00. 03	0. 1789	-0. 0560	0. 0261	-0. 1706	-0. 0341	0. 0289	0. 0019	00. 51	-10. 04	0. 0041	0. 0021	0. 0010	180. 0	-0. 0042	

TABLE 3(A)  
FLARE PACK ON  
ETRA = +10° EXTRACT = +1/2"

SER. REYN.	HACH.	INCID.	LIFT.	PITCH.	DRAE	NORMAL	AXIAL.	CLSR.	BASE.	RINC.	SLIP.	CROSS.	VAN H.	ROLL H.	RNG.	SIDE F
002	0.457	0.902	06.35	0.7770	-0.1348	0.1480	-0.7895	-0.0585	0.6048	0.0025	06.35	-10.02	0.0023	0.0023	-0.0014	179.3 -0.0024
003	0.455	0.900	04.24	0.6467	-0.1623	0.1032	-0.6327	-0.0528	0.4182	0.0023	04.24	-10.04	0.0020	0.0017	0.0002	180.0 -0.0024
004	0.457	0.904	02.10	0.4737	-0.1975	0.0724	-0.4762	-0.0529	0.2243	0.0024	02.10	-10.01	0.0023	0.0020	0.0009	180.0 -0.0034
005	0.457	0.901	-00.04	0.2364	-0.1865	0.0532	-0.2364	-0.0513	0.0557	0.0021	-00.04	-10.01	0.0041	0.0022	0.0008	180.0 -0.0042
006	0.450	0.880	06.34	0.7854	-0.1444	0.1373	-0.7959	-0.0473	0.6167	0.0024	06.34	-10.02	0.0025	0.0024	-0.0010	179.9 -0.0025
007	0.450	0.881	04.24	0.6584	-0.1690	0.0972	-0.6639	-0.0462	0.4334	0.0024	04.24	-10.04	0.0027	0.0018	0.0000	180.0 -0.0028
008	0.450	0.880	02.11	0.4980	-0.1950	0.0680	-0.5003	-0.0484	0.2479	0.0020	02.11	-10.02	0.0060	0.0028	0.0011	180.0 -0.0061
009	0.450	0.879	-00.02	0.2646	-0.1794	0.0495	-0.2647	-0.0477	0.0692	0.0019	-00.02	-10.01	0.0051	0.0026	0.0009	180.0 -0.0052
010	0.460	0.861	06.35	0.7941	-0.1268	0.1348	-0.8042	-0.0437	0.6306	0.0024	06.35	-10.04	0.0007	0.0017	-0.0006	179.9 -0.0007
011	0.458	0.859	04.25	0.6729	-0.1606	0.0939	-0.6781	-0.0417	0.4527	0.0020	04.25	-10.04	0.0016	0.0017	0.0003	180.0 -0.0017
012	0.457	0.861	02.12	0.5145	-0.1853	0.0643	-0.5166	-0.0433	0.2645	0.0019	02.12	-10.02	0.0042	0.0025	0.0012	180.0 -0.0043
013	0.457	0.860	-00.04	0.2648	-0.1663	0.0455	-0.2649	-0.0437	0.0709	0.0019	-00.04	-10.02	0.0043	0.0027	0.0011	180.0 -0.0046
014	0.457	0.860	06.35	0.8068	-0.1421	0.1257	-0.8158	-0.0375	0.6225	0.0022	06.34	-10.01	-0.0081	0.0012	0.0009	180.0 -0.0000
015	0.453	0.840	06.34	0.7890	-0.1360	0.1279	-0.7985	-0.0377	0.5335	0.0019	04.29	-10.01	0.0014	0.0017	0.0008	180.0 -0.0015
016	0.453	0.839	04.26	0.6947	-0.1636	0.0900	-0.6996	-0.0362	0.4823	0.0019	04.26	-10.01	0.0015	0.0015	0.0003	180.0 -0.0016
017	0.453	0.839	02.15	0.5152	-0.1542	0.0535	-0.5017	-0.0329	0.0129	0.0019	02.15	-10.02	0.0044	0.0026	0.0014	180.0 -0.0045
018	0.453	0.840	02.14	0.5261	-0.1769	0.0687	-0.5281	-0.0192	0.2766	0.0018	02.14	-10.02	0.0044	0.0049	0.0009	180.0 -0.0030
019	0.450	0.840	00.00	0.2525	-0.1539	0.0424	-0.2526	-0.0405	0.0637	0.0019	00.00	-10.02	0.0032	0.0028	0.0009	180.0 -0.0023
020	0.470	0.801	06.35	0.8068	-0.1421	0.1257	-0.8158	-0.0375	0.6307	0.0018	06.35	-10.01	-0.0081	0.0014	0.0011	180.0 -0.0000
021	0.470	0.802	04.29	0.7305	-0.1555	0.0834	-0.7348	-0.0267	0.5182	0.0018	02.12	-10.01	0.0014	0.0017	0.0008	180.0 -0.0015
022	0.467	0.799	02.15	0.4999	-0.1542	0.0535	-0.5017	-0.0329	0.0129	0.0019	02.15	-10.02	0.0039	0.0025	0.0013	180.0 -0.0040
023	0.467	0.799	00.00	0.2382	-0.1465	0.0405	-0.2383	-0.0389	0.0367	0.0018	00.00	-10.02	0.0049	0.0024	0.0009	180.0 -0.0030
024	0.447	0.752	06.35	0.8314	-0.1651	0.1175	-0.8194	-0.0233	0.6911	0.0024	06.34	-10.04	0.0021	0.0018	0.0006	180.0 -0.0022
025	0.447	0.750	04.25	0.7158	-0.1532	0.0735	-0.7194	-0.0189	0.5123	0.0018	04.25	-10.01	0.0028	0.0024	0.0016	180.0 -0.0028
026	0.449	0.750	02.12	0.4573	-0.1484	0.0514	-0.4690	-0.0323	0.2182	0.0018	02.12	-10.01	0.0035	0.0026	0.0013	180.0 -0.0016
027	0.449	0.751	00.00	0.2255	-0.1406	0.0393	-0.2256	-0.0376	0.0308	0.0018	00.00	-10.01	0.0043	0.0025	0.0008	180.0 -0.0044
028	0.449	0.752	06.27	0.8322	-0.1722	0.1115	-0.8405	-0.0178	0.6941	0.0020	06.27	-10.01	0.0031	0.0023	0.0039	180.0 -0.0032
029	0.449	0.752	00.00	0.2255	-0.1406	0.0393	-0.2256	-0.0376	0.0308	0.0018	00.00	-10.01	0.0043	0.0027	0.0022	180.0 -0.0028
030	0.426	0.699	06.35	0.8314	-0.1651	0.1175	-0.8194	-0.0233	0.6911	0.0024	06.34	-10.04	0.0027	0.0018	0.0006	180.0 -0.0022
031	0.426	0.700	02.10	0.4462	-0.1449	0.0477	-0.4477	-0.0295	0.1989	0.0018	02.10	-10.04	0.0034	0.0023	0.0012	180.0 -0.0034
032	0.426	0.700	00.00	0.2255	-0.1406	0.0393	-0.2256	-0.0376	0.0308	0.0018	00.00	-10.04	0.0043	0.0026	0.0009	180.0 -0.0046
033	0.427	0.701	00.00	0.2181	-0.1353	0.0385	-0.2182	-0.0368	0.0474	0.0018	00.00	-10.04	0.0015	0.0025	0.0016	180.0 -0.0016
034	0.444	0.500	06.20	0.7725	-0.1564	0.0962	-0.7785	-0.0100	0.5967	0.0023	06.20	-10.04	0.0037	0.0027	0.0022	180.0 -0.0036
035	0.444	0.502	04.13	0.5905	-0.1436	0.0618	-0.5936	-0.0167	0.3486	0.0023	04.13	-10.04	0.0037	0.0037	0.0037	180.0 -0.0036
036	0.444	0.501	02.07	0.3980	-0.1339	0.0447	-0.3999	-0.0282	0.1586	0.0024	02.07	-10.04	0.0044	0.0025	0.0014	180.0 -0.0042
037	0.444	0.500	00.00	0.1987	-0.1234	0.0369	-0.1988	-0.0151	0.0394	0.0019	00.00	-10.04	0.0051	0.0027	0.0012	180.0 -0.0032

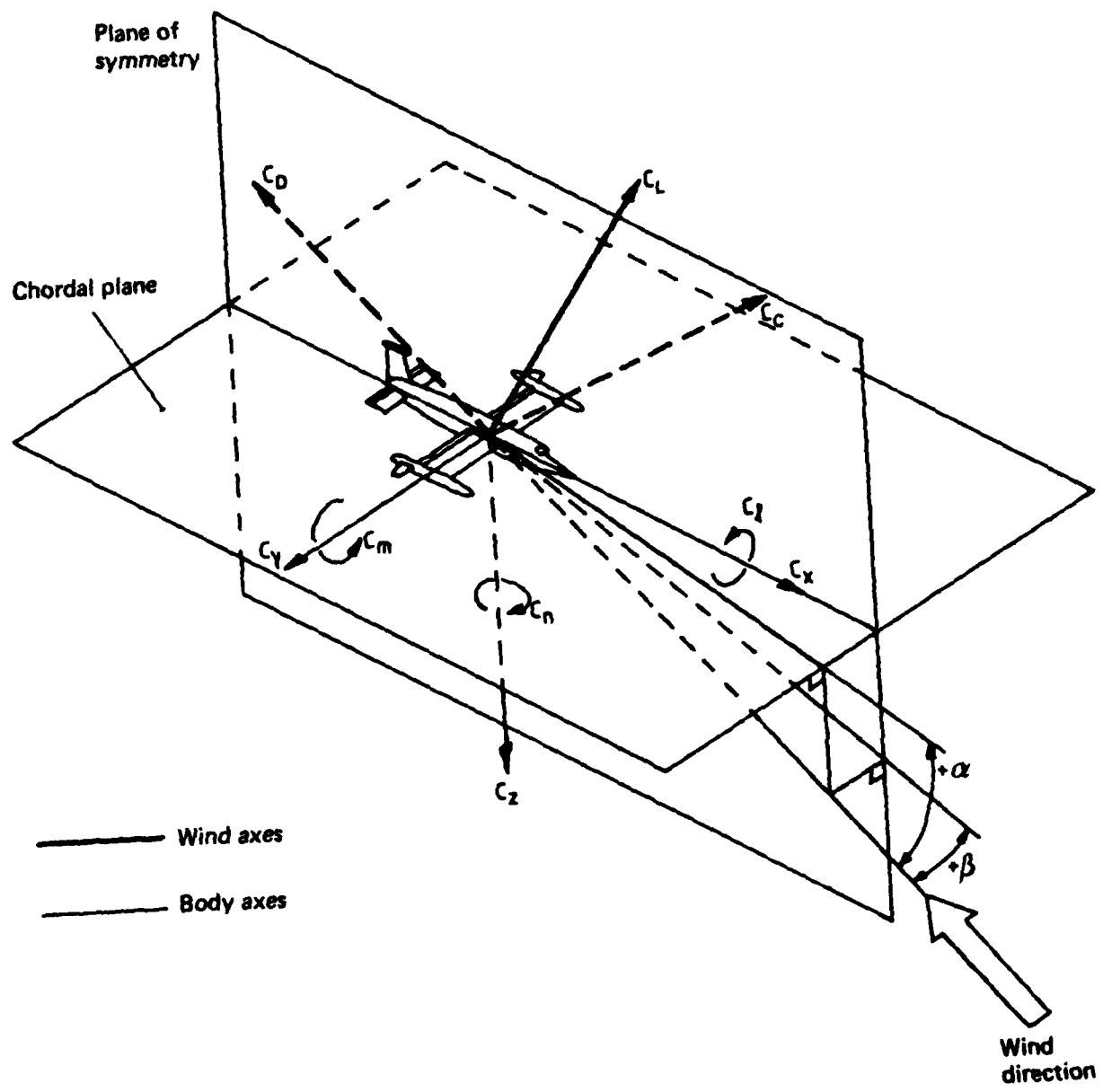


FIG. 1 FORCE AND MOMENT AXES SYSTEM

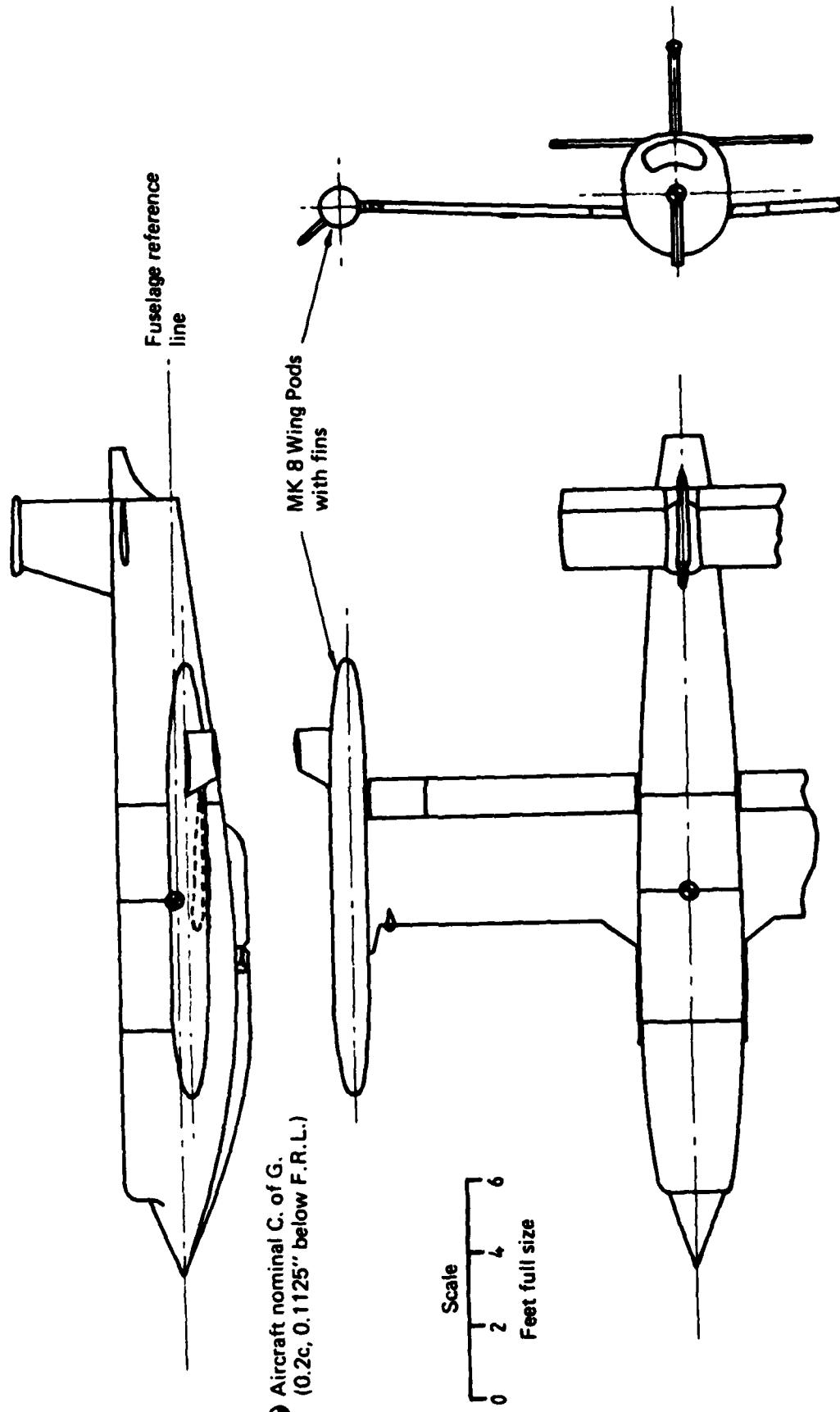


FIG. 2 SKETCH OF MODEL

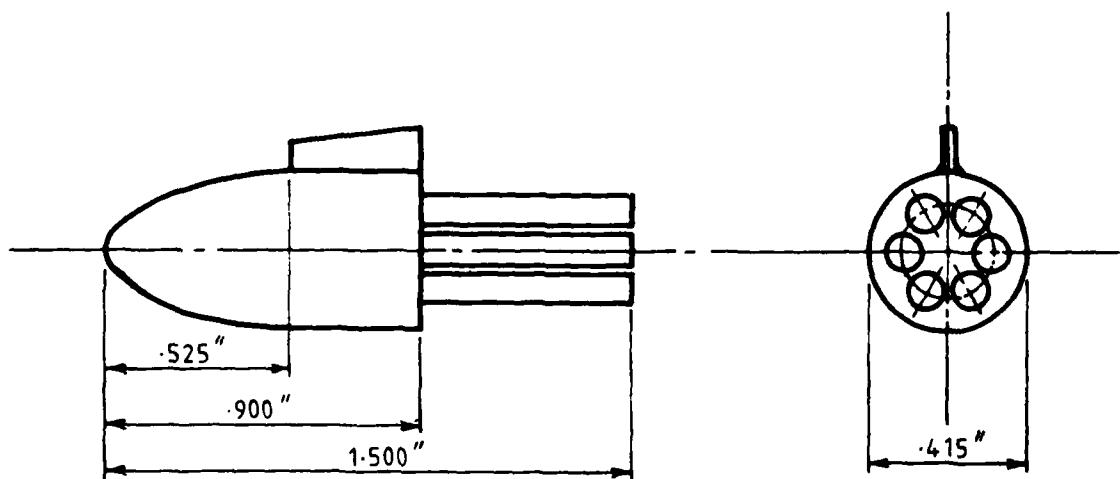
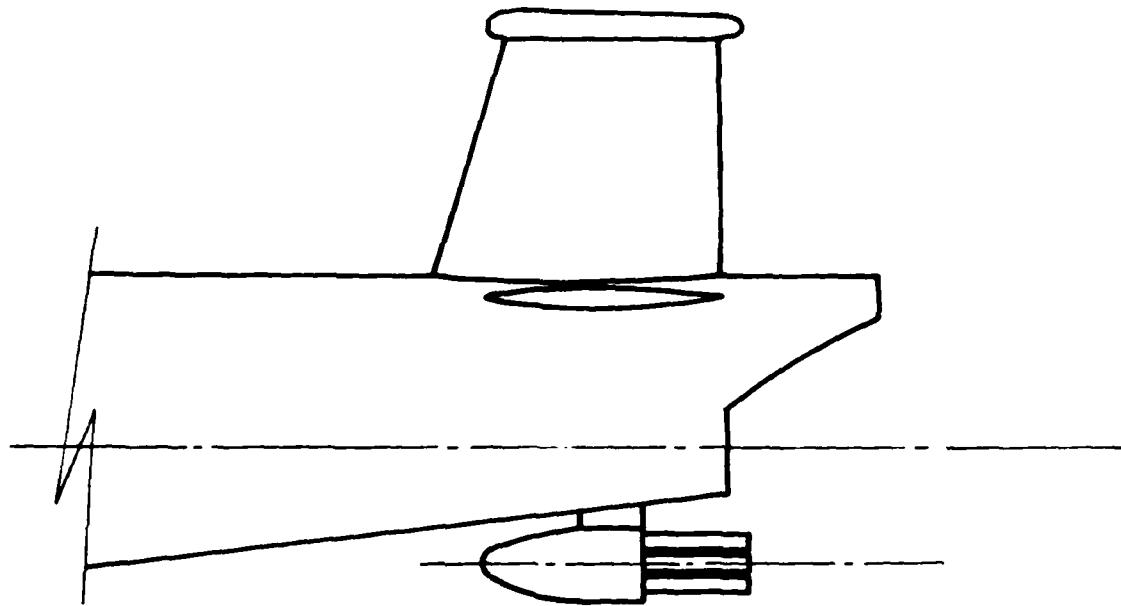


FIG. 3 SKETCH OF FLARE PACK

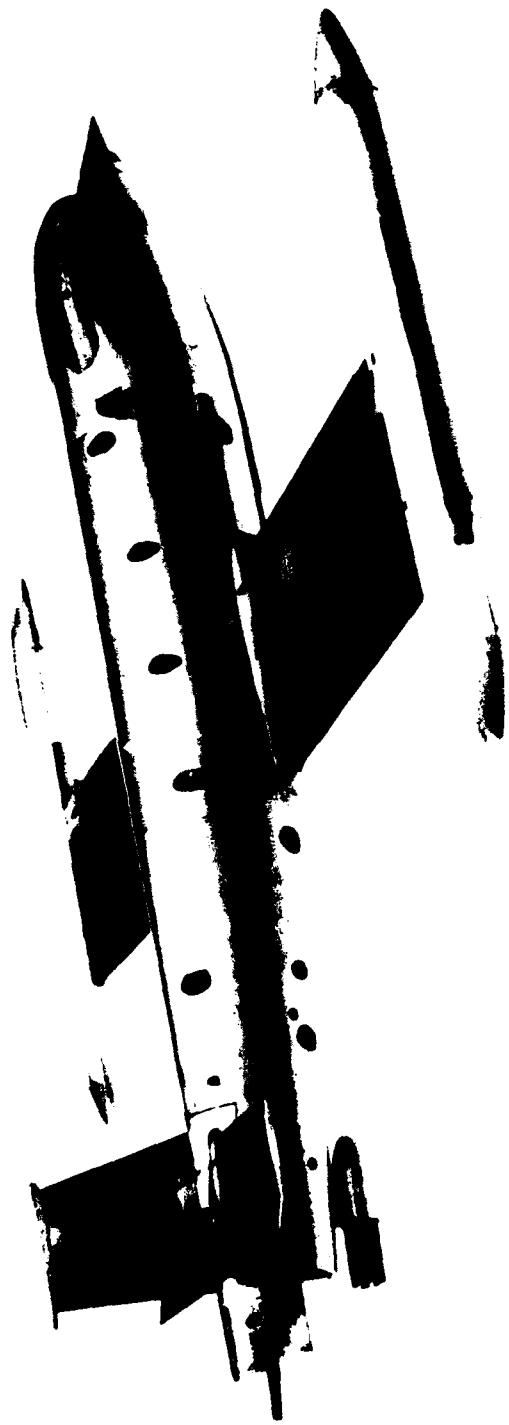


FIG. 4 PHOTOGRAPH OF MODEL - THREE QUARTER REAR VIEW



FIG. 5 PHOTOGRAPH OF MODEL - THREE QUARTER FRONT VIEW

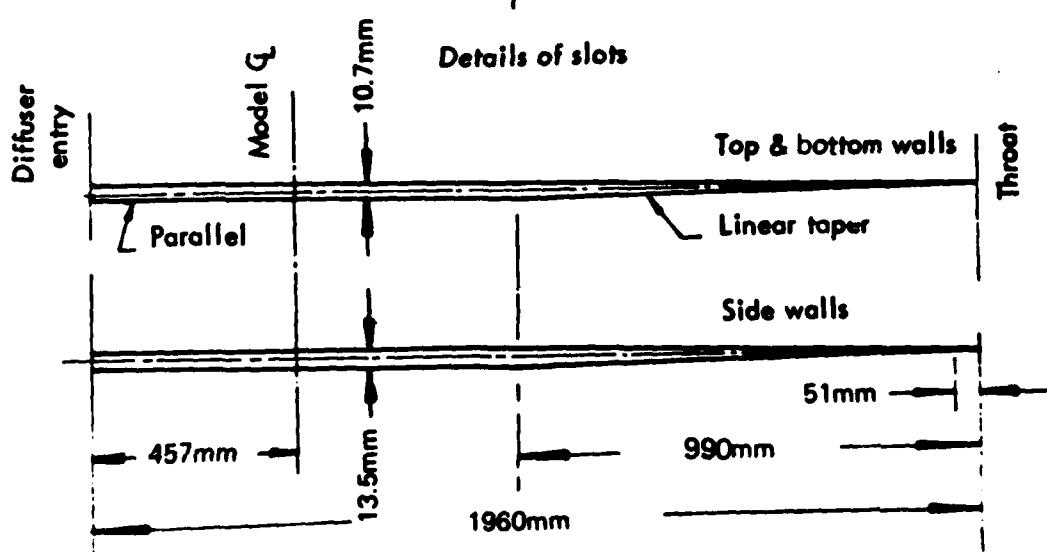
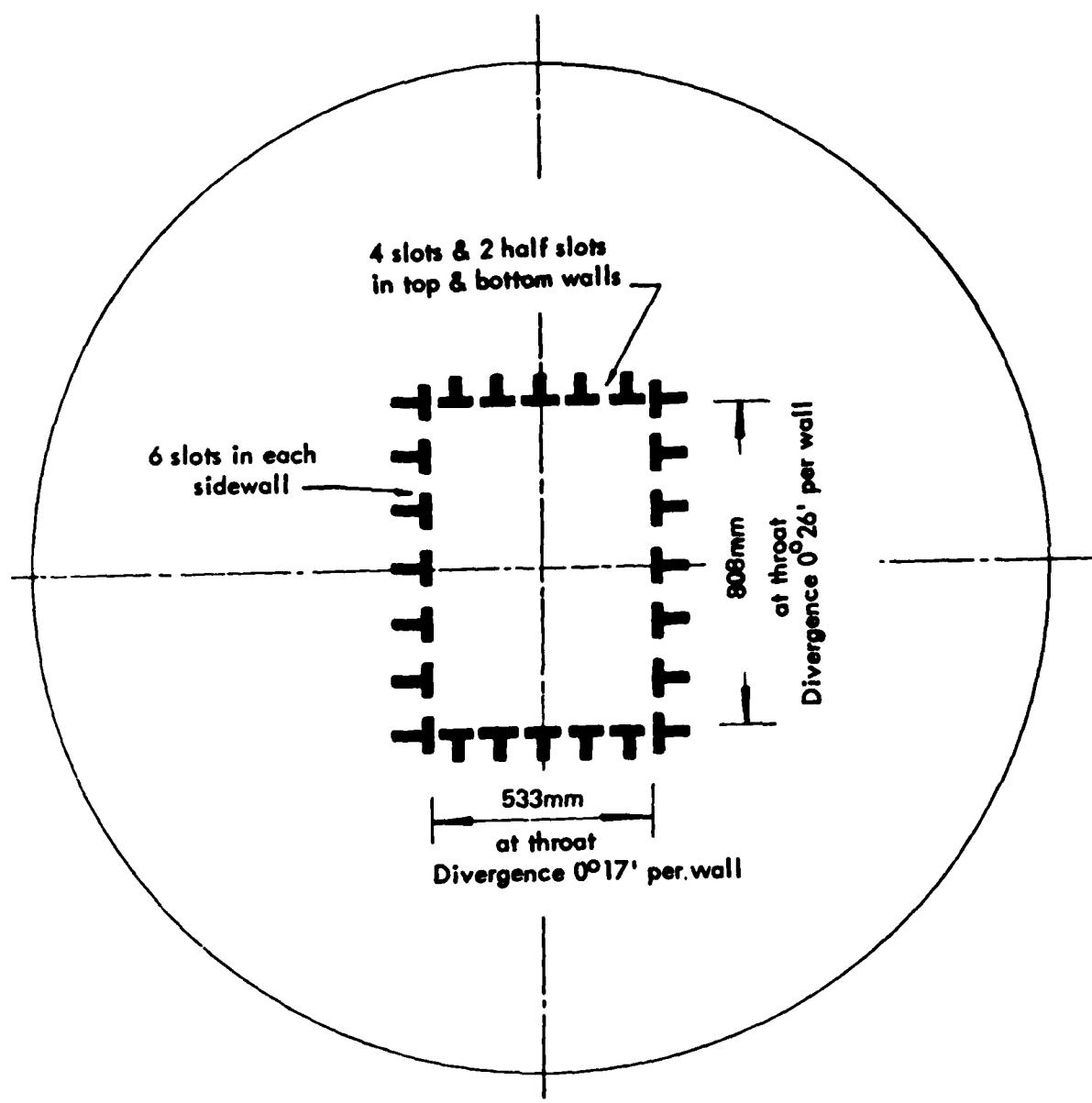
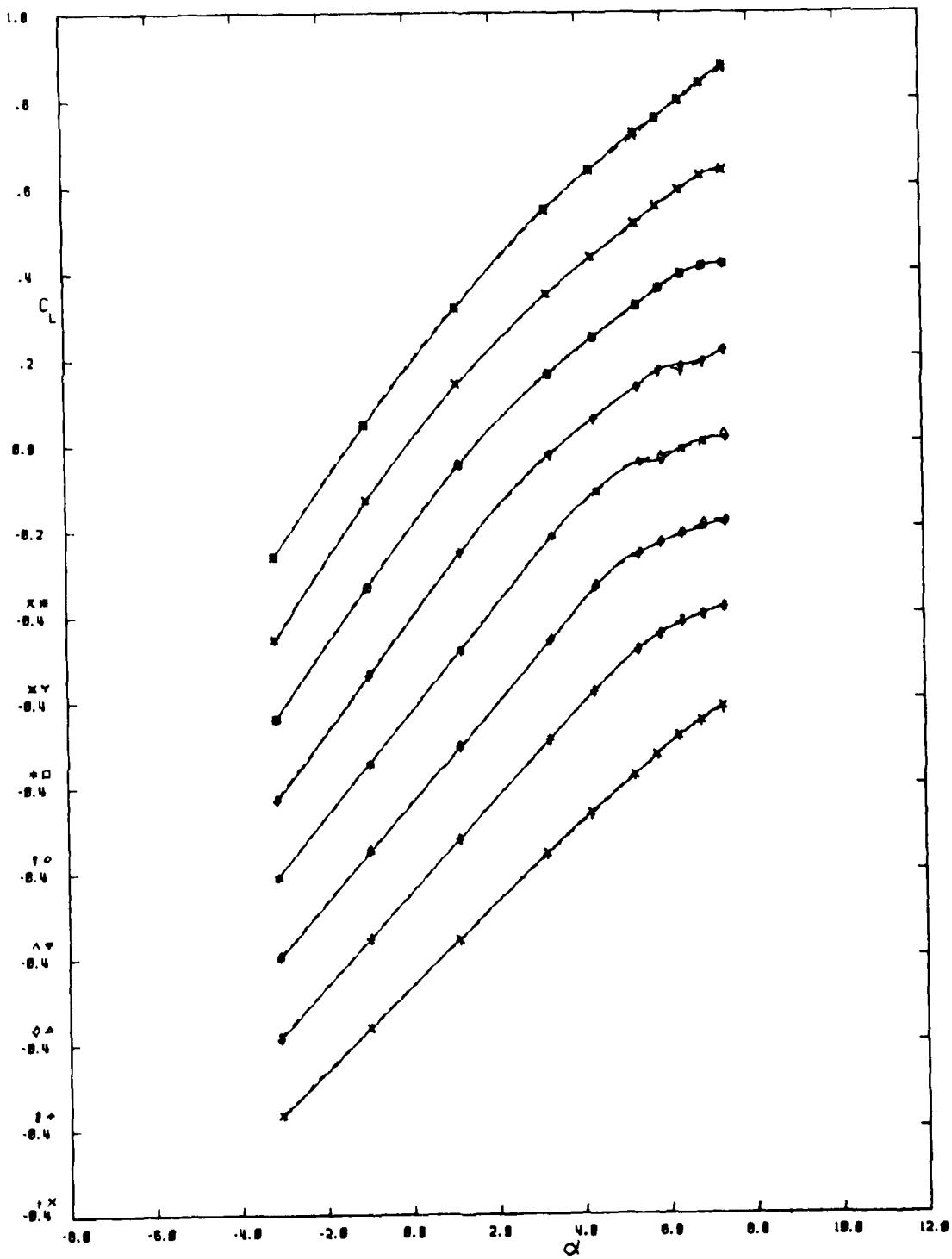


FIG. 6 DETAILS OF SLOTTED TEST SECTION

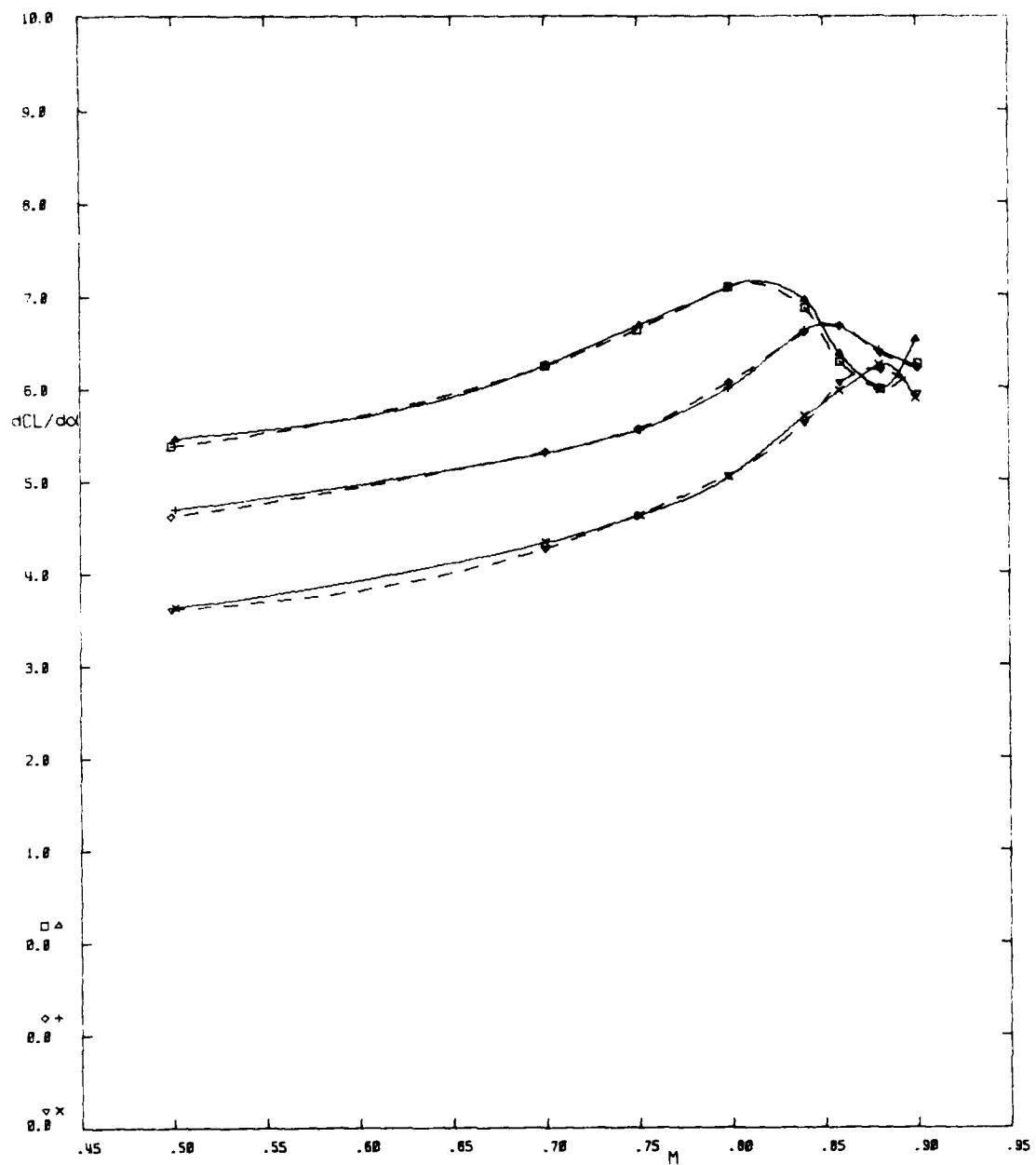


SYMBOL	M	SYMBOL	M
$x$	.50	$\wedge$	.50
$+$	.70	$\circ$	.70
$\Delta$	.75	$\square$	.75
$\diamond$	.80	$\triangledown$	.80
$\square$	.85	$\wedge$	.85
$\triangledown$	.90	$\diamond$	.90
$\circ$	.90	$x$	.90

CLEAN AIRCRAFT

FLARE PACK ON

FIGURE 7. VARIATION OF LIFT COEFFICIENT WITH INCIDENCE.



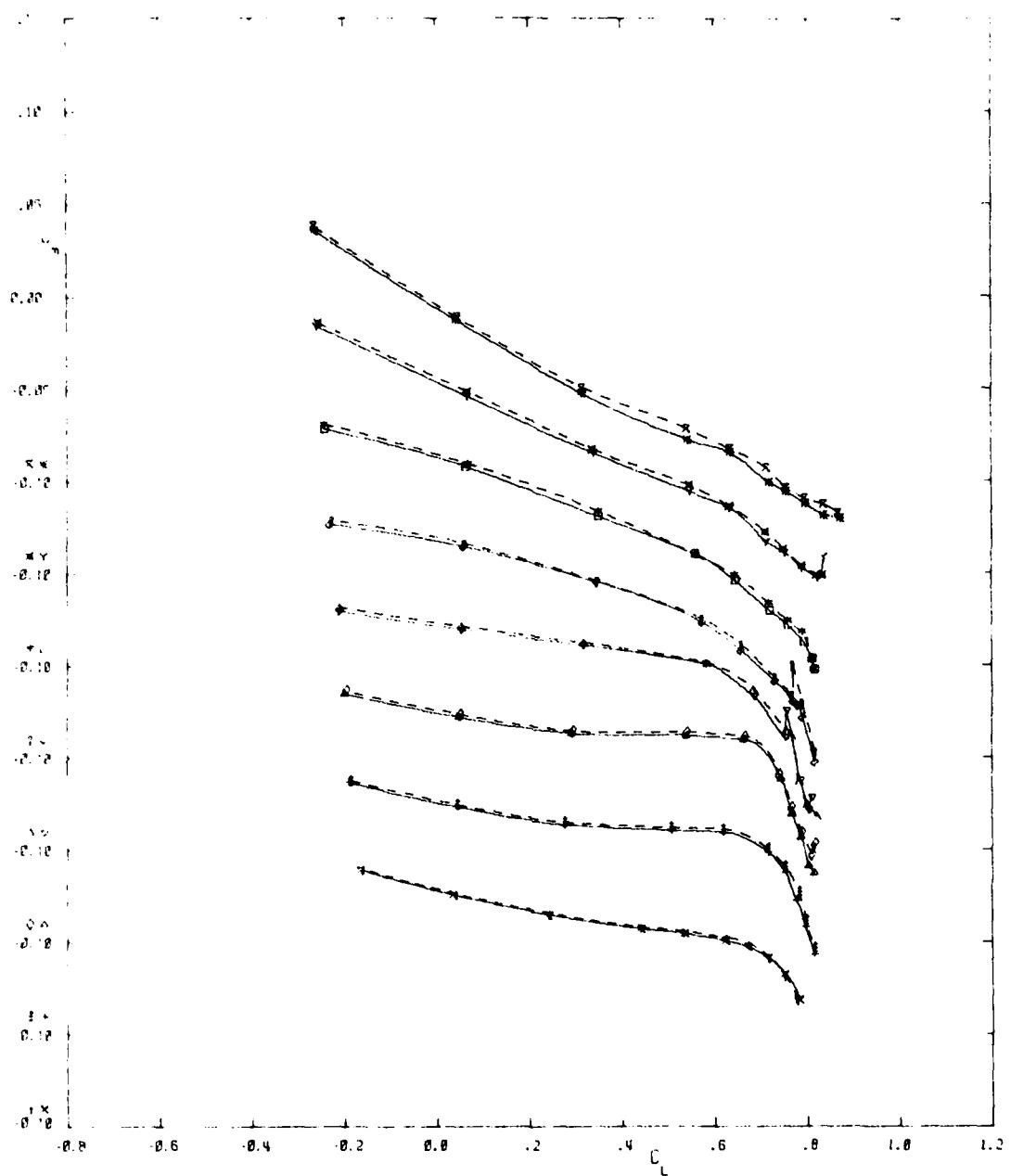
SYMBOL       $C_L$

---

CLEAN AIRCRAFT     
  $\times$       .00  
 $\triangle$       .20  
 $\diamond$       .40  
 $\nabla$       .00  
 $\diamond$       .20

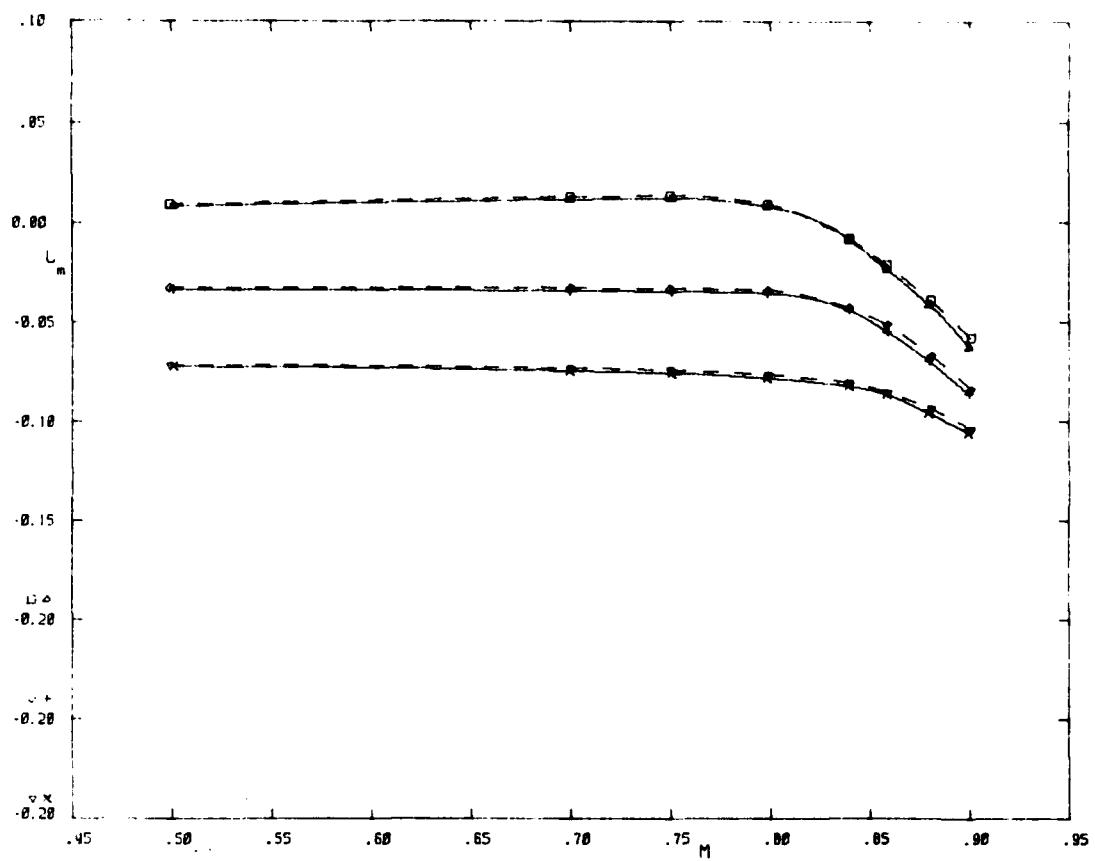
FLARE PACK ON     
  $\square$       .00  
 $\square$       .40

FIGURE 8. VARIATION OF LIFT CURVE SLOPE WITH MACH NUMBER.



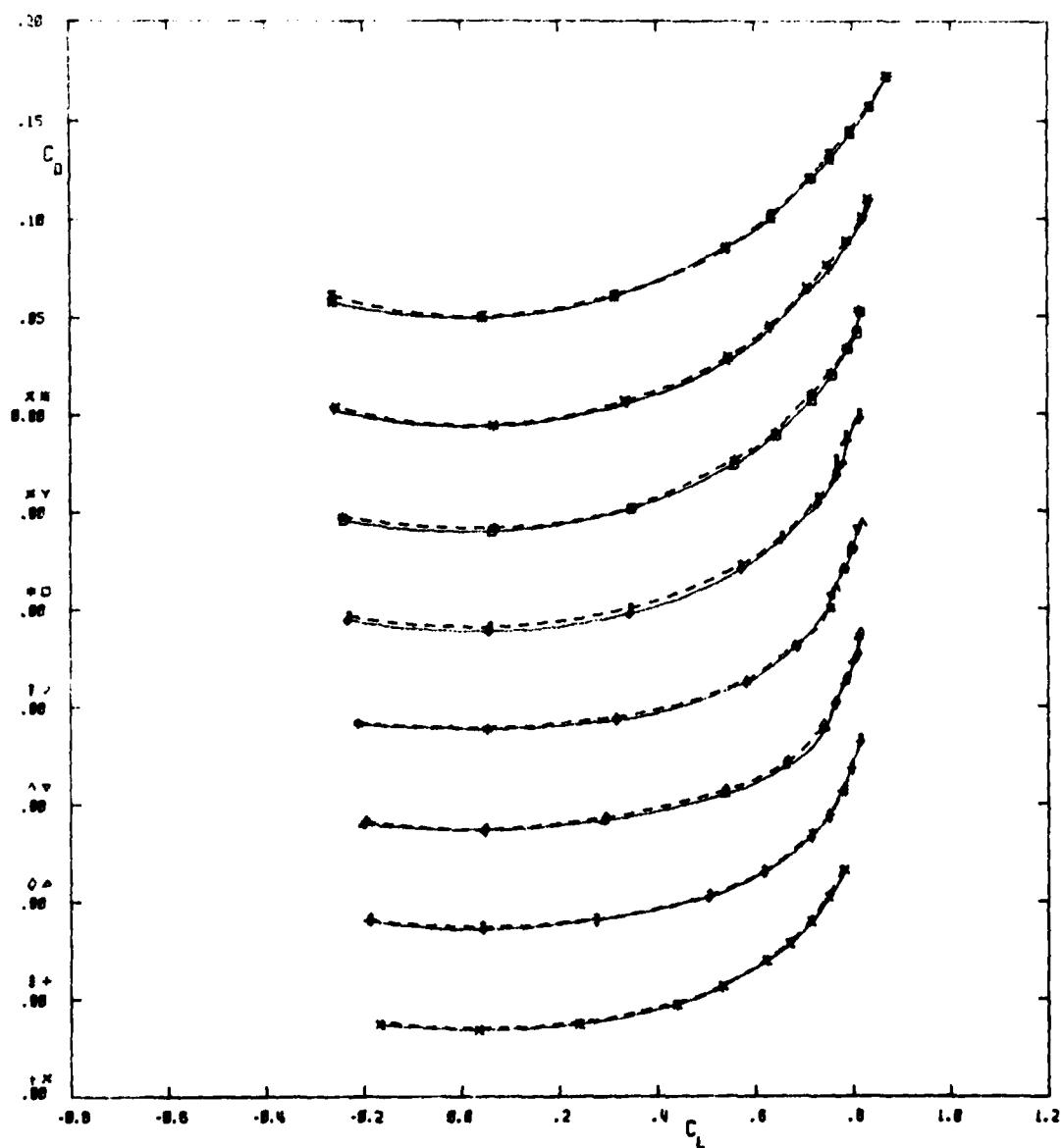
	SYMBOL	M	SYMBOL	M
CLEAN AIRCRAFT	x	.50	†	.50
	*	.70	‡	.70
	△	.75	○	.75
	▽	.80	FLARE PICK ON	
	◊	.84	†	.84
	○	.86	*	.86
	▽	.88	□	.88
	■	.90	×	.90

FIGURE 9. VARIATION OF PITCHING MOMENT COEFFICIENT WITH LIFT COEFFICIENT



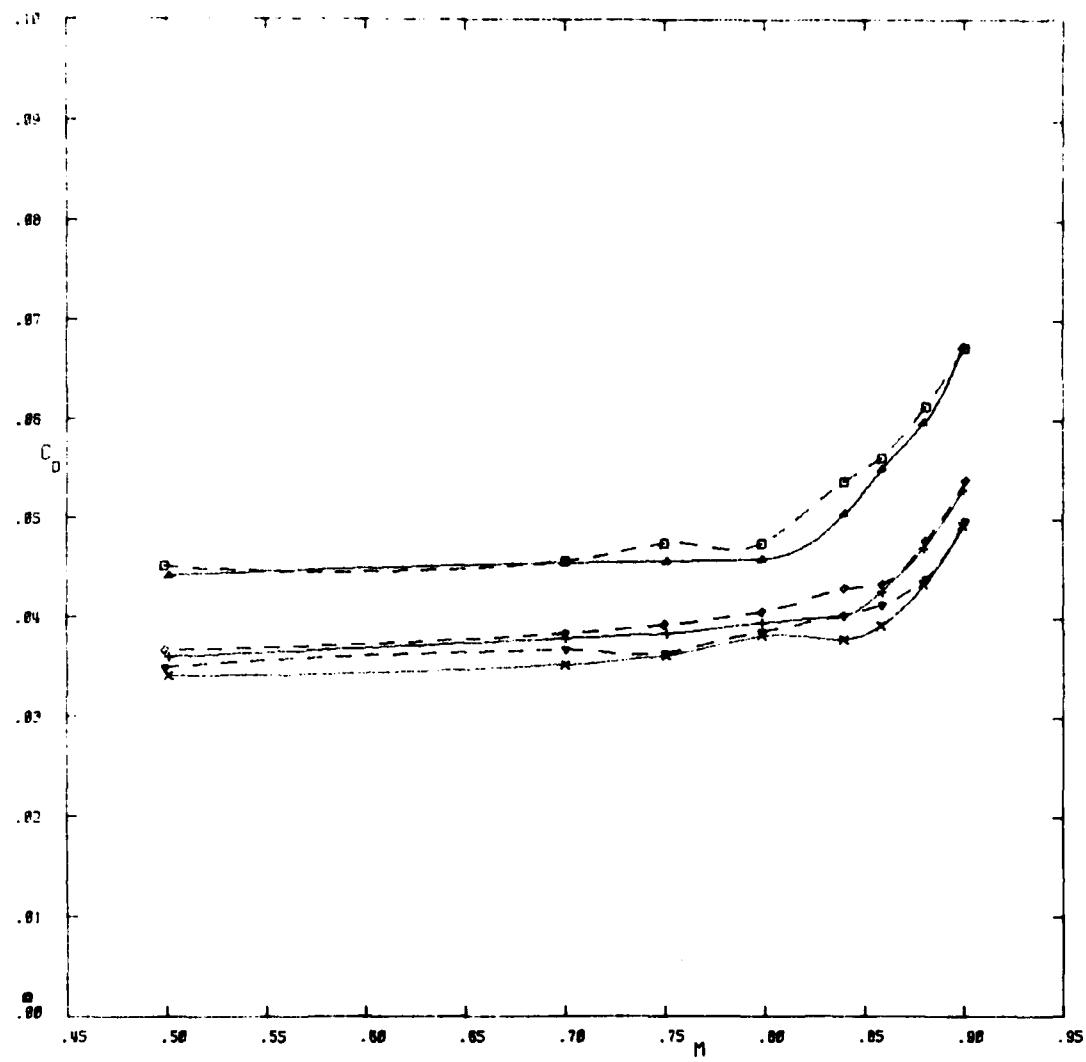
SYMBOL	$C_m$
<u>CLEAN AIRCRAFT</u>	
x	.00
+	.20
△	.40
▼	.60
<u>FLARE PACK ON</u>	
◊	.20
□	.40

FIGURE 10. VARIATION OF PITCHING MOMENT COEFFICIENT WITH MACH NUMBER.



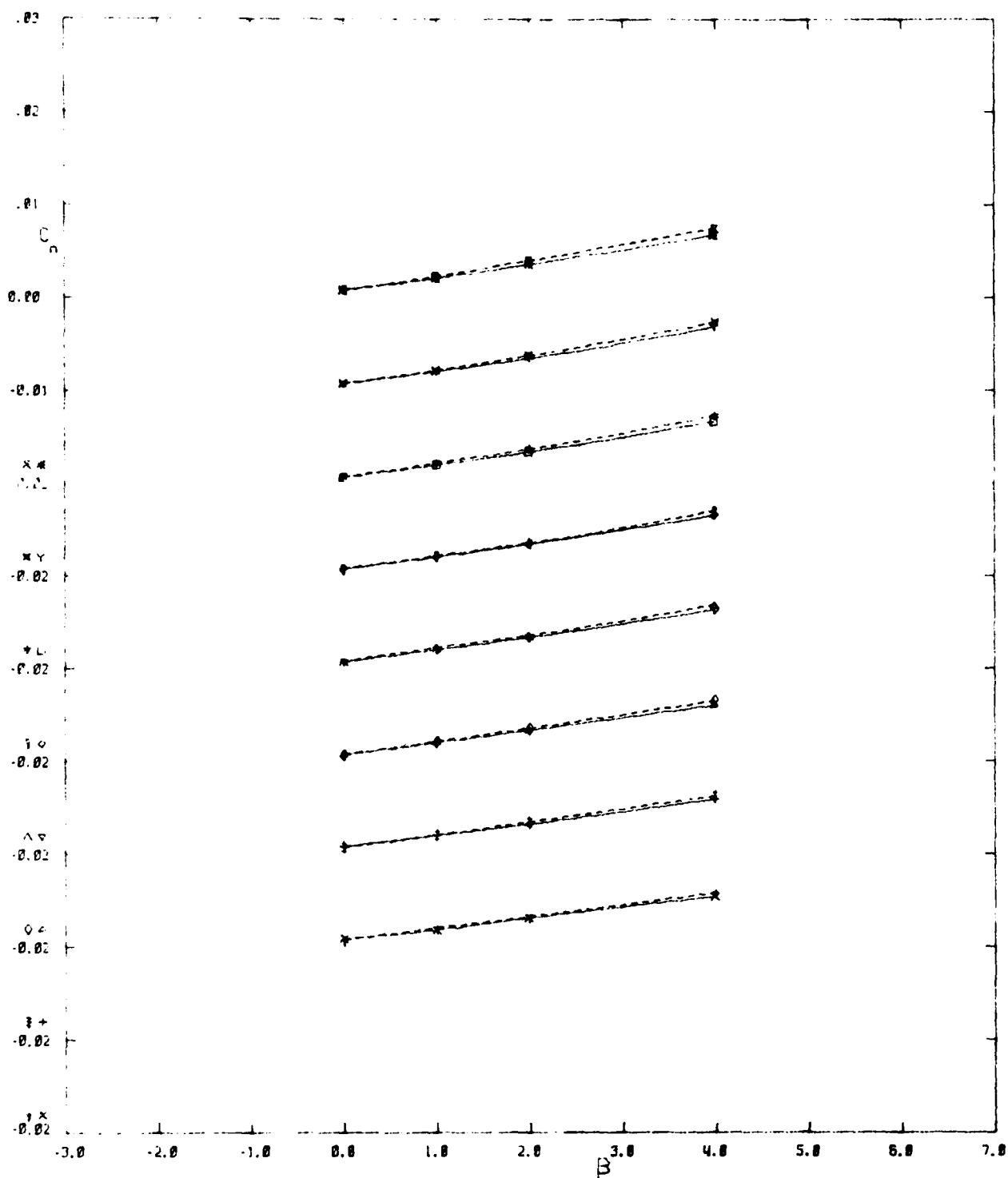
SYMBOL	M	SYMBOL	M
X	.50	†	.50
+	.70	‡	.70
△	.75	○	.75
▼	.80	◊	.80
□	.85	◆	.85
▽	.90	×	.90
■	.95	✗	.95
<hr/>		<hr/>	
CLEAN AIRCRAFT		FLARE PACK ON	
◆	.98	▲	.98
○	—	†	.98
□	—	‡	.98
▽	—	○	.98
■	—	◆	.98

FIGURE 11. VARIATION OF DRAC COEFFICIENT WITH LIFT COEFFICIENT.



SYMBOL	$C_d$
CLEAN AIRCRAFT	.06
+ .28	.08
△ .48	.09
▼ .08	
FLARE PACK ON	.28
□ .48	

FIGURE 12. VARIATION OF DRAG COEFFICIENT WITH MACH NUMBER.



SYMBOL	M	SYMBOL	M
X	.58	†	.58
+	.78	‡	.78
△	.75	◊	.75
*	.88	▲	.88
□	.84	†	.84
Y	.88	‡	.88
*L	.98	◊	.98

CLEAN AIRCRAFT      FLARE PACK ON

FIGURE 13. VARIATION OF YAWING MOMENT COEFFICIENT WITH SIDESLIP.

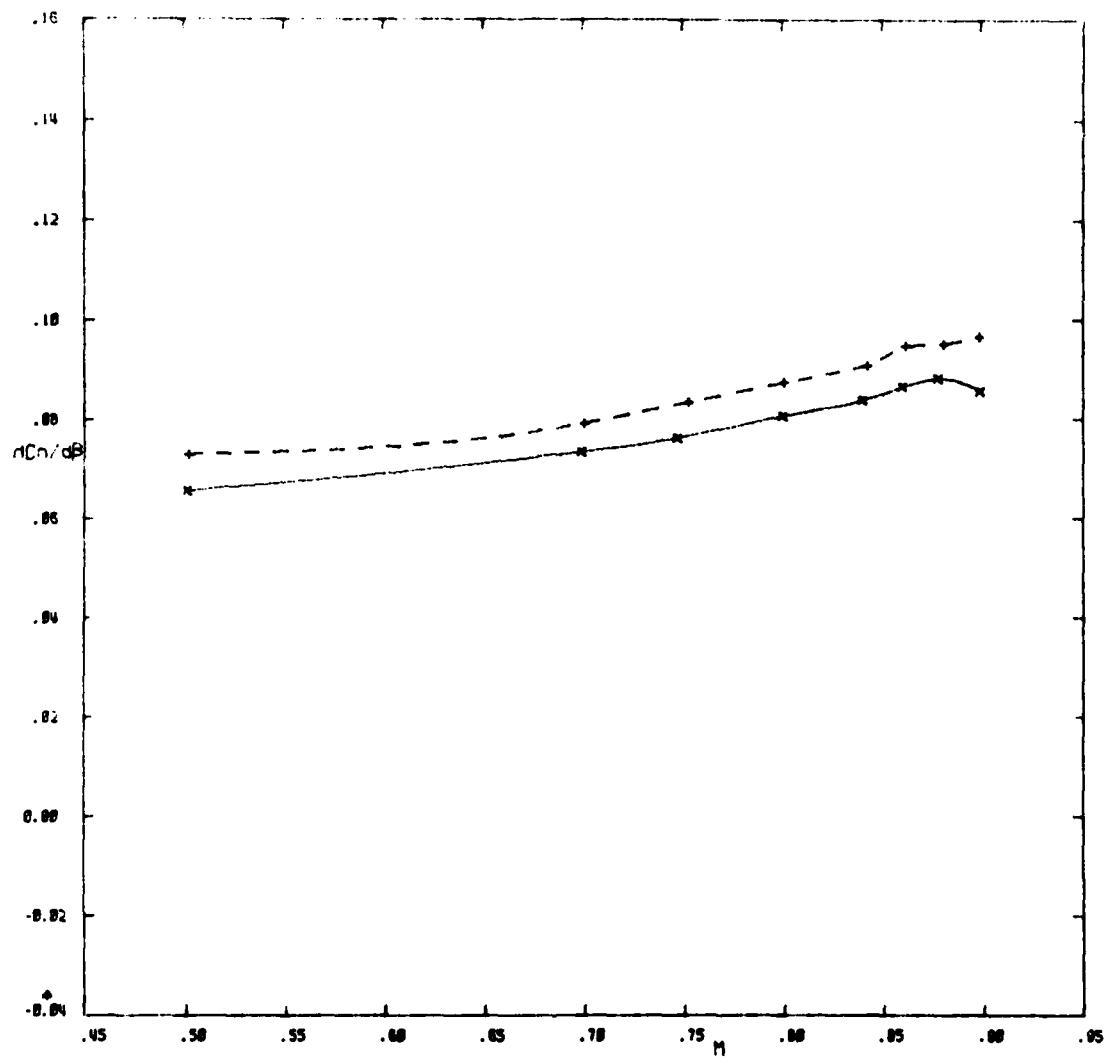
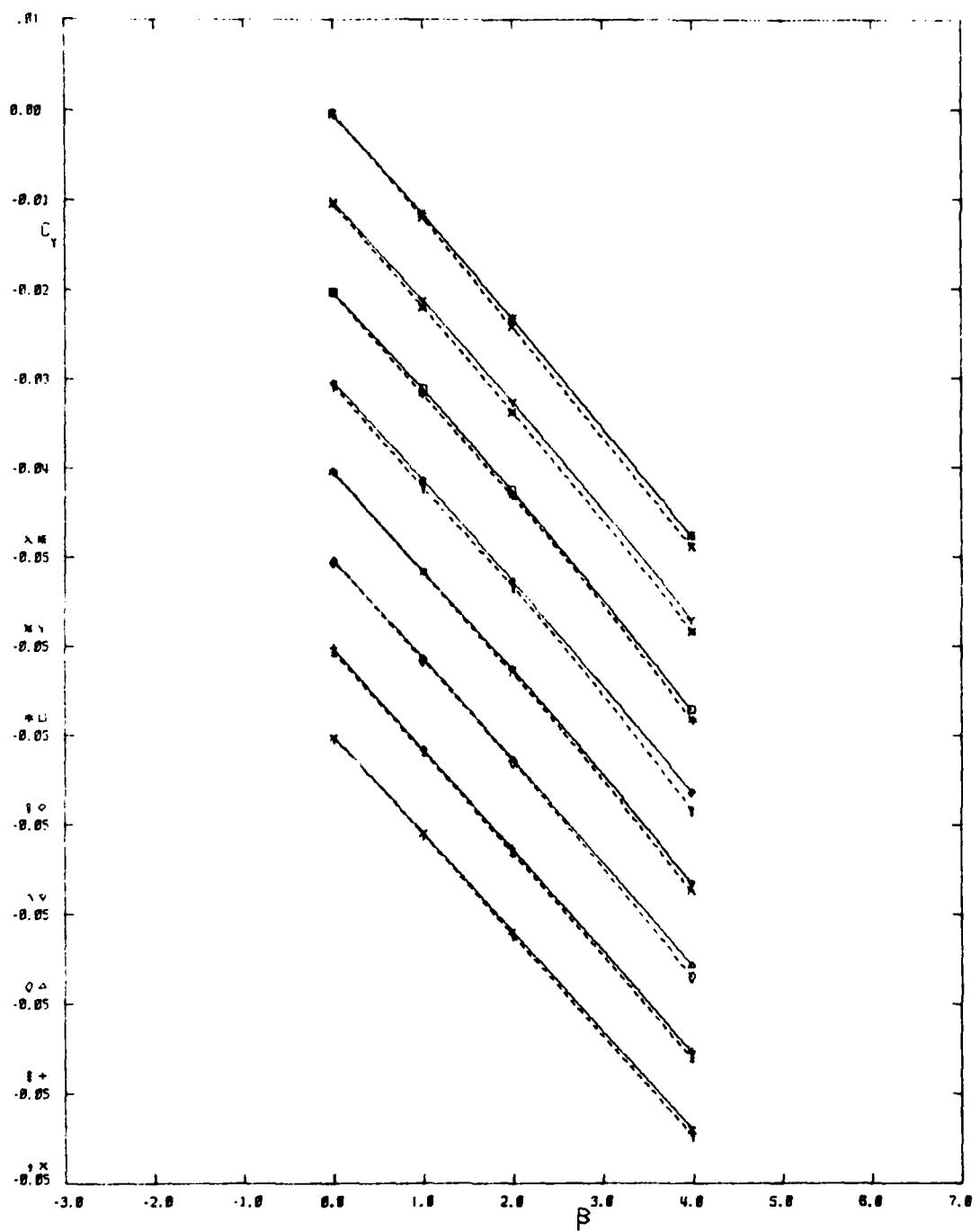


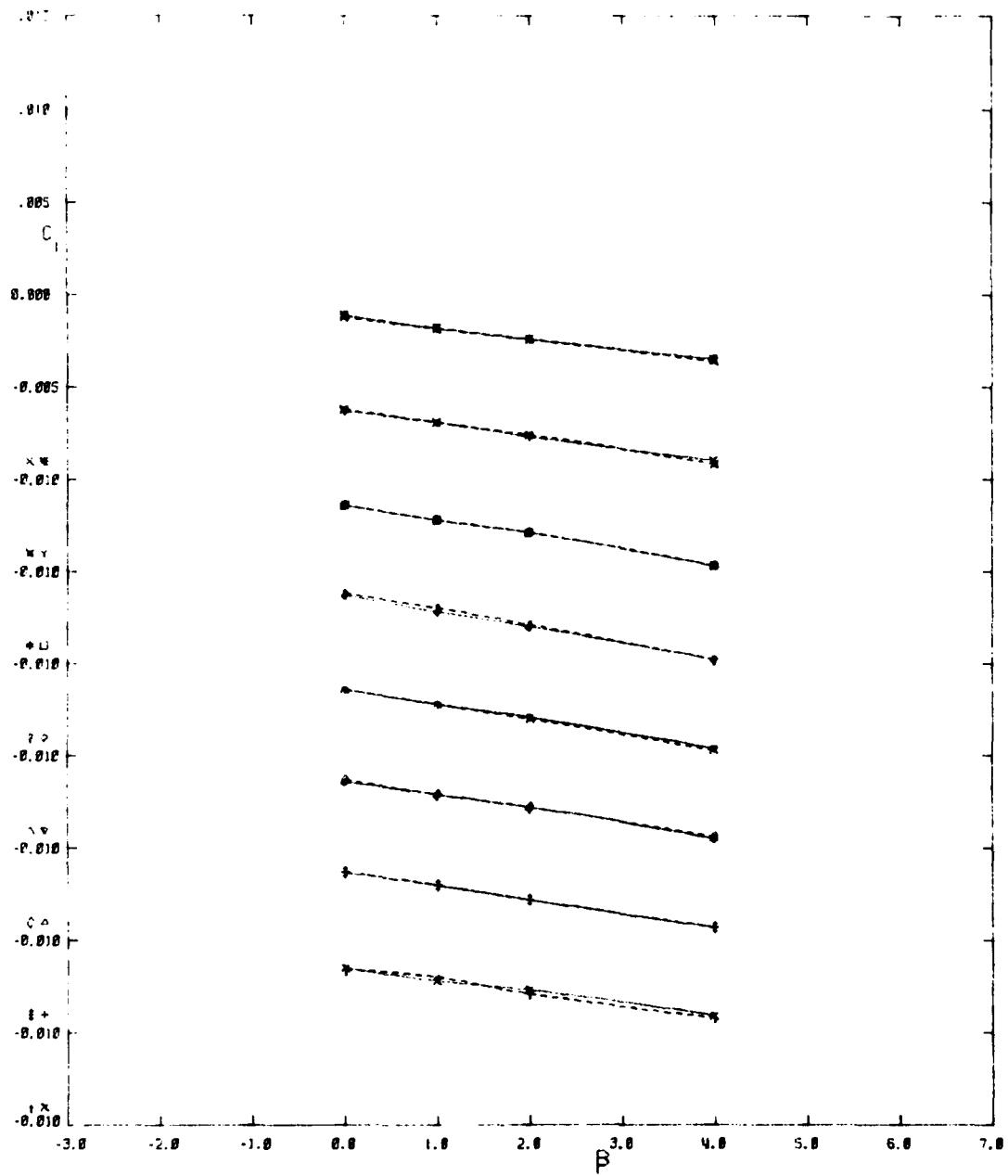
FIGURE 14. VARIATION OF WEATHERCOCK STABILITY DERIVATIVE WITH MACH NUMBER



SYMBOL	M	SYMBOL	M
x	.50	+	.50
+	.70	*	.70
o	.75	o	.75
v	.84	- - -	.84
o	.86	!	.86
y	.89	*	.89
n	.90	x	.90

CLEAN AIRCRAFT      FLAME PACK ON

FIGURE 15. VARIATION OF SIDE FORCE COEFFICIENT WITH SIDESLIP.



SYMBOL	$M$	SYMBOL	$M$
$\times$	.50	$\square$	.50
$+$	.70	$\circ$	.70
$\Delta$	.75	$\diamond$	.75
$\nabla$	.80	$\wedge$	.80
$\diamond$	.84	$\dagger$	.84
$\square$	.86	$*$	.86
$\circ$	.88	$\times$	.88
$\wedge$	.90	$x$	.90

FIGURE 16. VARIATION OF ROLLING MOMENT COEFFICIENT WITH SIDESLIP.

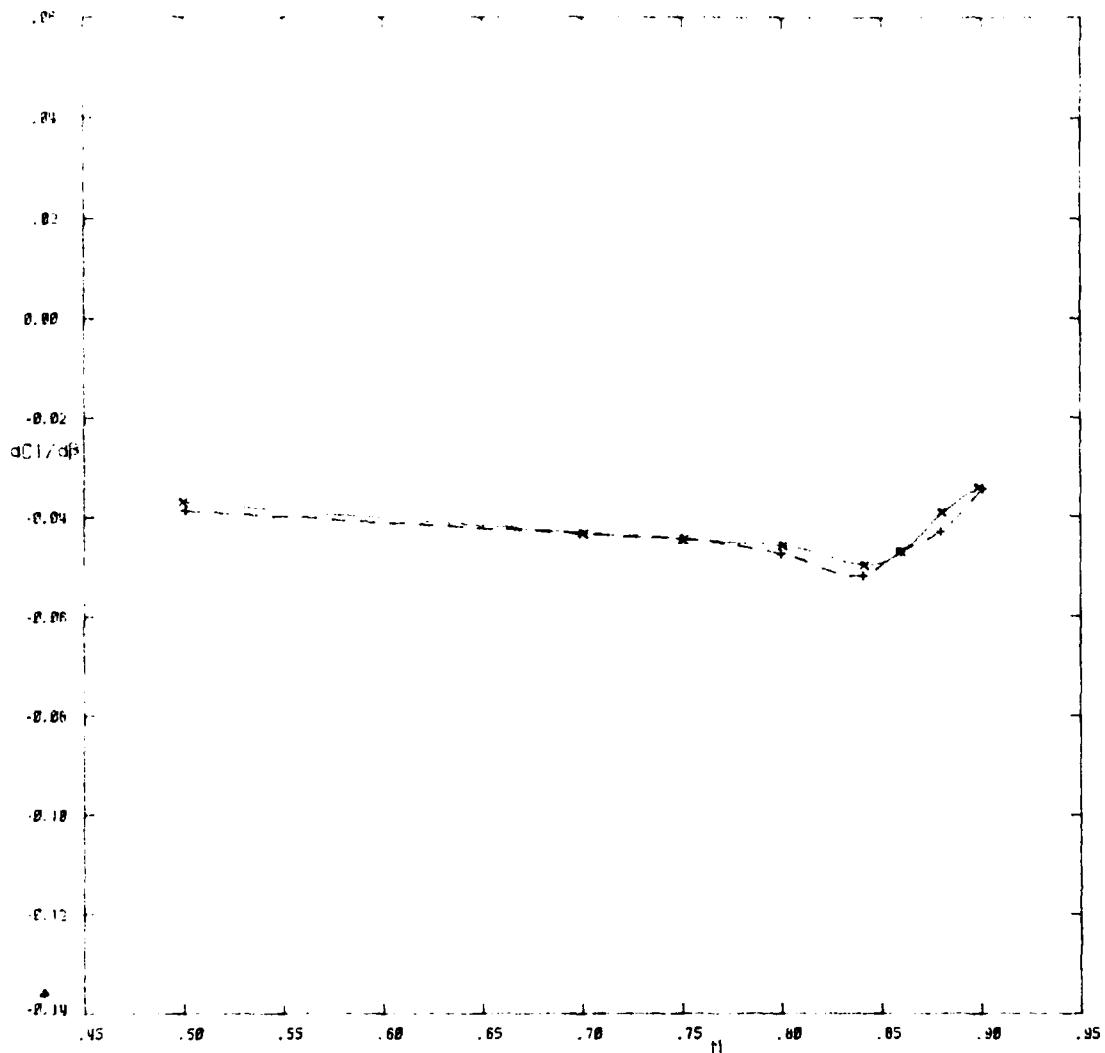


FIGURE 17. VARIATION OF EFFECTIVE DIHEDRAL WITH MACH NUMBER.

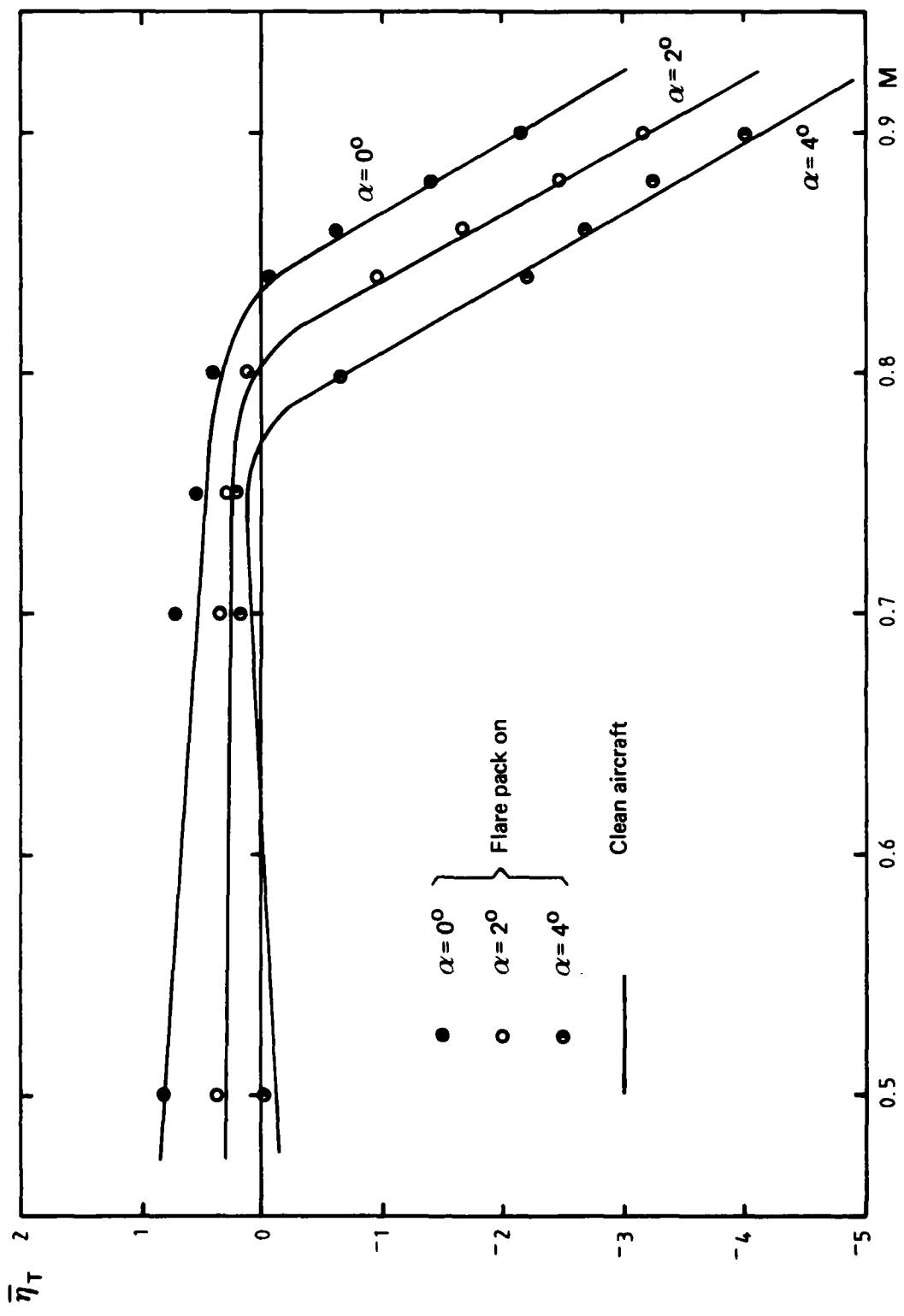


FIG. 18 EFFECT OF FLARE PACK ON TAILPLANE ANGLE TO TRIM

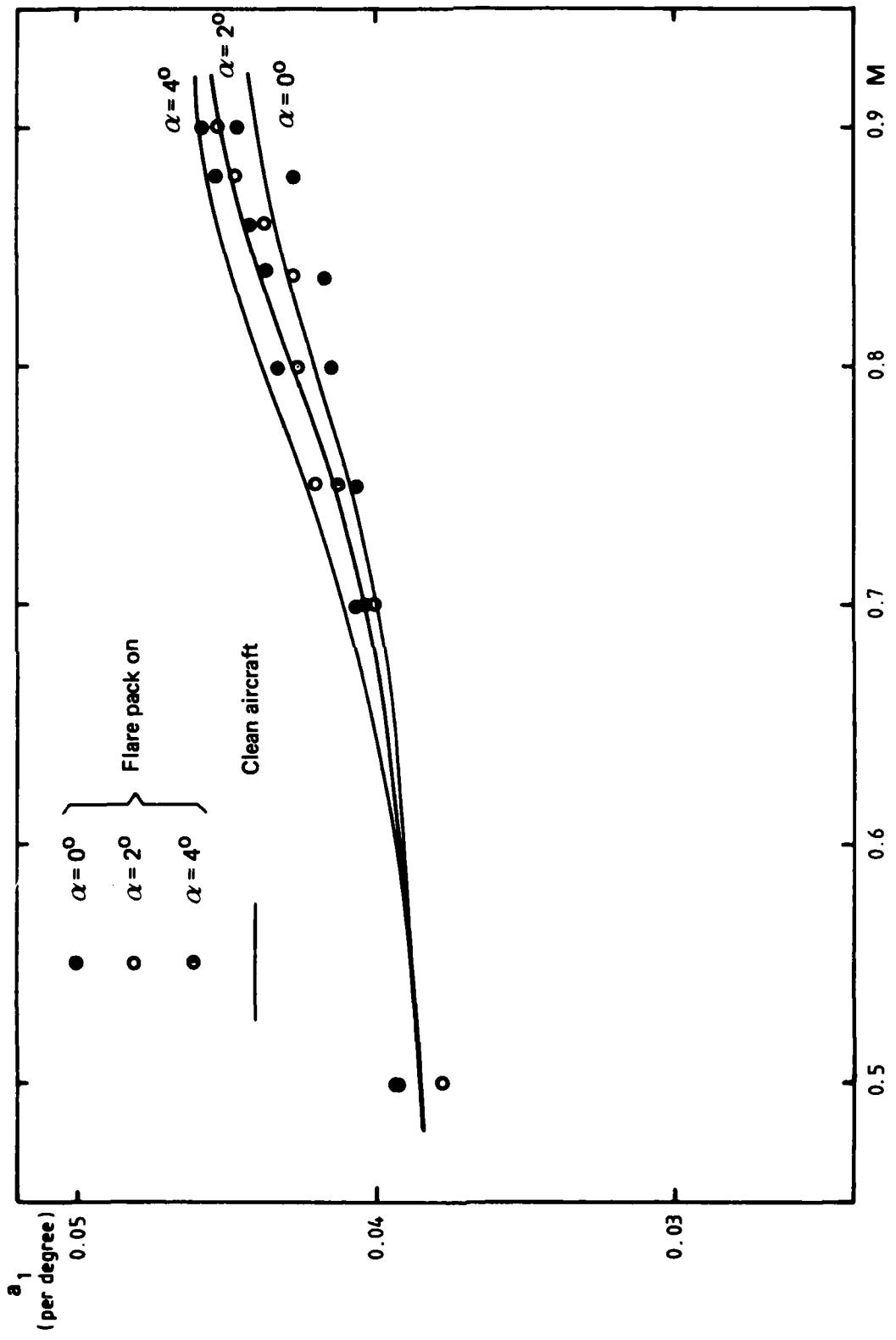


FIG. 19 EFFECT OF FLARE PACK ON TAILPLANE EFFECTIVENESS

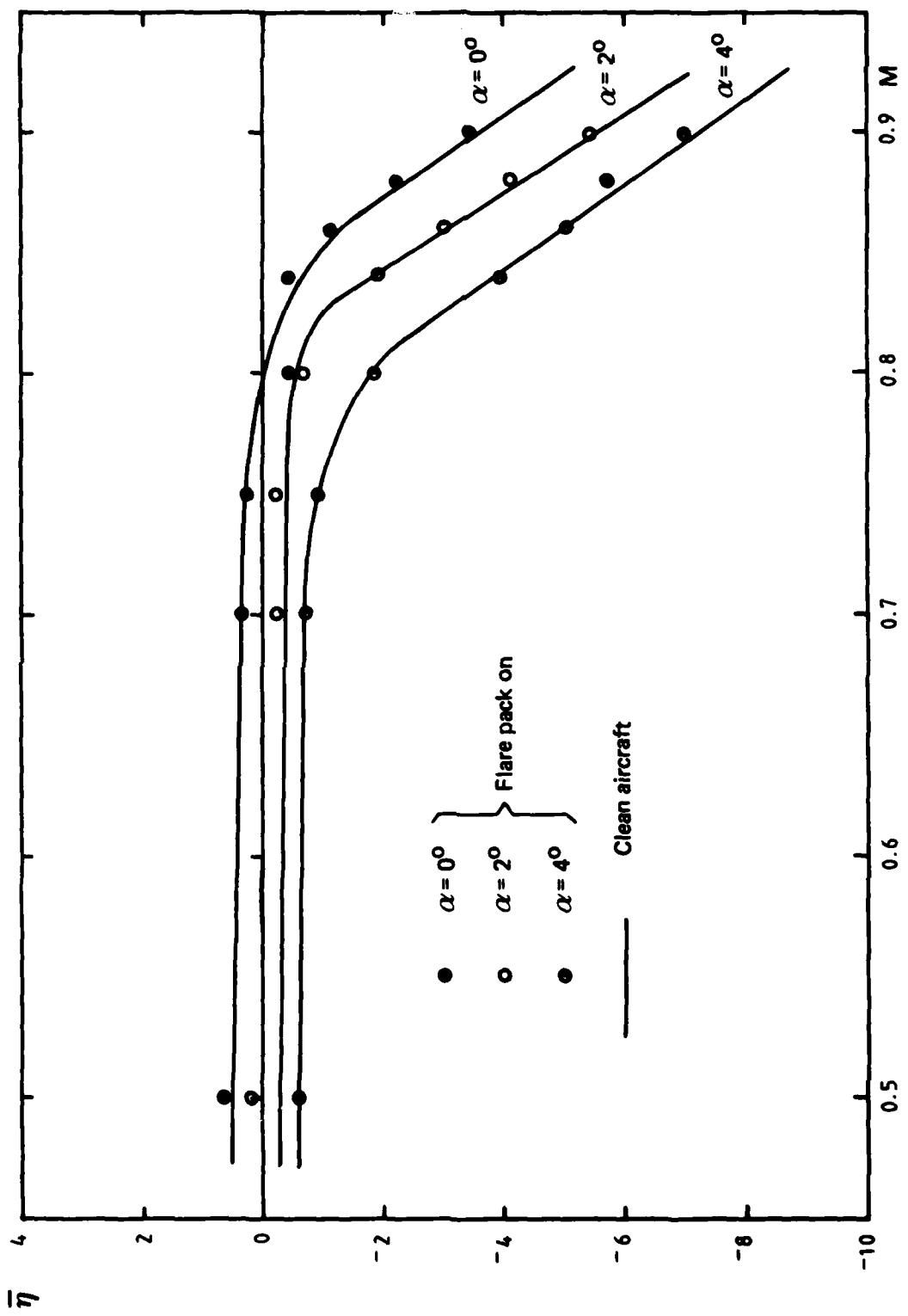


FIG. 20 EFFECT OF FLARE PACK ON ELEVATOR ANGLE TO TRIM

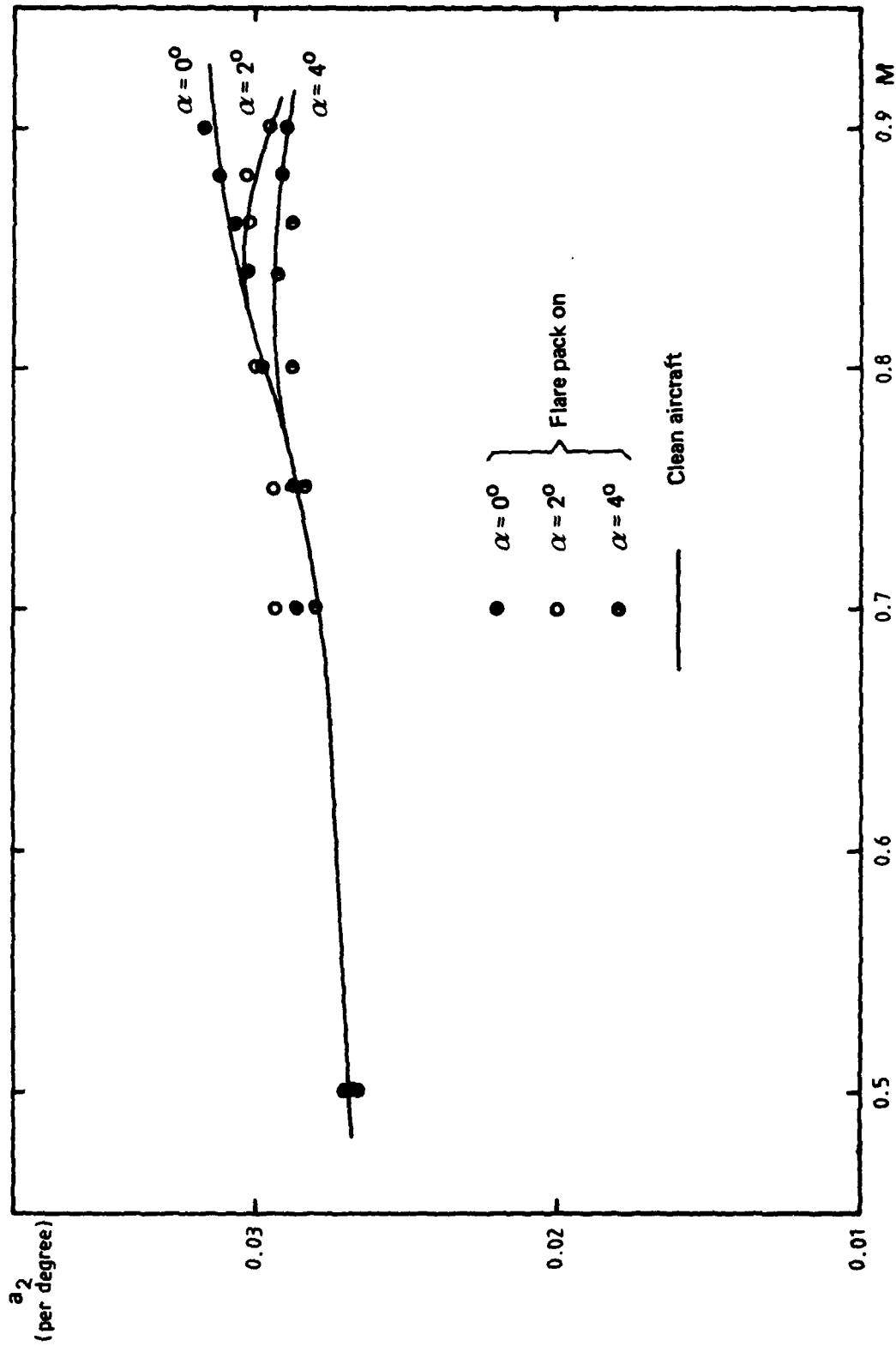


FIG. 21 EFFECT OF FLARE PACK ON ELEVATOR EFFECTIVENESS

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